

# 1      **Adaptive single-loop reweighted numerical integration for estimating** 2      **response moment functions under hybrid aleatory and epistemic uncertainties**

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## 9      **Abstract**

10      Efficient estimation of response moment functions under hybrid aleatory and epistemic uncertainties  
11      remains challenging because it requires multidimensional integration and repeated evaluations of  
12      complex system responses. This paper proposes an adaptive single-loop reweighted numerical  
13      integration (adaptive SLRNI) approach for evaluating the response mean and standard deviation  
14      functions, as well as their bounds, within the framework of the direct probability integral method. The  
15      original SLRNI approach relies on complex Voronoi partitioning and empirically specified numbers of  
16      representative points. To overcome these limitations, an adaptive probability-equalized  $k$ -nearest-  
17      neighbor (APEK) point-selection strategy is devised to generate uniformly distributed representative  
18      points with balanced probability weights. An adaptive batch-enrichment mechanism incrementally adds  
19      points according to a prescribed precision criterion, which eliminates empirical parameter selection and  
20      allows parallel computation. Moreover, the generalized probability density integral equations (PDIEs)  
21      are introduced to describe the probabilistic mapping between hybrid uncertain inputs and system  
22      responses. The response moment functions are derived from the PDIEs and then reformulated through  
23      a generalized reweighting scheme using a complete auxiliary density. By combining the APEK point-  
24      selection strategy with the single-loop reweighting scheme, the proposed adaptive SLRNI enables

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25 accurate computation of moment functions and their bounds without nested sampling. The approach is  
26 non-intrusive and applicable to nonlinear multiphysics coupled dynamic systems. Four examples,  
27 including a mathematical problem, a cantilever tube, a 120-bar truss and an offshore wind turbine tower,  
28 are presented to demonstrate the accuracy, efficiency, and versatility of the proposed approach.

29 **Keywords:** Hybrid uncertainties; Response moment functions; Adaptive single-loop reweighted  
30 numerical integration; APEK point-selection strategy; Parallel computation

## 31 **1. Introduction**

32 Uncertainties in engineering structural systems arise from various sources, including geometric  
33 parameters, material properties, and environmental loads. These uncertainties are commonly  
34 categorized into two types [1,2]: aleatory uncertainty, which results from inherent randomness, and  
35 epistemic uncertainty, which stems from limited data or incomplete knowledge. In practice, these two  
36 types of uncertainty rarely occur in isolation and more often coexist and interact [3]. Therefore, rigorous  
37 quantification of both aleatory and epistemic uncertainties is essential for reliable performance  
38 assessment and safety evaluation of structural systems [4].

39 Several imprecise probability models have been developed to represent hybrid uncertainties, such as  
40 evidence theory [5–8], fuzzy possibility models [9–12], and probability boxes (p-boxes) [13–16].  
41 Among these, p-boxes have attracted considerable attention because of their structural simplicity and  
42 their ability to explicitly distinguish between aleatory and epistemic uncertainties. In general, p-boxes  
43 are classified into distribution-free p-boxes [17] and parametric p-boxes [13]. The present study focuses  
44 on the parametric p-box representations. Under parametric p-boxes, the statistical moments of structural  
45 responses become functions of interval-valued distribution parameters rather than fixed quantities. For  
46 real-world engineering problems, closed-form expressions of these moment functions are often  
47 unavailable. Moreover, their numerical computation can be computationally intensive, as it requires  
48 multidimensional numerical integration even for a single realization of distribution parameters.  
49 Nevertheless, response moment functions play a fundamental role in structural uncertainty

50 quantification because they characterize the probabilistic behavior of structural systems and provide  
51 essential information for reliability assessment and decision-making.

52 For analyzing response moments of systems under parametric p-boxes, existing methods can  
53 generally be divided into double-loop and single-loop approaches. Double-loop approaches include  
54 sampling-based nested methods, in which the outer loop performs sampling while the inner loop  
55 conducts response simulations. Representative approaches include the double-loop Monte Carlo  
56 simulation (DLMCS) [18], the interval Monte Carlo method [19] and the vertex-based DLMCS [20,21].  
57 However, these methods can become computationally prohibitive for complex structural models.  
58 Another class of double-loop approaches consists of the optimization-integration nested methods, where  
59 interval parameters are treated as design variables in the outer loop and numerical integration is  
60 performed in the inner loop. Representative techniques include the optimized parameter sampling [20],  
61 the optimized univariate dimension-reduction method (OUDRM) [22] and the optimized sparse grid  
62 numerical integration method (OSGNI) [23,24]. Although more efficient than sampling-based nested  
63 methods, these approaches rely on gradient-based optimizers and may suffer from convergence to local  
64 optima. To mitigate this issue, global optimization techniques, such as genetic algorithms [25] and  
65 Bayesian global optimization [26], have been introduced. Nevertheless, these methods mainly provide  
66 the bounds of response statistics and cannot provide complete response moment functions.

67 To alleviate the computational burden associated with nested strategies, single-loop approaches have  
68 been proposed. Representative methods include the extended MCS [27], the non-intrusive imprecise  
69 stochastic simulation (NISS) [28] and the polynomial chaos expansion (PCE)-based methods [29,30].  
70 The extended MCS and NISS eliminate nested simulations but still rely on repeated probabilistic  
71 sampling, which may remain computationally expensive for high-dimensional problems. PCE-based  
72 methods have been extended to parametric p-box propagation through augmented stochastic spaces or  
73 conditional surrogate constructions. Such dimensional augmentation leads to a rapid growth of  
74 polynomial basis terms as the combined uncertainty dimension increases, thereby giving rise to the  
75 curse of dimensionality. Recent advances in surrogate-based active learning methods, such as the non-

76 intrusive imprecise probabilistic integration (NIPI) [31], the collaborative and adaptive Bayesian  
77 optimization (CABO) [32] and the parallel Bayesian quadrature optimization (PBQO) [33], have further  
78 improved computational efficiency through adaptive sampling strategies. Nevertheless, surrogate  
79 construction typically assumes a certain level of smoothness in the response surface, which may reduce  
80 accuracy for strongly nonlinear or non-smooth structural responses.

81 Most existing single-loop or optimization-integration nested approaches are primarily designed to  
82 compute the bounds of response statistics, rather than calculating the complete response moment  
83 functions over the epistemic parameter domain. Accordingly, the evaluation of complete response  
84 moment functions under parametric p-boxes remains computationally demanding, especially for  
85 complex finite element (FE) models and nonlinear dynamic structures with moderate to high levels of  
86 overall uncertainty. Therefore, developing accurate and efficient algorithms that avoid nested  
87 constructions and dimensional enlargement remains a crucial open challenge.

88 Recently, the direct probability integral method (DPIM) [34,35] has been proposed as a unified  
89 framework for stochastic uncertainty quantification, with several subsequent developments reported in  
90 [36–39]. Inspired by DPIM, a single-loop reweighted numerical integration (SLRNI) approach [40] was  
91 previously developed for reliability analysis under parametric p-boxes. Although it can be extended to  
92 moment evaluation, the original SLRNI employs a generalized F-discrepancy (GF)-based point-  
93 selection strategy [41] to generate representative points and their assigned probabilities for numerical  
94 integration. However, this procedure requires Voronoi partitioning and a large number of Monte Carlo  
95 samples for probability assignment, while the number of representative points must also be specified in  
96 advance.

97 To address these limitations, a novel adaptive probability-equalized  $k$ -nearest-neighbor (APEK)  
98 point-selection strategy is proposed. By integrating the suggested reweighting scheme with this point-  
99 selection strategy, an adaptive SLRNI approach is developed to evaluate the response mean and standard  
100 deviation functions under parametric p-boxes within a single-loop computational framework. The  
101 proposed approach enables the direct reconstruction of response moment functions over the epistemic

102 parameter domain. The main contributions of this work are summarized as follows:

103 (1) Generalized probability density integral equations (PDIEs) are formulated to characterize the  
104 probabilistic mapping between hybrid aleatory-epistemic uncertain inputs and structural responses.  
105 Based on these formulations, rigorous theoretical expressions for response moment functions under  
106 parametric p-boxes are derived.

107 (2) By introducing a complete auxiliary density function, the moment functions are reformulated into  
108 a numerically tractable form. On this basis, a reweighting-based single-loop computational framework  
109 is established to evaluate the response mean and standard deviation functions without dimensional  
110 augmentation or nested simulation.

111 (3) A novel APEK point-selection strategy is proposed to improve the efficiency and stability of  
112 numerical integration. The assigned probabilities are estimated using local  $k$ -nearest-neighbor volumes,  
113 and the representative point set is adaptively enriched according to a prescribed accuracy criterion.

114 Unlike surrogate-based approaches such as PCE constructed in augmented stochastic spaces, the  
115 proposed framework reformulates response moment functions through a reweighting mechanism in the  
116 original uncertainty space. This formulation avoids dimensional augmentation while preserving a  
117 strictly single-loop computational form.

118 The remainder of this paper is organized as follows. Section 2 formulates the problem considered in  
119 this study and introduces the generalized PDIEs for quantifying hybrid aleatory and epistemic  
120 uncertainties. Section 3 presents the theoretical basis and implementation procedure of the adaptive  
121 SLRNI approach. Section 4 presents several representative examples to validate the performance of the  
122 proposed approach. Section 5 provides concluding remarks and outlines directions for future research.

## 123 **2. Preliminaries**

124 This section presents a unified theoretical foundation based on the generalized PDIEs for hybrid  
125 uncertainty quantification. Section 2.1 formulates the hybrid aleatory-epistemic uncertainty model, and  
126 Section 2.2 derives the associated response moment functions via PDIEs.

## 127 2.1 Hybrid aleatory-epistemic uncertainty model

128 For a static structural system under parametric p-boxes, the response of interest  $Z$  is defined as:

$$129 \quad Z = g(\mathbf{X} | \boldsymbol{\theta}), \quad \boldsymbol{\theta} \in D_{\boldsymbol{\theta}} \quad (1)$$

130 where  $\mathbf{X} = (X_1, X_2, \dots, X_n)$  is an  $n$ -dimensional input random vector whose components  $X_i$  are assumed  
131 to be mutually independent;  $g(\cdot)$  denotes the structural response function that maps random vector  $\mathbf{X}$  to  
132 the response  $Z$ . Aleatory uncertainty is associated with the inherent randomness of  $\mathbf{X}$ , and the probability  
133 distribution type of each  $X_i$  is assumed to be known. Epistemic uncertainties arise from incomplete  
134 knowledge of the distribution parameters (e.g., the mean and standard deviation) of the input variables  
135  $\mathbf{X}$ . Epistemic uncertainties are collected in an interval vector  $\boldsymbol{\theta}$ , whose admissible domain  $D_{\boldsymbol{\theta}}$  is defined  
136 as:

$$137 \quad D_{\boldsymbol{\theta}} = [\theta_1^l] \times \dots \times [\theta_m^l]. \quad (2)$$

138 where each component  $\theta_i^l = [\theta_i^L, \theta_i^U]$  is defined over an interval with lower and upper bounds  $\theta_i^L$  and  $\theta_i^U$  ;  
139  $m$  is the number of interval parameters; here, the symbol ‘ $\times$ ’ denotes the Cartesian product. Thus, the  
140 joint probability density function (PDF) of  $\mathbf{X}$  conditional on a given  $\boldsymbol{\theta}$  is denoted by  $p_{\mathbf{X}}(\mathbf{x}|\boldsymbol{\theta})$ .

141 For a dynamic structural system under such parametric p-boxes, the dynamic response function can  
142 be expressed as:

$$143 \quad Z(t) = g(\mathbf{X} | \boldsymbol{\theta}, t), \quad \boldsymbol{\theta} \in D_{\boldsymbol{\theta}},$$

144 where  $Z(t)$  denotes the structural response at time  $t$  for a given realization of the input uncertainties. The  
145 extreme-value response within a given time interval  $(0, T]$  is often of particular interest, since it  
146 significantly influences system performance and first-passage reliability. The extreme-value response  
147  $Z_{\text{ext}}$  is defined as:

$$148 \quad Z_{\text{ext}} = g_{\text{ext}}(\mathbf{X} | \boldsymbol{\theta}) = \max_{t \in (0, T]} \{g(\mathbf{X} | \boldsymbol{\theta}, t)\}, \boldsymbol{\theta} \in D_{\boldsymbol{\theta}}. \quad (3)$$

149 where  $g_{\text{ext}}(\cdot)$  represents the extreme-value operator that maps the dynamic response over the time  
150 interval  $(0, T]$  to the maximum value  $Z_{\text{ext}}$ .

