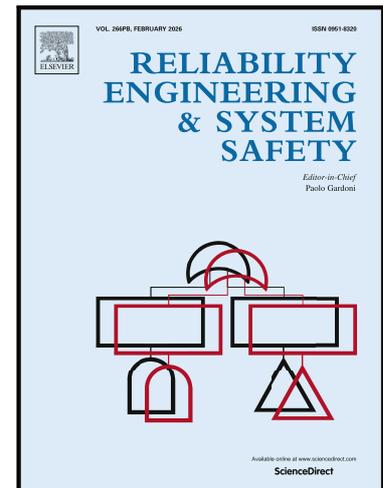


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1 First Excursion Probability Sensitivity in Stochastic Linear Dynamics 2 by means of Multidomain Line Sampling

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15 Abstract

16 This contribution presents a novel framework for estimating the sensitivity of first excursion
17 probabilities. The focus is on linear dynamic systems with non-proportional damping subject
18 to stationary or non-stationary Gaussian excitation. The sensitivity is estimated with respect
19 to structural parameters of the system, including material properties and geometric dimensions
20 of the elements. In dynamical systems, calculating both the first excursion probability and its
21 sensitivity is done in a high-dimensional space, making the task challenging and computationally
22 expensive. In this regard, the multidomain Line Sampling framework exploits linearity to obtain
23 sensitivity estimates as a byproduct of the first excursion probability evaluation. The results show
24 that the presented technique is highly efficient compared to different methods in the literature, as
25 demonstrated through two numerical examples involving small- and large-scale models.

26 *Keywords:* First excursion probability, Sensitivity analysis, Gaussian loading, Linear systems,
27 Multidomain Line Sampling

28 Highlights:

- 29 • Sensitivity of failure probability estimated using multidomain Line Sampling
- 30 • Sensitivity is a byproduct of the first excursion probability
- 31 • Sensitivities of eigenvalues and eigenvectors required
- 32 • Non-proportional damping considered in the formulation

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1. Introduction

The development of stochastic models has become paramount to achieving a better understanding of the behavior of mechanical and structural systems. Stochastic representation of external loads provides a more realistic depiction of real-life phenomena. For example, stochastic processes offer more accurate representations of events like earthquakes and wind loads in structural systems, as well as vibrations and impact loads in mechanical systems. It is also noted that, in many practical cases and applications, the system reliability is dominated by the uncertainty in the excitation as commonly assumed in structural vibration serviceability, wind engineering and earthquake engineering applications (see e.g. [1] and [2]). In the context of dynamical systems, the theory of random vibrations introduces the concept of the so-called first excursion probability, which measures the chance that one or more responses of interest exceed a prescribed threshold during stochastic excitation [1]. This metric is particularly useful as a measure of serviceability [3, 4, 5] when the system behavior remains linear during stochastic excitation, allowing the response of interest to be calculated while materials operate within their linear range. To accurately estimate the first excursion probability while taking advantage of system linearity, several advanced simulation techniques have been developed considering the loading as a Gaussian process and deterministic structural models, including Efficient Importance Sampling (EIS) [6], the Domain Decomposition Method (DDM) [7], Directional Importance Sampling (DIS) [8], and multidomain Line Sampling (mLS) [9].

Although the information provided by the first excursion probability is essential for assessing system reliability [10, 11, 12, 13, 14], it is also important to study its sensitivity to possible changes in system parameters. This problem has been widely discussed in the literature [15, 16, 17, 18, 19], and two main research branches can be identified for the sensitivity of the first excursion probability [20]. The first group studies sensitivity with respect to distribution parameters of random variables [21, 22]. These parameters can include, for instance, the mean value or standard deviation of a given random variable representing uncertainty in, e.g. loading. The second group, which is the focus of this work, studies sensitivity with respect to deterministic system parameters. Methods that consider second-order moment approximations [23], combinations of stochastic simulation with Bayes' theorem for sensitivity estimates [24, 25], and the combination of stochastic simulation with local approximations of the responses of interest [26, 27, 28], are various known approaches within this group. In practical applications, system parameters such as the geomet-

64 ric dimensions of structural members or their elastic properties are subject to uncertainty (e.g.,
65 manufacturing tolerances), but they are still specified and controlled by the designer through
66 nominal values. For the systems considered here, the uncertainty due to the stochastic loading
67 is dominant, so modelling the structural parameters by their nominal values is a reasonable as-
68 sumption. Accordingly, this work explicitly considers local (derivative-based) sensitivities, which
69 provide valuable information on the magnitude of change at the nominal design, whereas global
70 sensitivities attribute output variance to uncertain inputs and require probabilistic models for the
71 parameters, a complementary perspective not pursued here. In this context, sensitivity is calcu-
72 lated as the partial derivative of the first excursion probability with respect to a system parameter.
73 This gradient calculation involves solving an integral in a high-dimensional space, which usually
74 has no closed-form solution.