151 This study investigates the moment functions of the response of interest and the extreme-value

152 response for static and dynamic structural systems, as defined in Eqs. (1) and (3), respectively. In the  
 153 present formulation, the interval parameters are assumed to be mutually independent.

## 154 **2.2 Formulation of response moment functions via generalized PDIE**

155 Within the framework of the DPIM [34,35], PDIEs are established to characterize the propagation of  
 156 pure random uncertainty. In Ref. [42], recent theoretical developments of PDIEs are presented via the  
 157 probability conservation at the differential element and Dirac delta function.

158 In this section, the generalized PDIEs are formulated to describe the relationship between PDFs of  
 159 hybrid uncertain inputs and PDFs of structural responses. Explicit expressions for response moment  
 160 functions are subsequently derived based on the generalized PDIEs.

161 For a static structural system subjected to parametric p-boxes, the PDF  $p_Z(z|\boldsymbol{\theta})$  of response at **the**  
 162 given epistemic parameters  $\boldsymbol{\theta}$  can be expressed as:

$$163 \quad p_Z(z|\boldsymbol{\theta}) = \int_{\Omega_X} \delta[z - g(\mathbf{x})] p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}, \quad (4)$$

164 where  $\delta(\cdot)$  is the Dirac delta function;  $\Omega_X$  denotes **the admissible domain (support) of the input random**  
 165 **variables.**

166 The  $i$ -th raw moment function  $m_i(\boldsymbol{\theta})$  of response  $Z$  is defined as:

$$167 \quad m_i(\boldsymbol{\theta}) = E[g^i(\mathbf{x}|\boldsymbol{\theta})] = \int_{\Omega_Z} z^i p_Z(z|\boldsymbol{\theta}) dz. \quad (5)$$

168 Substituting Eq. (4) into Eq. (5) yields:

$$169 \quad m_i(\boldsymbol{\theta}) = \int_{\Omega_Z} \int_{\Omega_X} \delta[z - g(\mathbf{x})] z^i p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x} dz = \int_{\Omega_X} \left\{ \int_{\Omega_Z} \delta[z - g(\mathbf{x})] z^i dz \right\} p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}. \quad (6)$$

170 By utilizing the sifting property of the Dirac delta function,  $m_i(\boldsymbol{\theta})$  can be rewritten as:

$$171 \quad m_i(\boldsymbol{\theta}) = \int_{\Omega_X} g^i(\mathbf{x}) p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}. \quad (7)$$

172 From Eq. (7), the response mean and variance functions can be obtained as:

$$173 \quad \mu(\boldsymbol{\theta}) = m_1(\boldsymbol{\theta}) = \int_{\Omega_X} g(\mathbf{x}) p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}, \quad (8)$$

174 and

175 
$$\sigma^2(\boldsymbol{\theta}) = m_2(\boldsymbol{\theta}) - (m_1(\boldsymbol{\theta}))^2 = \int_{\Omega_X} g^2(\mathbf{x}) p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x} - \mu(\boldsymbol{\theta})^2. \quad (9)$$

176 For a dynamic structural system under hybrid uncertainties, the generalized PDIE can be formulated  
177 as:

178 
$$p_Z(z|\boldsymbol{\theta}, t) = \int_{\Omega_X} \delta[z(t) - g(\mathbf{x}, t)] p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}, \quad (10)$$

179 where  $p_Z(z|\boldsymbol{\theta}, t)$  denotes the instantaneous PDF of dynamic response at  $\boldsymbol{\theta}$ .

180 The generalized PDIEs in Eqs. (4) and (10) characterize the propagation of the PDFs of the input  
181 variables to the output response PDFs for static and dynamic structural systems, respectively.

182 In this study, the extreme-value response for dynamic structural systems is of particular interest. Its  
183 PDF can be expressed as:

184 
$$p_{Z_{\text{ext}}}(z_{\text{ext}}|\boldsymbol{\theta}) = \int_{\Omega_X} \delta[z_{\text{ext}} - g_{\text{ext}}(\mathbf{x})] p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}. \quad (11)$$

185 The mean and variance functions of the extreme-value response are obtained from Eq. (11), yielding:

186 
$$\mu_{\text{ext}}(\boldsymbol{\theta}) = \int_{\Omega_X} g_{\text{ext}}(\mathbf{x}) p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x}, \quad (12)$$

187 and

188 
$$\sigma_{\text{ext}}^2(\boldsymbol{\theta}) = \int_{\Omega_X} g_{\text{ext}}^2(\mathbf{x}) p_X(\mathbf{x}|\boldsymbol{\theta}) d\mathbf{x} - \mu_{\text{ext}}(\boldsymbol{\theta})^2. \quad (13)$$

189 The lower and upper bounds of the moment functions are determined through the following  
190 optimization problems.

191 Bounds on the mean:

192 
$$\begin{aligned} \mu^L &= \min_{\boldsymbol{\theta} \in D_\theta} \mu(\boldsymbol{\theta}) & \mu_{\text{ext}}^L &= \min_{\boldsymbol{\theta} \in D_\theta} \mu_{\text{ext}}(\boldsymbol{\theta}) \\ \mu^U &= \max_{\boldsymbol{\theta} \in D_\theta} \mu(\boldsymbol{\theta}) & \mu_{\text{ext}}^U &= \max_{\boldsymbol{\theta} \in D_\theta} \mu_{\text{ext}}(\boldsymbol{\theta}) \end{aligned} \quad (14)$$

193 Bounds on the standard deviation:

194 
$$\begin{aligned} \sigma^L &= \min_{\boldsymbol{\theta} \in D_\theta} \sigma(\boldsymbol{\theta}) & \sigma_{\text{ext}}^L &= \min_{\boldsymbol{\theta} \in D_\theta} \sigma_{\text{ext}}(\boldsymbol{\theta}) \\ \sigma^U &= \max_{\boldsymbol{\theta} \in D_\theta} \sigma(\boldsymbol{\theta}) & \sigma_{\text{ext}}^U &= \max_{\boldsymbol{\theta} \in D_\theta} \sigma_{\text{ext}}(\boldsymbol{\theta}) \end{aligned} \quad (15)$$

195 Once the response moment functions are obtained, their bounds can be evaluated through  
196 optimization over the epistemic parameter domain. However, direct evaluation of these moment

197 functions is computationally demanding, as it involves repeated response simulations and high-  
198 dimensional numerical integration. The present work aims to estimate response moment functions and  
199 their bounds through an adaptive single-loop **reweighted computational framework**.

### 200 **3. The proposed adaptive SLRNI approach**

201 Section 3.1 outlines the proposed SLRNI framework. Section 3.2 reformulates the response moment  
202 functions using an auxiliary density. Section 3.3 **introduces the adaptive point-selection strategy and its**  
203 **integration with the reweighting scheme**. Section 3.4 summarizes the implementation procedure.

#### 204 **3.1. Overview of the adaptive SLRNI**

205 The basic response moment formulas (8)–(13) involve multidimensional integrals that couple  
206 random variables and interval parameters. Direct evaluation is computationally demanding, especially  
207 for complex structural models. To address this difficulty, a complete auxiliary density is first introduced  
208 to reformulate moment functions into **the reweighted integral forms**. The proposed approach combines  
209 the reweighted formulations with an adaptive probability-equalized point-selection strategy **to evaluate**  
210 **the response mean and standard deviation functions**. Once the moment functions are obtained, **their**  
211 **bounds are determined through optimization over the epistemic parameter space** without additional  
212 model evaluations.

#### 213 **3.2. Auxiliary density-based response moment functions**

214 Following the idea introduced in [40], a **complete** auxiliary density function is adopted to reformulate  
215 the response moment functions.

216 For a static structural system, the response mean function in Eq. (8) can be reformulated as:

$$217 \quad \mu(\boldsymbol{\theta}) = \int_{\Omega_x} \frac{g(\mathbf{x}) p_x(\mathbf{x} / \boldsymbol{\theta})}{\phi(\mathbf{x})} \phi(\mathbf{x}) d\mathbf{x}, \quad (16)$$

218 where  $\phi(\mathbf{x})$  denotes the complete auxiliary density [40], **which is defined as:**

$$219 \quad \phi(\mathbf{x}) = \frac{1}{\lambda} \int_{D_\theta} p(\mathbf{x} / \boldsymbol{\theta}) d\boldsymbol{\theta},$$

220 where  $\lambda$  is a normalization constant that ensures  $\phi(\mathbf{x})$  is a valid PDF satisfying  $\phi(\mathbf{x}) > 0$  for  $\mathbf{x} \in \Omega_{\mathbf{x}}$  and  
 221  $\int_{\Omega_{\mathbf{x}}} \phi(\mathbf{x}) d\mathbf{x} = 1$ . Here,  $\phi(\mathbf{x})$  represents a normalized PDF obtained by integrating the conditional PDF  
 222  $p_{\mathbf{x}}(\mathbf{x}|\boldsymbol{\theta})$  over the interval parameter space. **The auxiliary density shares the same support as the target**  
 223 **distribution and remains strictly positive wherever the target density is nonzero.** For numerical  
 224 **robustness, a small positive lower bound is imposed during the evaluation of the auxiliary density.**

225 Similarly, the response variance function (9) can be rewritten as:

$$226 \quad \sigma^2(\boldsymbol{\theta}) = \int_{\Omega_{\mathbf{x}}} \frac{g(\mathbf{x})^2 p_{\mathbf{x}}(\mathbf{x}|\boldsymbol{\theta})}{\phi(\mathbf{x})} \phi(\mathbf{x}) d\mathbf{x} - \left[ \int_{\Omega_{\mathbf{x}}} \frac{g(\mathbf{x}) p_{\mathbf{x}}(\mathbf{x}|\boldsymbol{\theta})}{\phi(\mathbf{x})} \phi(\mathbf{x}) d\mathbf{x} \right]^2. \quad (17)$$

227 For a dynamic structural system, the response mean and variance functions become:

$$228 \quad \mu_{\text{ext}}(\boldsymbol{\theta}) = \int_{\Omega_{\mathbf{x}}} \frac{g_{\text{ext}}(\mathbf{x}) p_{\mathbf{x}}(\mathbf{x}|\boldsymbol{\theta})}{\phi(\mathbf{x})} \phi(\mathbf{x}) d\mathbf{x}, \quad (18)$$

229 and

$$230 \quad \sigma_{\text{ext}}^2(\boldsymbol{\theta}) = \int_{\Omega_{\mathbf{x}}} \frac{g_{\text{ext}}(\mathbf{x})^2 p_{\mathbf{x}}(\mathbf{x}|\boldsymbol{\theta})}{\phi(\mathbf{x})} \phi(\mathbf{x}) d\mathbf{x} - \left[ \int_{\Omega_{\mathbf{x}}} \frac{g_{\text{ext}}(\mathbf{x}) p_{\mathbf{x}}(\mathbf{x}|\boldsymbol{\theta})}{\phi(\mathbf{x})} \phi(\mathbf{x}) d\mathbf{x} \right]^2. \quad (19)$$

231 **These reformulated expressions form the basis for the adaptive numerical integration procedure**  
 232 **developed in the following section.** For simplicity, the model response is uniformly denoted by  $z$ ,  
 233 representing either the static response or the extreme-value response of dynamic systems.

### 234 **3.3. APEK point-selection strategy**

235 The APEK point-selection strategy is developed to generate representative points and their assigned  
 236 probabilities for numerical integration of response moments in the reweighted computation described  
 237 in Section 3.4. The key idea is to discretize the probability space induced by the auxiliary density using  
 238 statistically representative samples. The procedure starts from an initial Sobol point set and iteratively  
 239 enriches the representative points until convergence is achieved. The overall procedure consists of **the**  
 240 **following five steps**, which **are** illustrated in Fig. 1.

#### 241 **Step (1) Probability equalization under auxiliary density.**

242 **Start.** The strategy begins with an initial Sobol sample set  $\mathbf{U}^{(0)} = \{\mathbf{u}_i\}_{i=1}^{N_0} \subset [0, 1]^n$  of size  $N_0$  at iteration

243  $\ell = 0$ . Sobol sequences are used because of their low-discrepancy and space-filling sampling properties.  
 244 A two-stage probability equalization procedure is then performed to obtain the uniformly distributed  
 245 **representative** points. First, each dimension  $\mathbf{u}_j$  is mapped to the physical space by the inverse cumulative  
 246 distribution function (CDF)  $\Phi_j^{-1}$  of the auxiliary density:

$$247 \quad \bar{\mathbf{x}}_j = \Phi_j^{-1}(\mathbf{u}_j), j = 1, 2, \dots, n,$$

248 where the superscript ‘1’ denotes the first transformation and the superscript ‘0’ corresponds to the  
 249 current iteration index  $\ell = 0$ . Rank-based probability equalization is subsequently applied:

$$250 \quad \bar{\mathbf{u}}_j = (\text{rank}(\bar{\mathbf{x}}_j) - 0.5) / N_0 .$$

251 Next,  $\bar{\mathbf{u}}_j$  is mapped again with the inverse CDF  $\Phi_j^{-1}$ :

$$252 \quad \mathbf{x}_j^{(0)} = \Phi_j^{-1}(\bar{\mathbf{u}}_j)$$

253 to obtain the initial probability-equalized representative points  $\mathbf{x}^{(0)}$ . Structural responses  $\mathbf{z}^{(0)} = g(\mathbf{x}^{(0)})$   
 254 (**responses of interest** or extreme-value responses) are evaluated at these points.

### 255 **Adaptive enrichment**

256 At each iteration  $\ell$  ( $\ell = 1, 2, \dots$ ), the representative point set is enriched by appending a Sobol batch  
 257 of size  $N_{\text{grow}}: U_{\text{new}}^{(1)} = \{\mathbf{u}_i^{(1)}\}_{i=1}^{N_{\text{grow}}} \subset [0, 1]^n$ . Only the newly added batch undergoes the two-stage probability  
 258 equalization to produce the additional representative points  $\mathbf{x}_{\text{new}}^{(\ell)}$  and their model response **evaluations**  
 259  $\mathbf{z}_{\text{new}}^{(1)} = g(\mathbf{x}_{\text{new}}^{(1)})$ . The representative points and corresponding response evaluations are then updated  
 260 through row-wise concatenation:

$$261 \quad \mathbf{x}^{(\ell)} = \left[ \mathbf{x}^{(\ell-1)}; \mathbf{x}_{\text{new}}^{(\ell)} \right], \mathbf{z}^{(\ell)} = \left[ \mathbf{z}^{(\ell-1)}; \mathbf{z}_{\text{new}}^{(\ell)} \right].$$

262 Because enrichment is performed in batches, structural response evaluations can be estimated in  
 263 parallel, which further improves the computational efficiency of the proposed method.

264 Steps (2)–(5) are then applied to the updated representative point set  $\mathbf{x}^{(\ell)}$ . In these steps, the points  
 265 contained in the set  $\mathbf{x}^{(1)}$  are swept using index  $q$ ,  $q = 1, 2, \dots, N_1$ ,  $N_1 = N_0 + 1N_{\text{grow}}$  denotes the total  
 266 number of representative points at iteration  $\ell$ .

267 **Step (2) Normalization using analytical envelope.** The representative point set is normalized into  
 268 the unit interval:  $\tilde{x}_{qj} = (x_{qj} - a_j) / (b_j - a_j) \in [0,1]^n$ , so that all coordinates lie on a consistent metric scale.

269 This ensures that subsequent ***k*-nearest-neighbor** local volume computations are well-conditioned.

270 The analytical envelope  $[a_j, b_j]$  is constructed for geometric comparability across dimensions and to  
 271 handle boundary effects. Three typical cases are considered:

272 (i) precise distributions with known finite bound (**such as** a precise uniform):  $[a_j, b_j] = [A_j, B_j]$ ;

273 (ii) precise unbounded distributions characterized by mean and standard deviation (like precise  
 274 normal):  $[a_j, b_j] = [\mu_j - K\sigma_j, \mu_j + K\sigma_j]$ ;  $K$  is the envelope multiplier parameter, and is chosen between  
 275 6 and 8 (integer values).

276 (iii) imprecise distributions with interval-valued parameters or strictly positive support (like  
 277 imprecise lognormal/gamma):  $[a_j, b_j] = [\max\{0, \mu_j^L - K\sigma_j^U\}, \mu_j^U + K\sigma_j^U]$ .

278 **Step (3) *k*-nearest-neighbor local volume calculation.** To determine the local region represented by  
 279 each representative point  $\mathbf{x}_q$ , the *k*-nearest-neighbor radius  $r_q$  is computed under the Chebyshev ( $L_\infty$ )  
 280 metric, inspired by standard *k*-nearest-neighbor density estimation techniques [43,44]. **The metric  $L_\infty$  is**  
 281 **adopted because it matches the hyper-rectangular epistemic domain and enables stable axis-aligned**  
 282 **neighborhood construction.** For each point  $\tilde{\mathbf{x}}_q \in [0,1]^n$ ,  $r_q$  is defined as the distance to its *k*-th nearest  
 283 neighbor  $\mathfrak{X}_{(k)}$  under the  $L_\infty$  metric:

$$284 \quad r_q = \left\| \mathfrak{X}_q - \mathfrak{X}_{(k)} \right\|_\infty. \quad (20)$$

285 Specifically, the distance of each  $\tilde{\mathbf{x}}_q$  to every other point is evaluated and sorted, and the *k*-th smallest  
 286 distance is taken as  $r_q$ . The value of  $r_q$  differs from point to point because it adapts automatically to the  
 287 local point density. This means that dense regions yield smaller radii and sparse regions yield larger  
 288 ones. A small value of  $k \in \{6, 7, 8\}$  **is adopted in accordance with empirical guidelines** for *k*-nearest-  
 289 neighbor density estimation [43,44], which helps control estimator variance in higher dimensions.

290 The radius  $r_q$  defines a hypercube neighborhood centered at  $\tilde{\mathbf{x}}_q$  whose volume is given by:

291 
$$V_q = \prod_{j=1}^n \max(0, \min(\tilde{x}_{qj} + r_q, 1) - \max(\tilde{x}_{qj} - r_q, 0)). \quad (21)$$

292 Although neighborhood-based estimators can be sensitive to dimensionality, the probability-  
 293 equalized representative points exhibit approximately uniform distributions that reduce the effective  
 294 intrinsic dimensionality. In practice, the method remains robust for **moderate** dimensions  
 295 (approximately up to 15 uncertain variables) in hybrid uncertainty problems.

296 **Step (4) Assigned probabilities computation.** Each representative point  $\mathbf{x}_q$  receives a deterministic  
 297 weight proportional to its local volume. The weight is then normalized to produce the assigned  
 298 probabilities  $P_q$ :

299 
$$P_q = \tilde{P}_q / \sum_k \tilde{P}_k, \quad \tilde{P}_q = V_q \phi(\tilde{\mathbf{x}}_q). \quad (22)$$

300 where  $\phi(\tilde{\mathbf{x}}_q)$  is the auxiliary joint density at each point  $\tilde{\mathbf{x}}_q \in [0,1]^n$ . Hence, the assigned probability  $P_q$   
 301 **reflects** the local occupied  $k$ -nearest-neighbor volume at  $\mathbf{x}_q$ .

302 **Step (5) Standard deviation computation at the test set**

303 **Selecting** a batch of fixed Sobol points  $\boldsymbol{\Theta}^{\hat{a}} = \{\Theta_k\}_{k=1}^M$ ,  $\boldsymbol{\Theta}^{\hat{a}} \in \mathbf{D}_\theta$  as the test point set, the standard  
 304 deviation  $\hat{\sigma}^{(\ell)}$  at  $\boldsymbol{\Theta}^{\hat{a}}$  is calculated through Eq. (26):

305 
$$\hat{\sigma}^{(\ell)} = \sqrt{\sum_q w_q(\Theta_k) (z_q - \hat{\mu}^{(\ell)})^2}, \quad \hat{\mu}^{(\ell)} = \sum_q w_q(\Theta_k) z_q, \quad (23)$$

306 which is used in the stopping criterion. After the first enrichment iteration, the algorithm evaluates the  
 307 stopping criterion to determine whether further enrichment is required.

308 **Stopping criterion.**

309 The relative change of standard deviation between two **successive** iterations is estimated:

310 
$$D_\infty^\sigma = \max_{1 \leq k \leq M} \frac{|\hat{\sigma}^{(\ell)} - \hat{\sigma}^{(\ell-1)}|}{|\hat{\sigma}^{(\ell-1)}|}, \quad (\mathbf{L}_\infty \text{ norm}) \quad (24)$$

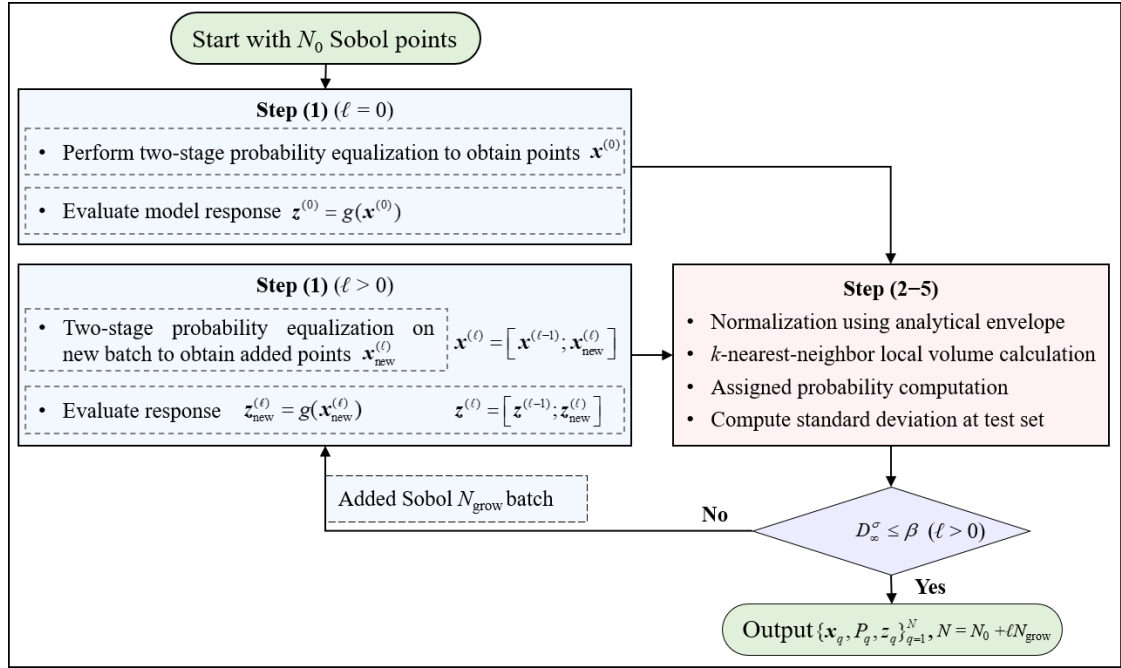
311 The iterative enrichment process terminates when:

312 
$$D_\infty^\sigma \leq \beta, \quad (25)$$

313 where  $\beta$  ( $\beta \leq 0.02$ ) is the given threshold. Once this condition is satisfied, the algorithm terminates and

314 outputs:  $\{\mathbf{x}_q, P_q, z_q\}_{q=1}^N$ , where  $N = N_0 + 1N_{\text{grow}}$ . Otherwise, the procedure returns to step (1) for further  
 315 enrichment. The maximum threshold value of 0.02 is selected to balance accuracy and computational  
 316 efficiency. Smaller values of  $\beta$  improve accuracy but require more representative points and more  
 317 computations, while larger values ( $\beta > 0.02$ ) reduce computation time but may slightly decrease  
 318 accuracy. Numerical validation indicates that the method remains stable within this range.

319 The convergence criterion is evaluated on a fixed Sobol set  $\Theta^{\text{a}}$  to ensure consistent low-discrepancy  
 320 coverage of the epistemic parameter space. Since the response moments are monitored over the entire  
 321 interval domain, the criterion reflects global functional convergence rather than local pointwise errors.