75 Despite the fact that linear structural models under Gaussian loading constitute a classical
76 setting in stochastic dynamics, the estimation of the first excursion probability sensitivities in a
77 high-dimensional and non-stationary context remains computationally challenging and cannot be
78 addressed with available analytical approaches. In most cases, sensitivity analysis is carried out as
79 a post processing step that reuses the samples originally generated for reliability estimation [16,
80 20, 22]. Although methods such as DIS [9] and DDM [29] can compute reliability and sensitivities
81 from the same set of model evaluations for this class of systems, their sensitivity estimators
82 often exhibit high variance and the associated computational cost remains substantial. DIS and
83 DDM are directional sampling methods [30, 31] that explore the standard Gaussian space along
84 directions anchored at the origin. Line Sampling methods take a different approach, evaluating
85 lines that need not pass through the origin, which provides an alternative way of exploring the
86 space. One such method is mLS, whose efficiency for first excursion probability estimation has
87 been demonstrated, and extending this approach to sensitivity estimation constitutes a natural
88 and promising next step. Accordingly, improving sensitivity estimation is crucial, given that
89 derivative information has broad applications in model calibration, risk evaluation, and risk-
90 informed decision-making [10, 32, 33, 34] and, together with the first excursion probability, in
91 reliability-based design optimization [34, 35, 2] and network applications [36].

92 In this work, an extension of the mLS framework for calculating the sensitivity of the first
93 excursion probability is proposed. The methodology allows for the calculation of sensitivity esti-
94 mates using mLS for both reliability and sensitivity analyses, and it is applicable to small- and

95 large-scale problems. The scope of the method includes linear systems subject to Gaussian load-
 96 ing, and the systems considered involve non-proportional damping. The latter consideration offers
 97 a more generalized methodology than the one proposed in [28], and is numerically more efficient
 98 than the methods proposed in [37, 29]. The consideration of non-proportional damping allows a
 99 closer representation of the modeled system response to real-world behavior [38]. Additionally, the
 100 special structure of the failure domain is exploited by mLS, which efficiently explores the failure
 101 domain to produce accurate and computationally inexpensive reliability and sensitivity estimates.
 102 The novelty of the proposed sensitivity extension of mLS is linked to the way exploration lines are
 103 placed relative to the elementary failure domains. Since the lines are not forced to pass through
 104 the origin and are guided by the elementary domains, they often intersect the corresponding limit-
 105 state boundary in a near-perpendicular way, which is more informative for sensitivity estimates.
 106 Directional methods such as DIS and DDM, in contrast, use rays originating at the origin, which
 107 can intersect the boundary at less informative angles and therefore require more simulations to
 108 achieve the same accuracy in sensitivity estimates. As a result, the proposed approach achieves
 109 a notable computational efficiency. Moreover, the sensitivity analysis of the spectral properties
 110 of the system is required [39], and sensitivity estimates are achieved as a byproduct of the first
 111 excursion probability estimator.

112 The rest of this work is organized as follows. The problem definition, together with the first
 113 excursion probability and its gradient, is presented in Section 2. Sections 3 and 4 cover the
 114 calculation of the first excursion probability and its gradient using mLS, respectively. Numerical
 115 implementation aspects are detailed in Section 5. The effectiveness of the method, including
 116 quantitative comparisons that highlight its technical advantages with respect to DIS and DDM, is
 117 demonstrated through two examples in Section 6, and finally, conclusions and future developments
 118 are discussed in Section 7.

119 2. Problem Statement

120 2.1. Gaussian Loading

121 The dynamic load, \mathbf{p} , applied to the system is modeled as a discrete Gaussian process of
 122 duration T , sampled at n_T time steps, each of duration Δt . The k -th time instant is defined as
 123 $t_k = (k - 1)\Delta t$ for $k = 1, \dots, n_T$. The expected value of this process at time t_k is denoted by μ_k ,
 124 the k -th element of the expected value vector $\boldsymbol{\mu}$, which has dimensions $n_T \times 1$. The associated

125 covariance matrix, Σ , is symmetric, bounded, and positive definite, with the covariance between
 126 t_{k_1} and t_{k_2} given by Σ_{k_1, k_2} , the (k_1, k_2) -th element of Σ . Using the Karhunen-Loève expansion
 127 [40, 41], the elements contained in the vector of the dynamic load \mathbf{p} are represented as:

$$p(t_k, \mathbf{z}) = \mu_k + \boldsymbol{\psi}_k^T \mathbf{z}, \quad k = 1, \dots, n_T, \quad (1)$$