322 Fig. 1 Specific procedure of the proposed adaptive point-selection strategy.  
 323

324 **Remark 1.** Four types of relative changes of response moments are examined on the Sobol test set:

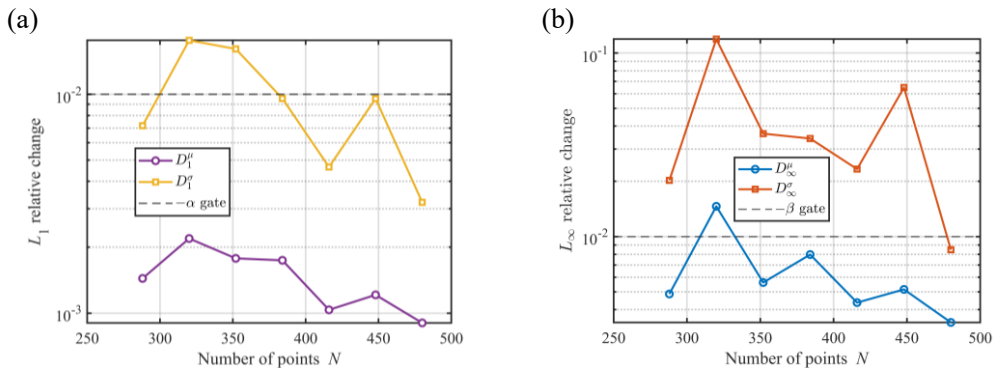
$$D_1^\mu = \frac{1}{M} \sum_{k=1}^M \frac{|\hat{\mu}^{(k)} - \hat{\mu}^{(k-1)}|}{|\hat{\mu}^{(k-1)}|}, \quad D_1^\sigma = \frac{1}{M} \sum_{k=1}^M \frac{|\hat{\sigma}^{(k)} - \hat{\sigma}^{(k-1)}|}{|\hat{\sigma}^{(k-1)}|}, \quad (\mathbf{L}_1 \text{ norm})$$

$$D_\infty^\mu = \max_{1 \leq k \leq M} \frac{|\hat{\mu}^{(k)} - \hat{\mu}^{(k-1)}|}{|\hat{\mu}^{(k-1)}|}, \quad D_\infty^\sigma = \max_{1 \leq k \leq M} \frac{|\hat{\sigma}^{(k)} - \hat{\sigma}^{(k-1)}|}{|\hat{\sigma}^{(k-1)}|}. \quad (\mathbf{L}_\infty \text{ norm})$$

326 The evolutions of these quantities with respect to the number of representative points are illustrated  
 327 in Fig. 2 (Example 1 in Section 4). The results show that the convergence behavior is primarily governed

328 by the relative change of the standard deviation  $D_\infty^\sigma$ , which is selected as the stopping indicator.

329



330

331

Fig. 2 Four different statistical magnitudes of moments at Sobol test set varying with simple size  $N$ .

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**Remark 2.** To assess the quality of representative point generation and probability assignment, the results of the APEK **point-selection strategy** are illustrated in Fig. 3. Fig. 3(a) shows that the representative points are uniformly distributed over the support domain without noticeable clustering or boundary bias. **The Gini index =  $1 - \sum_{i=1}^N P_q^2$  is employed to measure the evenness of the assigned probabilities [45]. A value close to zero indicates a more balanced probability distribution. As shown in Fig. 3(b), Gini index = 0.214 confirms a well-balanced probability assignment. The cumulative probability mass of the largest 1% and 5% assigned probabilities are 2.1% and 9.0%, respectively. These small values indicate that no small subset of representative points dominates the probability mass, which ensures numerical stability. Overall, these diagnostics demonstrate that the proposed APEK strategy generates a well-conditioned representative set suitable for accurate reweighted numerical integration.**

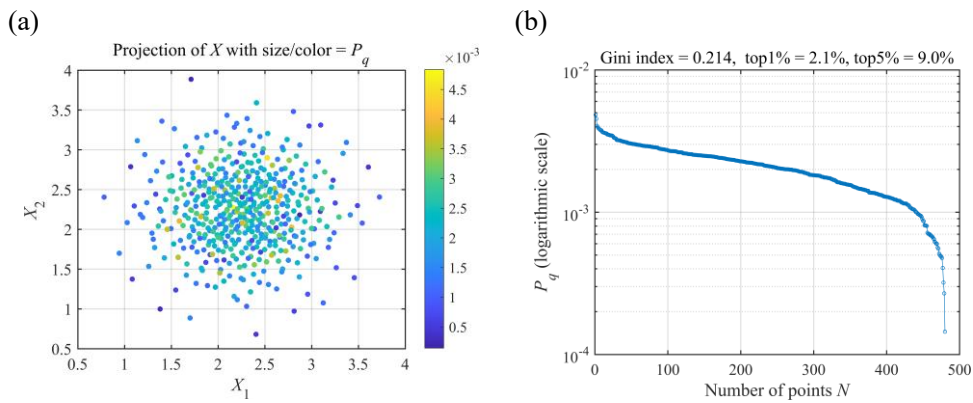


Fig. 3 (a) Spatial distribution of representative points and (b) assigned probabilities on a logarithmic scale.

### 345 **3.4. Evaluation of response moment functions and their bounds**

#### 346 **(1) Response moment function**

347 Given the representative points  $\mathbf{x}_q$  and weights  $P_q$  obtained from APEK point-selection strategy and  
348 the response evaluations  $z_q$ , the response moment functions at any interval parameter  $\boldsymbol{\theta}$  are computed  
349 via single-loop reweighted estimation based on Eqs. (16)–(19).

350 The estimated mean and standard deviation functions of response are calculated by:

$$351 \quad \hat{\mu}(\boldsymbol{\theta}) = \sum_{q=1}^N w_q(\boldsymbol{\theta}) z_q, \quad \hat{\sigma}(\boldsymbol{\theta}) = \sqrt{\sum_{q=1}^N w_q(\boldsymbol{\theta}) z_q^2 - \hat{\mu}^2(\boldsymbol{\theta})}, \quad (26)$$

352 in which  $w_q(\boldsymbol{\theta})$  denotes the reweighted assigned probabilities:

$$353 \quad w_q(\boldsymbol{\theta}) = \frac{P_q f(\mathbf{x}_q | \boldsymbol{\theta})}{\phi(\mathbf{x}_q)}, \quad \sum_{q=1}^N w_q(\boldsymbol{\theta}) = 1, \quad (27)$$

354 where  $f(\mathbf{x}_q | \boldsymbol{\theta})$  and  $\phi(\mathbf{x}_q)$  denote the target joint PDF and the auxiliary probability density at  $\mathbf{x}_q$ ,  
355 respectively.

356 The extreme-value response moments for a dynamic structural system are also calculated by Eqs.  
357 (26) and (27), where  $z_q$  denotes the extreme-value response at points  $\mathbf{x}_q$ .

#### 358 **(2) Response moment bounds**

359 The mean and standard deviation bounds are obtained by solving the optimization problems in Eq.  
360 (14) and Eq. (15) over the domain of  $\boldsymbol{\theta}$ . Since the optimization objectives directly employ the  
361 reweighted formulas in Eq. (26), the computation of moment bounds requires no additional structural  
362 response evaluations. The starfish optimization algorithm [46] is employed for its strong global search  
363 ability and low sensitivity to design parameters.

364 Overall, the proposed framework transforms the estimation of response moments and their bounds  
365 under parametric p-boxes into a single-loop reweighted estimation. It is fully non-intrusive because it  
366 operates solely on probabilistic descriptions of uncertain inputs and treats the structural model as a  
367 black-box, without requiring explicit expressions or gradient information of the response function. With  
368 its non-intrusive nature and parallel capability, the proposed method exhibits strong competitiveness for  
369 complex engineering structures and is further validated in Section 4.

### 370 3.5. Implementation procedure

371 The procedure of the adaptive SLRNI method is summarized below and illustrated in Fig. 4.

#### 372 Step 1: Initialization

373 Define the response function  $g$  and specify all uncertain inputs  $\mathbf{X}$ . For each imprecise random variable,  
374 define the distribution family and the bounds of the intervals  $\theta$ . Set the algorithm parameters:

- 375 • the initial size  $N_0 = 2^8 = 256$  provides a low-discrepancy starting set;
- 376 • the batch size  $N_{\text{grow}} \in \{24, 32\}$  takes integer values; the maximum size  $N_{\text{max}} = 4000$  provides a  
377 safe upper limit;
- 378 • the envelope multiplier  $K \in \{6, 7, 8\}$  and the  $k$ -nearest-neighbor order  $k \in \{6, 7, 8\}$  guarantee  
379 robust and well-conditioned neighborhood construction; the sensitivity of the hyperparameters  
380  $K$  and  $k$  is examined in Appendix A, where the results confirm the robustness of the proposed  
381 method within the recommended parameter range;
- 382 • the number of Sobol test points  $M$  is selected in the integer interval  $\{64, 96\}$ .

383 Overall, these parameter choices offer a reliable balance between accuracy and computational  
384 efficiency.

#### 385 Step 2: Construct the complete auxiliary density $\phi(\mathbf{x})$

386 For each input dimension, integrate the target family over its parameter interval to obtain the marginal  
387 PDF  $\phi_j$  of the auxiliary density function. Multiply all marginals to form the joint density  $\phi(\mathbf{x})$ :

$$388 \phi(\mathbf{x}) = \phi_1 \phi_2 \cdots \phi_j, \quad \phi_j = \frac{1}{\lambda_j} \int_{D_{\theta_j}} f(x_j | \theta_j) d\theta_j, \quad \text{for } j = 1, 2, \dots, n.$$

#### 389 Step 3: Perform APEK point-selection strategy

390 Step 3.1 Generate  $N_0$  Sobol points in the unit cube. Perform a two-stage probability equalization  
391 process to obtain the representative points  $\mathbf{x}^{(0)}$ , and compute the structural response at these points  $\mathbf{z}^{(0)}$ .

392 At iteration  $\ell$ , the added Sobol batch performs the two-stage probability equalization transformation to  
393 obtain the new points  $\mathbf{x}_{\text{new}}^\ell$ . Evaluate the response at these new points:  $\mathbf{z}_{\text{new}}^\ell$ . The representative point set  
394 and response evaluations are updated as:  $\mathbf{x}^{(\ell)} = [\mathbf{x}^{(\ell-1)}; \mathbf{x}_{\text{new}}^{(\ell)}]$  and  $\mathbf{z}^{(\ell)} = [\mathbf{z}^{(\ell-1)}; \mathbf{z}_{\text{new}}^{(\ell)}]$ .

395 Step 3.2 Compute per dimension analytical envelope  $[a_j, b_j]$  based on the distribution family and  
396 parameter bounds. Normalize all coordinates to the unit interval:  $\tilde{x}_{qj} = (x_{qj} - a_j) / (b_j - a_j) \in [0, 1]^n$ .

397 **Step 3.3** In the normalized space  $\tilde{\mathbf{x}}_q \in [0,1]^n$ , compute each point's  $k$ -nearest-neighbor radius  $r_q$  using  
 398 Eq. (20). Then compute the local volume  $V_q$  of the neighborhood through Eq. (21).

399 **Step 3.4** Evaluate the auxiliary density at  $\tilde{\mathbf{x}}_q$  and calculate the assigned probabilities via Eq. (22).

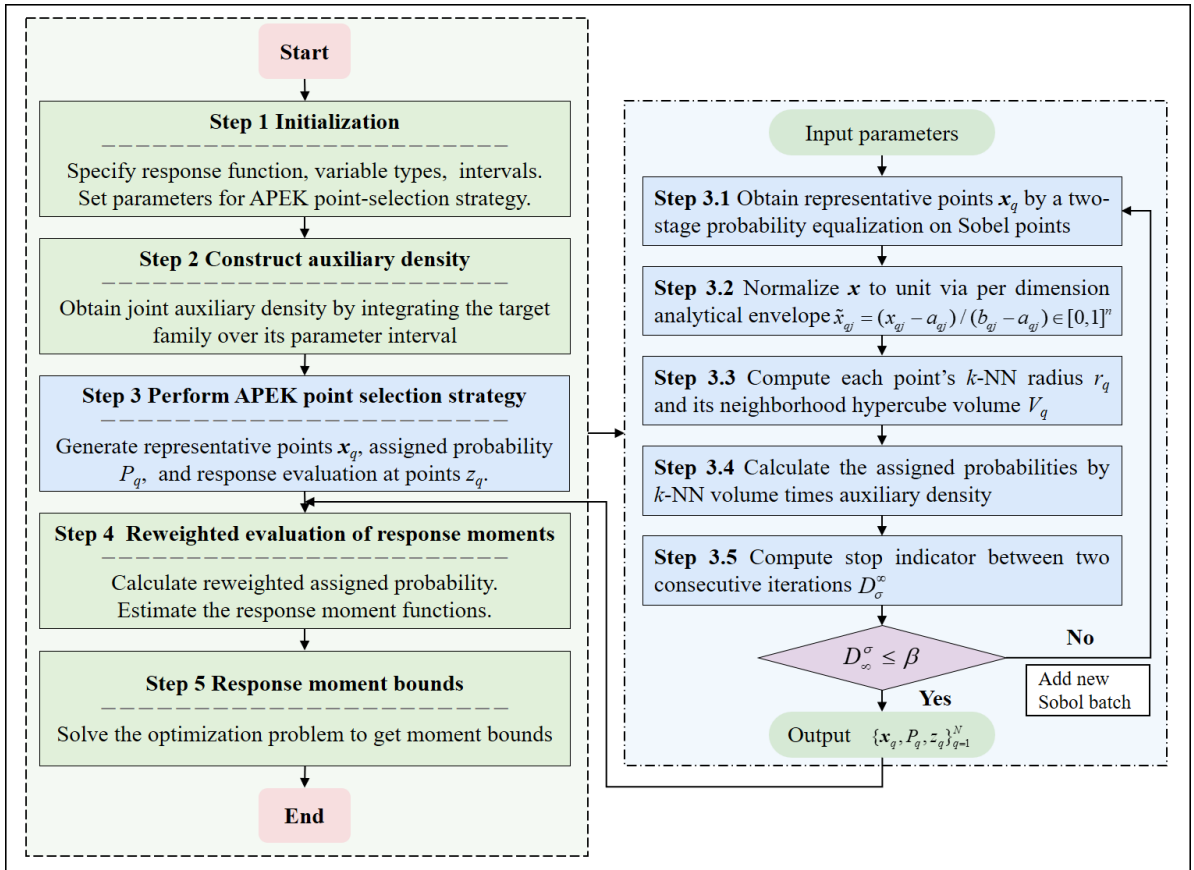
400 **Step 3.5** Compute the stopping indicator between two iterations using Eq. (24). If this indicator  
 401 satisfies the target precision condition defined in Eq. (25), terminate the procedure and output the final  
 402 data:  $\{\mathbf{x}_q, P_q, z_q\}_{q=1}^N$ , where  $N = N_0 + 1N_{\text{grow}}$ . Otherwise, the procedure returns to Steps 3.1–3.4.

403 **Step 4: Response moment functions by reweighted evaluation**

404 Calculate reweighted assigned probabilities  $w_q(\boldsymbol{\theta})$  by Eq. (27) and estimate response mean and  
 405 standard deviation functions via Eqs. (26).

406 **Step 5: Response moment bounds via optimization**

407 Compute the lower and upper bounds of response moments by solving the optimization problems  
 408 defined in Eqs. (14) and (15). The starfish optimization algorithm [46] is employed in this study.



409  
 410 Fig. 4 Flowchart of the proposed adaptive SLRNI for estimating response moment functions and their bounds.

411 **Remark 3.** If all distribution parameters are crisp values, the hybrid uncertainty model reduces to a  
412 classic probability model. **The proposed adaptive SLRNI framework is applicable to systems involving**  
413 **imprecise random variables, mixed imprecise-precise random uncertainties, and also to purely**  
414 **stochastic systems.**

#### 415 **4. Examples**

416 Four examples are presented to assess the accuracy, computational efficiency, and robustness of the  
417 proposed method. The first example is a mathematical benchmark problem, followed by a cantilever  
418 tube structure. The last two examples involve more complex structural systems: a 120-bar truss and a  
419 nonlinear offshore wind turbine tower.

#### 420 **Example 1: A test mathematical function**

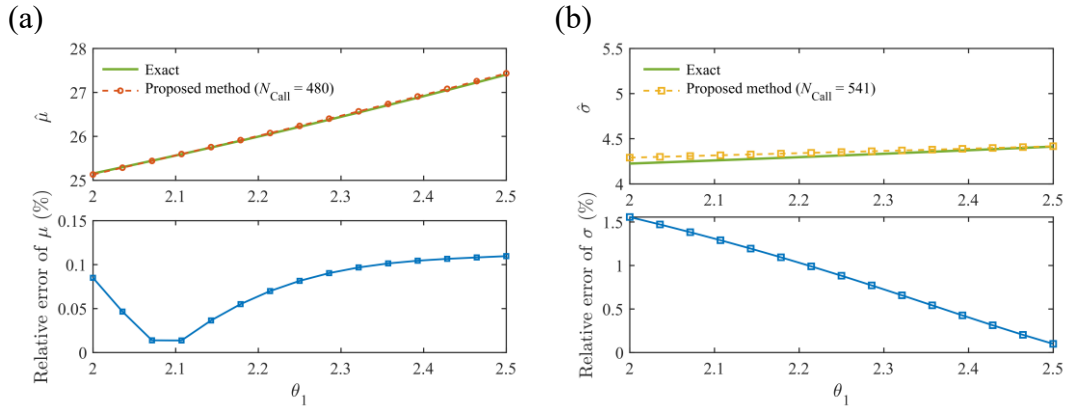
421 The response function of a mathematical problem is expressed as:

$$422 \quad Y = g(\mathbf{X}) = \sum_{i=1}^5 X_i^2$$

423 where  $X_i \sim N(\theta_1 \in [2, 2.5], \theta_2 \in [0.4, 0.45])$ ,  $\theta_1$  and  $\theta_2$  are the mean and standard deviation of input  
424 random variable  $X_i$ , respectively.

425 The mean and standard deviation of **the output response**  $Y$ , together with their relative errors with  
426 respect to the analytical solutions, are illustrated in Fig. 5 and Fig. 6. Note that **the response moments**  
427 **with respect to interval parameters are calculated by fixing the other intervals at their midpoint values**  
428 **in all examples of this study.** The relative errors between the proposed method and the exact solutions  
429 **remain small** (below 2%). Table 1 compares the bounds of response moments obtained using the  
430 proposed method, MCS-BGO (MCS combined with Bayesian global optimization), OSGNI [23] and  
431 OUDRM [22]. Although OSGNI and OUDRM improve computational efficiency compared with  
432 standard MCS, they only provide the bounds of response moments. In contrast, the proposed method  
433 accurately **captures the full variation of the response moment functions** over the interval parameter  
434 domain and provides more detailed statistical information than existing approaches.

435



436

437

Fig. 5 (a) Mean  $\hat{\mu}$  and (b) standard deviation  $\hat{\sigma}$  of  $Y$  by adaptive SLRNI and their errors relative to exact results.

438

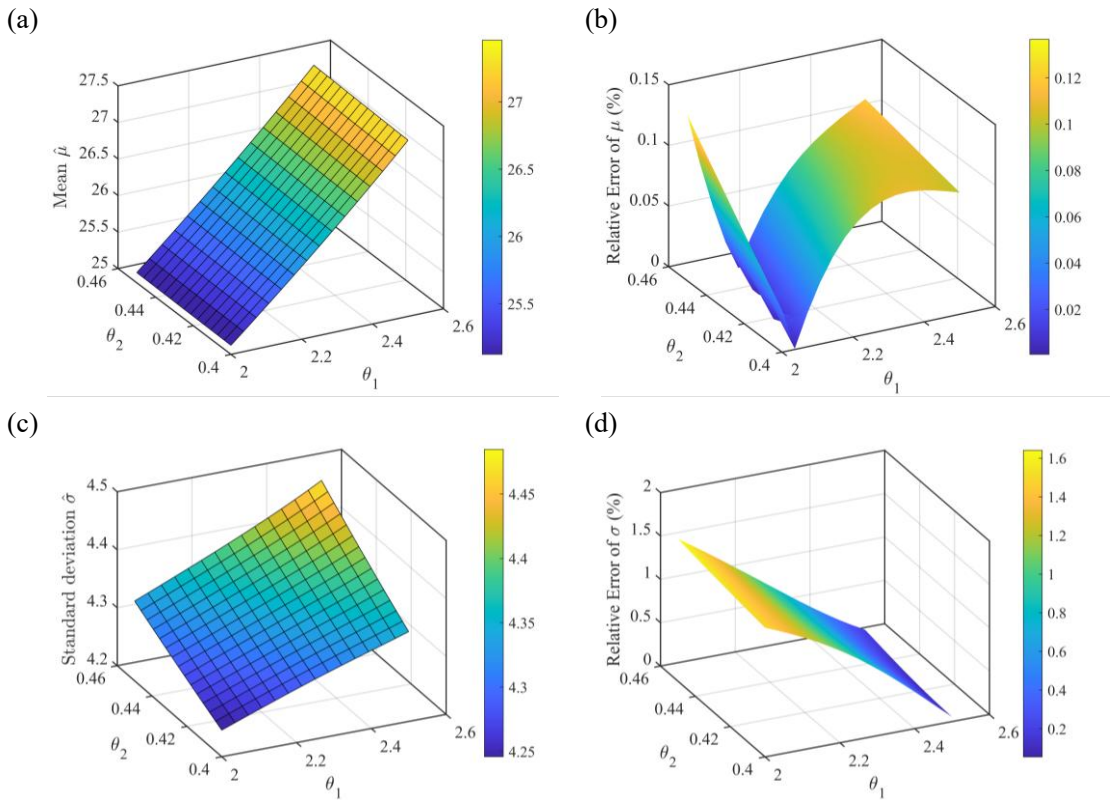
Table 1 Mean and standard deviation bounds of output responses by several methods.

Methods	$\hat{\mu}^L$	$\hat{\mu}^U$	$N_{\text{Call}}$	$\hat{\sigma}^L$	$\hat{\sigma}^U$	$N_{\text{Call}}$
Exact result	20.80	32.26	—	3.61	5.07	—
MCS-BGO	20.80	32.03	$20 \times 10^6$	3.62	5.04	$23 \times 10^6$
OSGNI	20.80	32.26	3763	3.61	5.07	3763
OU DRM	20.80	32.26	1643	3.61	5.07	1643
Proposed method	20.79	32.14	480	3.66	5.08	480

439

Note:  $N_{\text{call}}$  refers to the number of response function evaluations required.

440



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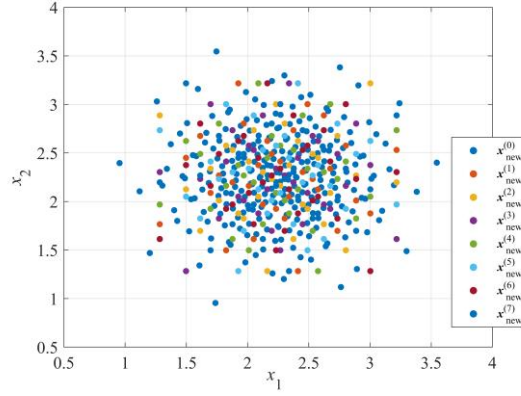
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Fig. 6 Surfaces of response mean and standard deviation using the proposed method: (a) and (c), and their errors relative to exact results: (b) and (d).

446 Fig. 7 illustrates the evolution of the representative points during the point-selection iterations. The  
 447 markers with different colors correspond to the initial Sobol batch of  $N_0 = 256$  and the **subsequent seven**  
 448 **adaptive successive** batches  $\mathbf{x}_{\text{new}}^{(i)}$  ( $i = 1, \dots, 7$ ). As the iteration proceeds, the algorithm automatically  
 449 inserts more points in the high-probability region around the mean of  $X_i$ , which explains the good  
 450 accuracy obtained with a relatively small total number of model evaluations. The distribution of  
 451 representative points gradually becomes more uniform as the representative set is enriched.



452  
 453 Fig. 7 Adaptive iterations of representative points in the input subspace  $(X_1, X_2)$  with  $N_0 = 256$  and  $N_{\text{grow}} = 32$ .

## 454 **Example 2: A cantilever tube**

455 To further verify the accuracy of the proposed method for structural application, a cantilever **tube**  
 456 modified from Refs. [18,21] is analyzed, as illustrated in Fig. 8. The structure involves eight random  
 457 variables, as listed in Table 2. The geometrical parameters are  $L_1 = 115.75$  mm and  $L_2 = 56.75$  mm.

458 The response of **interest** is the maximum von **Mises** stress of the **tube**, defined as:

$$459 \quad \sigma_{\text{max}} = \sqrt{\sigma_x^2 + 3\tau_{zx}^2}, \quad (28)$$

460 in which  $\sigma_x$  and  $\tau_{zx}$  denote the normal stress and shear stress, respectively, given by:

$$461 \quad \sigma_x = \frac{P + F_1 \sin \varphi_1 + F_2 \sin \varphi_2}{A} + \frac{Md}{2I}, \quad \tau_{zx} = \frac{Td}{4I}, \quad (29)$$

462 where  $A$ ,  $M$  and  $I$  are the cross-sectional area, bending moment, and moment of inertia, provided by:

$$463 \quad A = \frac{\pi}{4} [d^2 - (d - 2t_c)^2], \quad M = F_1 L_1 \cos \varphi_1 + F_2 L_2 \cos \varphi_2, \quad I = \frac{\pi}{64} [d^4 - (d - 2t_c)^4]. \quad (30)$$

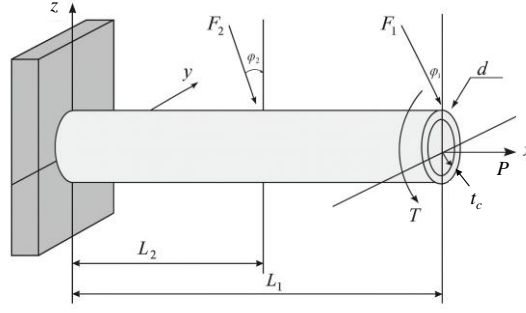


Fig. 8 A cantilever tube.

Table 2 Distribution types and parameters of input uncertain variables for cantilever tube.

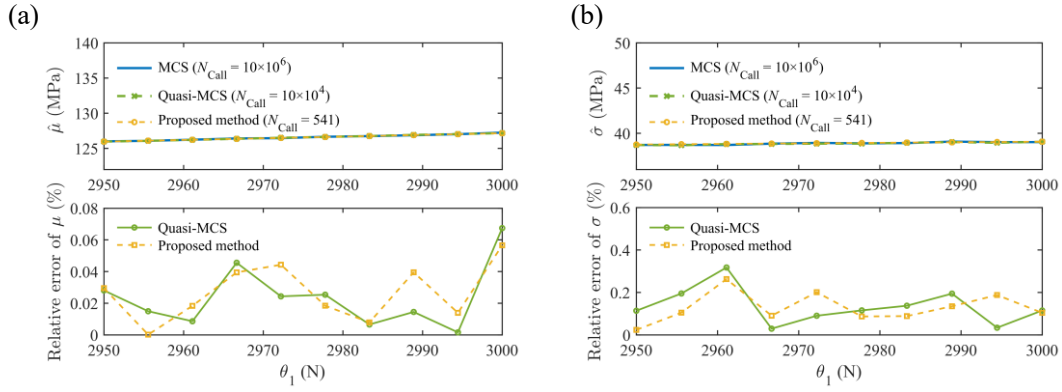
Variables	Distribution types	Parameter $\gamma^1$	Parameter $\gamma^2$
$P$ (N)	Uniform	10000	10100
$T$ (N · m)	Uniform	80	85
$\phi 1$ (°)	Normal	5	0.5
$\phi 2$ (°)	Normal	10	1.0
$F_1$ (N)	Lognormal	$\theta_1 \in [2950, 3000]$	$\theta_2 \in [295, 300]$
$F_2$ (N)	Lognormal	$\theta_3 \in [2950, 3000]$	$\theta_4 \in [295, 300]$
$t_c$ (mm)	Lognormal	$\theta_5 \in [5.0, 5.5]$	$\theta_6 \in [0.4, 0.6]$
$d$ (mm)	Lognormal	$\theta_7 \in [40, 45]$	$\theta_8 \in [5.0, 6.0]$

Note:  $\gamma^1$  and  $\gamma^2$  represent the mean and standard deviation of random variables for Normal and Lognormal distribution types, while they correspond to the minimum and maximum values for Uniform distribution in this paper.

The mean and standard deviation of the von Mises stress and their errors with respect to the MCS benchmark are presented in Fig. 9 and Fig. 10. The moment functions with respect to interval parameters are obtained by fixing the other intervals at their midpoint values. Table 3 summarizes the moment bounds estimated by MCS, Quasi-MCS-BGO (Quasi-MCS combined with Bayesian global optimization), OSGNI, OUDRM and the proposed method. The proposed method achieves approximately one order-of-magnitude improvement in numerical efficiency compared with the benchmark methods. The proposed method possesses strong potential as a practical and powerful tool for engineering uncertainty quantification. Two additional engineering examples are presented next to further validate the effectiveness of the proposed method. Notably, Fig. 9 shows the Quasi-MCS results with  $N_{\text{Call}} = 10^4$  yield satisfactory results. Considering the high computational cost of the following two complex examples, the Quasi-MCS results are adopted as the reference solutions for subsequent

480 validation.

481

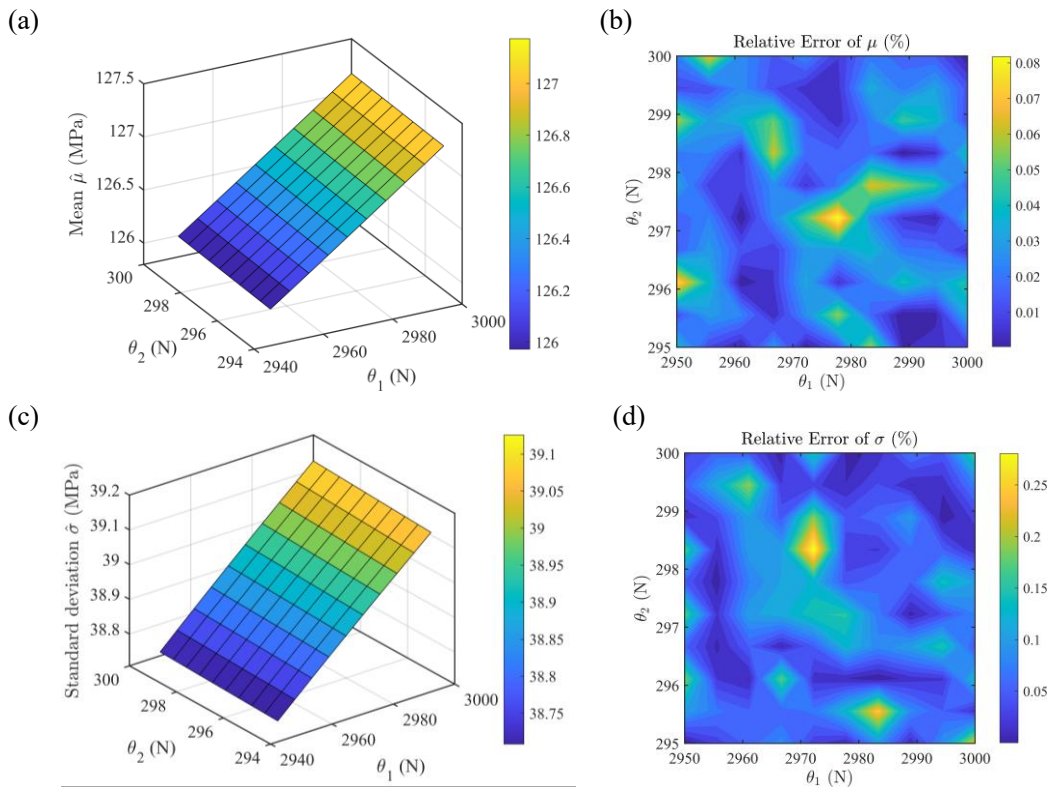


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Fig. 9 (a) Mean and (b) standard deviation curves and their error curves relative to MCS reference.

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Fig. 10 Mean and standard deviation by proposed method ( $N_{\text{Call}} = 572$ ): (a) and (c), and their errors relative to MCS reference with  $N_{\text{Call}} = 100 \times 10^6$ : (b) and (d).

490

Table 3 Mean and standard deviation bounds of von Mises stress obtained by several methods.

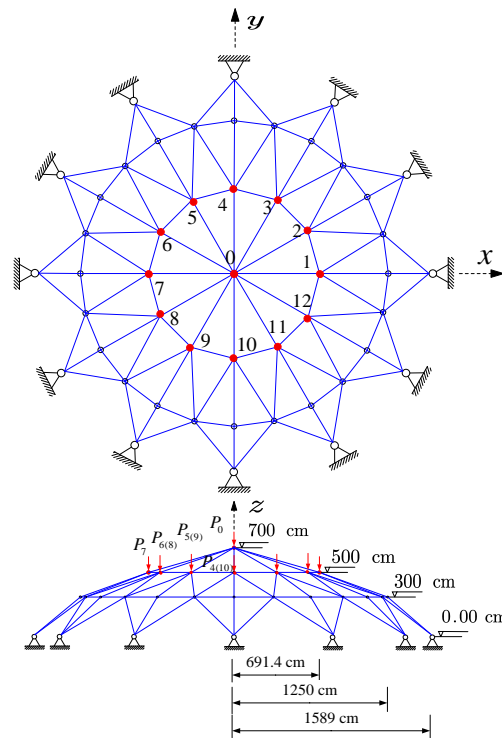
Methods	$\hat{\mu}^L$ (MPa)	$\hat{\mu}^U$ (MPa)	$N_{\text{Call}}$	$\hat{\sigma}^L$ (MPa)	$\hat{\sigma}^U$ (MPa)	$N_{\text{Call}}$
MCS-BGO	105.37	154.60	$14 \times 10^6$	27.77	55.63	$16 \times 10^6$
Quasi-MCS-BGO	105.48	154.26	$18 \times 10^4$	27.66	55.63	$27 \times 10^4$
OSGNI	105.44	154.41	11592	27.52	55.83	10304
OU DRM	105.42	154.32	3528	27.27	54.96	3136
Proposed method	105.32	154.38	572	27.68	56.53	572

491 **Example 3: Spatial truss FE structure**

492 As shown in Fig. 11, a 120-bar spatial truss structure subjected to thirteen vertical concentrated loads  
 493 is considered [47]. The truss structure is modeled as a three-dimensional FE model using the software  
 494 framework OpenSees (<https://opensees.berkeley.edu>). The FE model consists of 49 nodes and 120  
 495 elements. All elements are assumed to have the same cross-sectional area  $A$  and Young’s modulus  $E$ .  
 496 Thirteen static vertical concentrated loads  $P_0, P_1, \dots, P_{12}$  (kN) are applied to nodes 0–12 (integer values).  
 497 The distribution types and parameters of input uncertain variables are displayed in Table 4.

498 The response of interest is the vertical displacement of node. The corresponding response function is:

499 
$$g = V_1(E, A, P_0, P_1, \dots, P_{12}) . \tag{31}$$



500  
 501 Fig. 11 Top and side views for a 120-bar space truss structure.

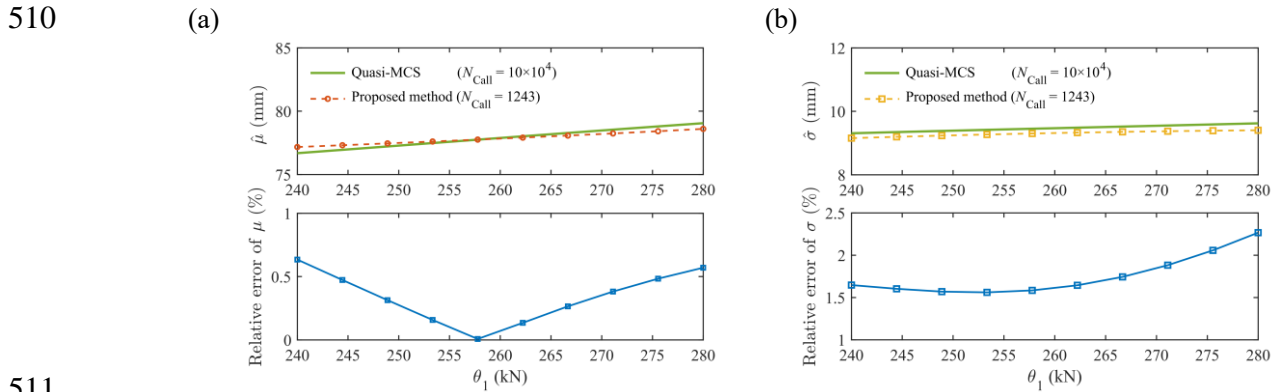
502 As shown in Fig. 12, Fig. 13 and Table 5, the estimated moments of the vertical displacement at node  
 503 0, as well as their bounds, are in close agreement with the Quasi-MCS benchmark. This case study  
 504 verifies the accuracy and efficiency of the proposed method for complex FE structures through  
 505 comparison with several existing approaches. From Table 5, the number of representative points  
 506 required by the proposed approach increases as the stopping indicator  $\beta$  decreases from 0.01 to 0.008 in

507 the iterative criterion, resulting in improved accuracy.

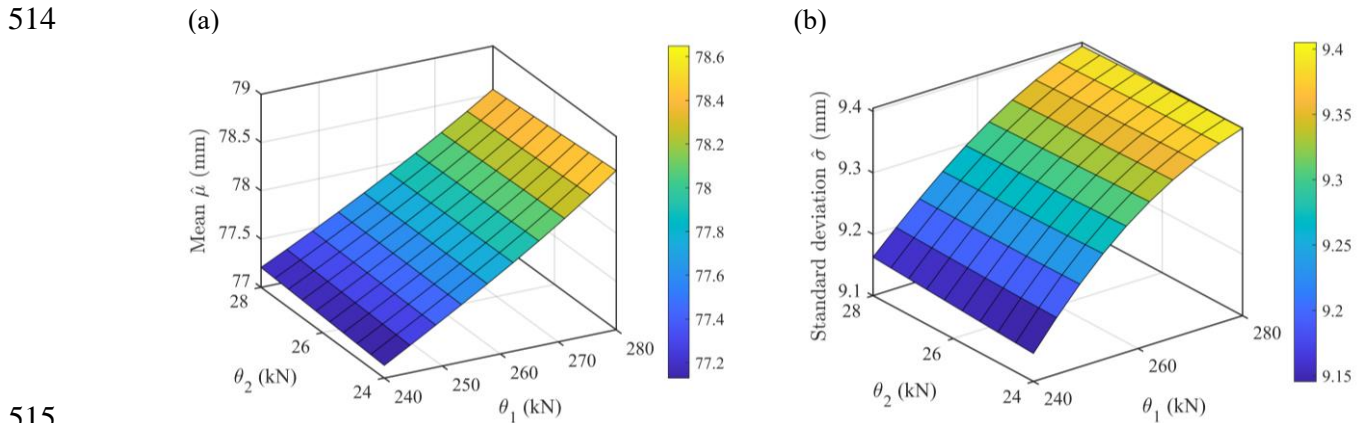
508 Table 4 Distribution information of input uncertainty variables for truss structure.

Random variable	Distribution	Parameter $\gamma^1$	Parameter $\gamma^2$
$E$ (GPa)	Gamma	180	200
$A$ (mm <sup>2</sup> )	Uniform	1600	1800
$P_{1(2,4,5,7,8,10,11)}$ (kN)	Lognormal	250	25
$P_0$ (kN)	Lognormal	$\theta_1 \in [240, 280]$	$\theta_2 \in [24, 28]$
$P_3$ (kN)	Lognormal	$\theta_3 \in [240, 280]$	$\theta_4 \in [24, 28]$
$P_6$ (kN)	Lognormal	$\theta_5 \in [240, 280]$	$\theta_6 \in [24, 28]$
$P_9$ (kN)	Lognormal	$\theta_7 \in [240, 280]$	$\theta_8 \in [24, 28]$
$P_{12}$ (kN)	Lognormal	$\theta_9 \in [240, 280]$	$\theta_{10} \in [24, 28]$

509 Note:  $\gamma^1$  and  $\gamma^2$  represent the mean and standard deviation of random variables for Gamma distribution type.



511 Fig. 12 (a) Means and (b) standard deviations of vertical displacements using the proposed method and errors  
512 relative to Quasi-MCS reference.  
513



515 Fig. 13 3D surfaces of displacement (a) mean and (b) standard deviation by the proposed method with  $N_{Call} = 1243$ .  
516

517 In terms of model evaluations, Quasi-MCS requires  $1.8 \times 10^5$  and  $1.9 \times 10^5$  FE model calls for  
518 estimating the mean and standard deviation, respectively, whereas the proposed approach requires only

519  $1.24 \times 10^3$  evaluations for both quantities. This corresponds to nearly two orders of magnitude reduction  
520 ( $\approx 99.3\%$  fewer model calls). **Different Quasi-MCS sample sizes are used for the mean and standard**  
521 **deviation because their optimization objectives correspond to different functions over the interval**  
522 **parameter domain, which exhibit different convergence behaviors.** The OSGNI and OUDRM do not  
523 provide complete moment functions and require 20244 and 7644 model evaluations, **while** the proposed  
524 method reduces the computational cost by approximately 94% and 84% relative to these two methods.  
525 These results demonstrate the efficiency and robustness of the proposed approach for hybrid-uncertainty  
526 analysis of complex structural systems.

527 Table 5 Mean and standard deviation bounds of vertical displacements obtained by several methods.

Methods	$\hat{\mu}^L$ (mm)	$\hat{\mu}^U$ (mm)	$N_{\text{Call}}$	$\hat{\sigma}^L$ (mm)	$\hat{\sigma}^U$ (mm)	$N_{\text{Call}}$
Quasi-MCS-BGO	75.14	80.88	$18 \times 10^4$	9.02	9.85	$19 \times 10^4$
OSGNI	75.19	80.80	20244	8.61	9.24	20244
OUDRM	75.21	80.81	7644	8.85	9.52	7644
Proposed method ( $\beta = 0.01$ )	76.43	79.58	1243	8.80	9.98	1243
Proposed method ( $\beta = 0.008$ )	76.40	79.71	1744	8.80	9.84	1744

#### 528 **Example 4: Offshore wind turbine tower**

529 As shown in Fig. 14, the monopile offshore wind turbine tower is idealized as a three degree-of-  
530 freedom (DOF) lateral shear-type nonlinear system supported by an elastic soil spring at the mudline  
531 [48]. The turbine is subjected to wind and wave loads at the top and lowest DOFs, respectively. The  
532 tower height is  $H = 90$  m with equal story **height  $h = H/3 = 30$  m**; rotor radius is  $R = 63$  m; elastic  
533 modulus is  $E = 210$  GPa; baseline bending rigidity is  $EI_0 = 3.0 \times 10^{10}$  N·m<sup>2</sup>. Six imprecise random  
534 variables and six precise random inputs are considered for the tower, as listed in Table 6.

535 The generalized displacements  $\mathbf{u}(t) = [u_1(t), u_2(t), u_3(t)]^T$  correspond to heights  $z = \{h, 2h, 3h\}$  along the  
536 tower. The response of interest is the peak displacement at **the** top, which is calculated by using the  
537 implicit Newmark integration scheme combined with Newton-Raphson iterations. The time domain is

538  $t \in [0, 180]$  with a constant time step of  $\Delta t = 0.01$  s .

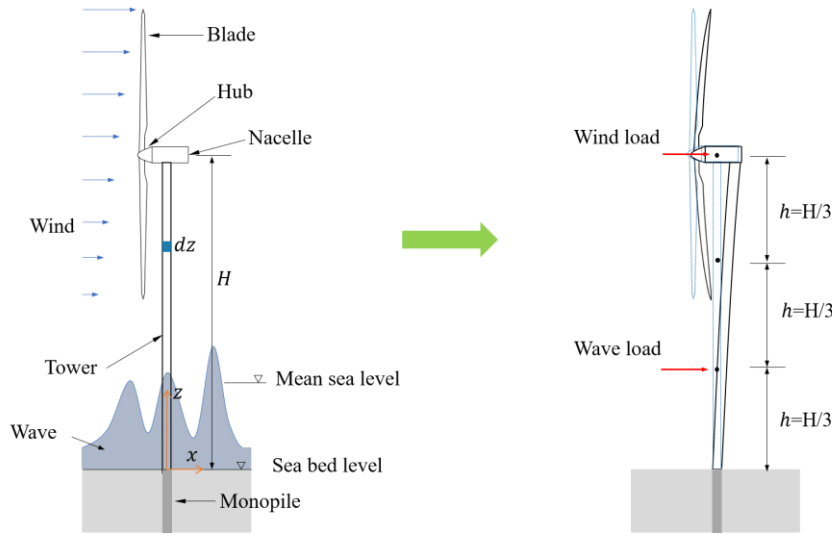
539 The nonlinear equation of motion governing the structural response is given by:

540 
$$\mathbf{M}\ddot{\mathbf{u}}(t) + \mathbf{C}\dot{\mathbf{u}}(t) + \mathbf{K}\mathbf{u}(t) + \mathbf{f}_{nl}(\mathbf{u}(t)) = \mathbf{F}(t), \quad (32)$$

541 where  $\mathbf{M}$ ,  $\mathbf{C}$  and  $\mathbf{K}$  are mass, stiffness, and damping matrices;  $\mathbf{f}_{nl}(\mathbf{u}(t))$  represents the nonlinear restoring

542 force;  $\mathbf{F}(t)$  denotes the environmental excitation. The detailed descriptions of these components are

543 provided in the Appendix B.



544  
545 Fig. 14 Configuration of the monopile offshore wind turbine tower and its equivalent simplified structural mode.

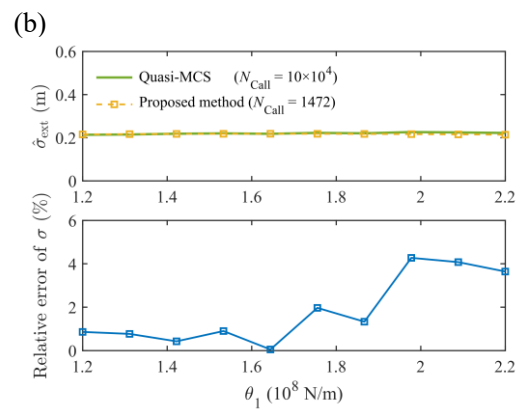
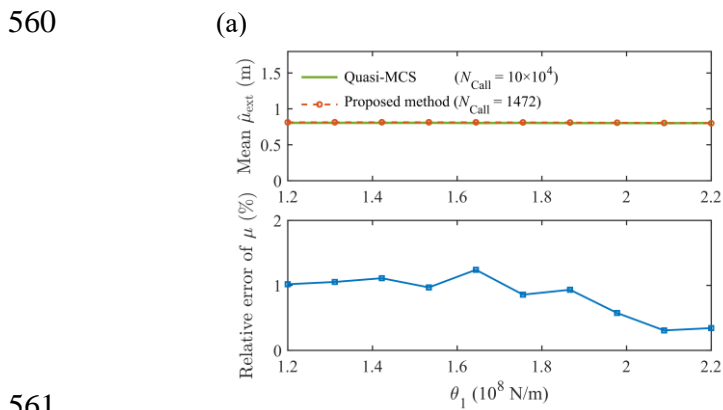
546 Table 6 Distribution information of input uncertainty variables.

Uncertain variables	Distribution types	Parameter $\gamma^1$	Parameter $\gamma^2$
$k_v$	Normal (truncated to $[0, +\infty)$ )	1.0	0.1
$H_s$ (m)	Lognormal	3.0	0.6
$f_p$	Uniform	0.08	0.14
$\Psi_w$	Uniform	0.90	1.10
$I_u$	Normal (truncated to $[0, +\infty)$ )	0.15	0.03
$\rho_{air}$ (kg/m <sup>3</sup> )	Normal (truncated to $[0, +\infty)$ )	1.225	0.05
$k_{soil}$ ( $\times 10^8$ N/m)	Gamma	$\theta_1 \in [1.2, 2.2]$	$\theta_2 \in [0.2, 0.6]$
$\zeta$	Gamma	$\theta_3 \in [0.015, 0.030]$	$\theta_4 \in [0.005, 0.010]$
$C_T$	Lognormal	$\theta_5 \in [0.80, 0.92]$	$\theta_6 \in [0.06, 0.10]$
$m_{nac}$ ( $\times 10^5$ kg)	Gamma	$\theta_7 \in [3.2, 3.8]$	$\theta_8 \in [0.3, 0.5]$
$m_{tow}$ ( $\times 10^5$ kg)	Gamma	$\theta_9 \in [1.9, 2.5]$	$\theta_{10} \in [0.2, 0.4]$
$\rho_s$ (kg/m <sup>3</sup> )	Gamma	$\theta_{11} \in [7800, 8050]$	$\theta_{12} \in [100, 200]$

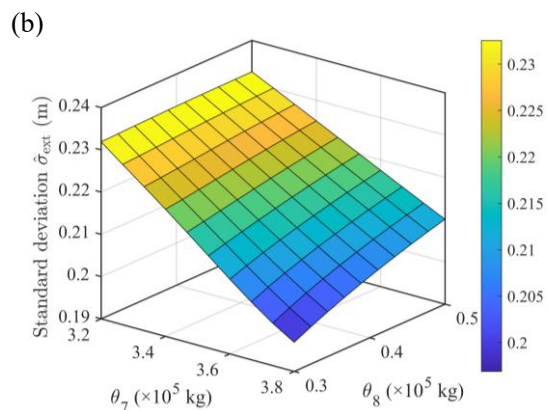
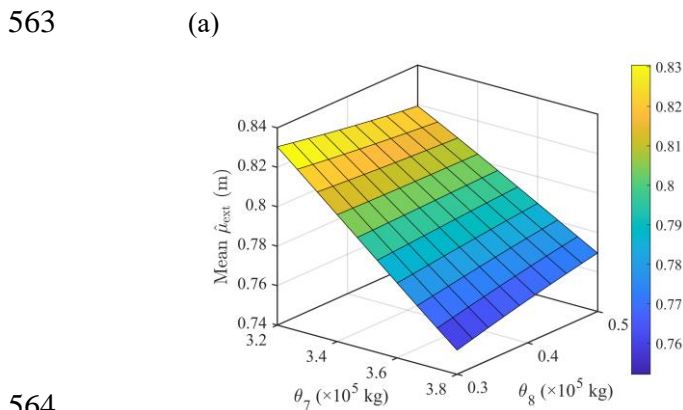
547 Note: the definitions of all symbols are provided in the Appendix.

548 To verify the accuracy of the proposed method for nonlinear dynamic problems, its results are  
 549 compared with the benchmark solutions obtained using Quasi-MCS, as shown in Fig. 15, Fig. 16 and  
 550 Table 7. The proposed adaptive SLRNI method provides **accurate and efficient** evaluation of the  
 551 response statistics for the nonlinear dynamic system. In contrast, OSGNI and OUDRM exhibit  
 552 noticeable deviations, **partly due to their reliance on gradient-based optimization and linearized**  
 553 **integration schemes**, as listed in Table 7. For nonlinear and time-dependent responses, these approaches  
 554 are susceptible to local convergence and numerical errors associated with low-order polynomial  
 555 approximations, which may lead to biased statistical estimates.

556 The adaptive SLRNI performs single-loop reweighted integration and eliminates the need for nested  
 557 optimization or sparse-grid discretization. This improves numerical stability, convergence behavior, and  
 558 computational efficiency, making the approach suitable for nonlinear and dynamic systems with strong  
 559 coupling effects.



561  
 562 Fig. 15 Mean and standard deviation curves of peak displacement at top and errors relative to Quasi-MCS reference.



564  
 565 Fig. 16 Moment surfaces of peak displacements by proposed approach ( $N_{\text{Call}} = 1472$ ): (a) mean; (b) standard deviation.

Table 7 Mean and standard deviation bounds of peak displacements by several methods.

Method	$\hat{\mu}_{\text{ext}}^L$ (m)	$\hat{\mu}_{\text{ext}}^U$ (m)	$N_{\text{Call}}$	$\hat{\sigma}_{\text{ext}}^L$ (m)	$\hat{\sigma}_{\text{ext}}^U$ (m)	$N_{\text{Call}}$
Quasi-MCS-BGO	0.664	1.018	$31 \times 10^4$	0.157	0.332	$27 \times 10^4$
OSGNI	0.537	0.965	39964	0.000	0.287	48500
OU DRM	0.683	1.384	18463	0.201	0.270	5994
Proposed method	0.666	0.980	1472	0.155	0.324	1472

## 567 5. Conclusions

568 To estimate response moment functions and their bounds for structures under parametric p-boxes,  
569 this study proposes an adaptive single-loop reweighted numerical integration (adaptive SLRNI)  
570 approach. Expressions for the response moments are derived from the generalized probability density  
571 integral equations (PDIEs) and reformulated through a reweighting strategy based on a complete  
572 auxiliary density. The resulting reweighted formulation is combined with an adaptive probability-  
573 equalized  $k$ -nearest-neighbor (APEK) point-selection technique to evaluate response moment functions  
574 and their bounds in a single-loop computational framework. The proposed framework is non-intrusive  
575 and applicable to black-box models and complex engineering systems. Four representative examples,  
576 including a mathematical problem, a cantilever tube, a 120-bar truss, and a nonlinear offshore wind  
577 turbine tower, are investigated to assess the performance of the proposed approach. The primary  
578 findings are summarized as follows.

579 (1) The proposed method provides accurate estimates of response moment functions and their bounds  
580 under hybrid uncertainties. For general structural systems, the errors in the estimated moments remain  
581 within approximately 2%, whereas those for finite element and nonlinear dynamic systems remain  
582 within about 6%. By contrast, some gradient-based approaches may converge to local optima for  
583 nonlinear structures, thereby leading to noticeable estimation errors.

584 (2) The proposed approach substantially reduces the number of structural model evaluations  
585 compared with existing methods. The adaptive SLRNI reduces model evaluations by approximately 99%  
586 compared with Quasi-MCS, and by roughly 80% and 90% relative to OSGNI and OU DRM, respectively.  
587 This improvement in efficiency, while maintaining comparable accuracy, demonstrates the

588 computational advantage of the proposed approach.

589 (3) The adaptive SLRNI incorporates an adaptive enrichment strategy and supports parallel  
590 computation. The APEK point-selection strategy generates uniformly distributed representative points  
591 with balanced probability weights. It enables accurate reweighted integration and allows incremental  
592 enrichment based on a prescribed convergence criterion.

593 Overall, the adaptive SLRNI is developed within the basis of the direct probability integral method,  
594 which provides a flexible and effective framework for hybrid aleatory-epistemic uncertainty  
595 quantification in advanced computational mechanics. The proposed approach is primarily intended for  
596 problems involving a moderate number of random and interval variables, since the performance of both  
597 the auxiliary density function and the APEK strategy may deteriorate in very high-dimensional settings.  
598 Future work will focus on developing more efficient adaptive point-selection strategies and  
599 incorporating surrogate modeling or active learning techniques to improve scalability. Additional  
600 research directions include extensions to reliability analysis, sensitivity analysis, reliability-based  
601 design optimization, and the quantification of instantaneous response probability density functions for  
602 dynamic systems under hybrid uncertainties. Extensions to dependent epistemic variables described by  
603 general convex constraint sets also constitute a promising direction for future research.

## 604 Declaration of competing interest

605 The authors declare that they have no known competing financial interests or personal relationships  
606 that could have appeared to influence the work reported in this paper.

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## 612 **Appendix A. Sensitivity of hyperparameters $K$ and $k$**

613 To examine the robustness of the proposed method with respect to the envelope multiplier  $K$  and  
614 number of nearest neighbors  $k$ , a sensitivity study is conducted for Example 1. The parameters are varied  
615 within the recommended integer range  $K, k \in \{6,7,8\}$ , and the resulting moment bounds and model  
616 evaluations are reported in Table A1.

617 **Table A1 Sensitivity analysis of hyperparameters  $K$  and  $k$  for the proposed method (Example 1).**

$K$	$k$	$\hat{\mu}^L$	$\hat{\mu}^U$	$\hat{\sigma}^L$	$\hat{\sigma}^U$	$N_{\text{Call}}$
6	8	20.79	32.14	3.66	5.08	480
6	7	20.84	32.15	3.62	5.13	542
6	6	20.84	32.17	3.64	5.09	542
7	6	20.84	32.17	3.64	5.09	542
8	6	20.84	32.17	3.64	5.09	542
8	7	20.84	32.15	3.62	5.13	542
8	8	20.80	32.14	3.66	5.08	480

618  
619 The results indicate negligible variations in the estimated mean and standard deviation bounds across  
620 the tested parameter combinations. The differences are mainly reflected in the number of model  
621 evaluations, confirming that the recommended range provides stable and reliable performance.

## 622 **Appendix B. Mass, stiffness, damping and loads for wind turbine tower**

### 623 **(1) Mass**

624 The total tower steel mass  $m_{\text{tow}}$  is evenly lumped to the three stories and nacelle mass  $m_{\text{nac}}$  is added at  
625 the top DOF. The mass matrix is given by:

$$626 \quad \mathbf{M} = \text{diag}(m_1, m_2, m_3), \quad m_1 = m_2 = \frac{m_{\text{tow}}}{3}, \quad m_3 = \frac{m_{\text{tow}}}{3} + m_{\text{nac}}.$$

### 627 **(2) Stiffness**

628 The linear stiffness is scaled to match the target fundamental frequency  $f_1^{\text{tgt}} = 0.3$  Hz:  
629  $\mathbf{K} = \gamma_K \mathbf{K}_p$ ,  $\gamma_K = (2\pi f_1^{\text{tgt}})^2 / \omega_{1,p}^2$ , where  $\mathbf{K}_p$  starts from a tridiagonal prototype with story stiffness  $k_{\text{story}}$  and  
630 the soil spring  $k_{\text{soil}}$  at the base:

631

$$\mathbf{K}_p = \begin{bmatrix} k_{\text{soil}} + k_{\text{story}} & -k_{\text{story}} & 0 \\ -k_{\text{story}} & 2k_{\text{story}} & -k_{\text{story}} \\ 0 & -k_{\text{story}} & k_{\text{story}} \end{bmatrix}.$$

632 **(3) Damping**

633 The damping uses Rayleigh form  $\mathbf{C} = \alpha_M \mathbf{M} + \beta_K \mathbf{K}$ . The coefficients  $\{\alpha_M, \beta_K\}$  are determined by  
 634 matching the **epistemically uncertain modal** damping ratio  $\zeta$  at the 1st and 3rd modes with natural  
 635 frequencies  $\{\omega_A, \omega_B\}$ :

636

$$\begin{bmatrix} \frac{1}{2\omega_A} & \frac{\omega_A}{2} \\ \frac{1}{2\omega_B} & \frac{\omega_B}{2} \end{bmatrix} \begin{bmatrix} \alpha_M \\ \beta_K \end{bmatrix} = \begin{bmatrix} \zeta \\ \zeta \end{bmatrix}.$$

637 **(4) Nonlinear duffing restoring at the top DOF is adopted:**

638

$$f_{\text{nl}}(\mathbf{u}) = [0, 0, \alpha_3 u_3^3]^T, \quad \alpha_3 \geq 0 (\text{N/m}^3),$$

639 where  $\alpha_3 = r K_{33} / u_{\text{ref}}^2$  and set  $r = 0.2$ ,  $u_{\text{ref}} = 0.30$  m.640 **(5) Environmental loads**641 • Wind load at hub (DOF 3):  $F_3(t)$ .

642 The instantaneous aerodynamic thrust is modeled as mean plus turbulent fluctuation:

643

$$F_3(t) = T_{\text{mean}} [1 + I_u u_\tau(t)], \quad T_{\text{mean}} = \frac{1}{2} C_T \rho_{\text{air}} A_{\text{rot}} V_{\text{mean}}^2,$$

644 where  $T_{\text{mean}}$  is mean thrust;  $I_u$  is turbulence intensity;  $u_\tau(t)$  is zero-mean, unit-standard-deviation  
 645 stochastic fluctuation obtained by low-pass filtering white noise;  $C_T$  is thrust coefficient;  $\rho_{\text{air}}$  is air  
 646 density;  $A_{\text{rot}} = \pi R^2$  is rotor swept area;  $V_{\text{mean}} = 10$  m/s is reference mean wind speed.

647 • Wave load at the lowest DOF (DOF 1):  $F_1(t)$ 

648 Wave load is represented by a lateral Morison-type force by a narrowband random process  
 649 synthesized as a sum of random-phase sinusoids:

650

$$F_1(t) = \frac{A(H_s, \Psi_w)}{N_h} \sum_{k=1}^{N_h} \sin(2\pi f_k t + \Phi_k), \quad f_k \sim N(f_p, (\gamma f_p)^2), \quad \Phi_k \sim U[0, 2\pi],$$

651 where  $H_s$  is the significant wave height and  $\Psi_w$  is an amplitude factor;  $N_h = 20$  is the number of

652 synthesized harmonics;  $f_k$  and  $\Phi_k$  are frequency and phase, which are normally and uniformly  
653 distributed, respectively;  $f_p$  is wave frequency for narrowband synthesis.  $\gamma = 0.2$  is the spectral  
654 bandwidth parameter.

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