128 where $p(t_k, \mathbf{z})$ is the loading at time t_k , and \mathbf{z} is a realization of a standard Gaussian random
 129 vector \mathbf{Z} with dimensions $n_{KL} \times 1$, where n_{KL} is the truncation order of the expansion ($n_{KL} \leq$
 130 n_T). Solving the eigenproblem $\Sigma \Xi = \Xi \Lambda$ for the largest n_{KL} eigenvalues of Σ , the matrix
 131 $\Psi = [\boldsymbol{\psi}_1, \boldsymbol{\psi}_2, \dots, \boldsymbol{\psi}_{n_T}]$ is obtained as $\Psi = \Lambda^{1/2} \Xi^T$, where each $\boldsymbol{\psi}_k$ (of dimensions $n_{KL} \times 1$) is the
 132 K-L (Karhunen-Loève) vector associated with the time instant t_k . Without loss of generality, this
 133 work assumes $\boldsymbol{\mu} = \mathbf{0}$.

134 In view of this discrete formulation and time-varying excitation, classical stationary-process
 135 tools (e.g., Rice's upcrossing formulas and diffusion-based Fokker-Planck equations) are not ap-
 136 plicable in the present setting.

137 2.2. Structural System

138 The structural system under study is linear, elastic, and damped, and it is subject to Gaussian
 139 loading $p(t, \mathbf{z})$. Considering n_D degrees of freedom, the system dynamics are governed by the
 140 following equation of motion [42]:

$$\mathbf{M}(\mathbf{y})\ddot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) + \mathbf{C}(\mathbf{y})\dot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) + \mathbf{K}(\mathbf{y})\mathbf{x}(t, \mathbf{y}, \mathbf{z}) = \mathbf{g}(\mathbf{y})p(t, \mathbf{z}), \quad t \in [0, T], \quad (2)$$

141 where \mathbf{x} , $\dot{\mathbf{x}}$, and $\ddot{\mathbf{x}}$ are the displacement, velocity, and acceleration vectors (dimensions $n_D \times 1$),
 142 and \mathbf{M} , \mathbf{C} , and \mathbf{K} are the mass, damping, and stiffness matrices of dimensions $n_D \times n_D$. The
 143 load coupling vector, \mathbf{g} , also has dimensions $n_D \times 1$, while the vector \mathbf{y} , with dimensions $n_Y \times 1$,
 144 contains the so-called structural design parameters. This vector groups parameters of the system
 145 under study which can be modified, such as geometric dimensions of beams. In the present work,
 146 \mathbf{y} is treated as a deterministic vector of nominal design values controlled by the designer, while
 147 the uncertainty is represented solely by the stochastic excitation. The alternative formulation, in
 148 which uncertainties in these structural parameters are modelled explicitly by random variables and
 149 their distribution parameters, corresponds to the first branch of sensitivity methods discussed in
 150 Section 1 and is not further pursued here. Therefore, to account for systems with non-proportional

151 damping, Equation (2) can be reformulated in an augmented form [43]:

$$\underbrace{\begin{bmatrix} \mathbf{0}_{n_D \times n_D} & \mathbf{M}(\mathbf{y}) \\ \mathbf{M}(\mathbf{y}) & \mathbf{C}(\mathbf{y}) \end{bmatrix}}_{[\mathbf{M}_a(\mathbf{y})]} \underbrace{\begin{Bmatrix} \ddot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) \\ \dot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) \end{Bmatrix}}_{\{\dot{\mathbf{q}}(t, \mathbf{y}, \mathbf{z})\}} + \underbrace{\begin{bmatrix} -\mathbf{M}(\mathbf{y}) & \mathbf{0}_{n_D \times n_D} \\ \mathbf{0}_{n_D \times n_D} & \mathbf{K}(\mathbf{y}) \end{bmatrix}}_{[\mathbf{K}_a(\mathbf{y})]} \underbrace{\begin{Bmatrix} \dot{\mathbf{x}}(t, \mathbf{y}, \mathbf{z}) \\ \mathbf{x}(t, \mathbf{y}, \mathbf{z}) \end{Bmatrix}}_{\{\mathbf{q}(t, \mathbf{y}, \mathbf{z})\}} = \underbrace{\begin{Bmatrix} \mathbf{0}_{n_D \times 1} \\ \mathbf{g}(\mathbf{y}) \end{Bmatrix}}_{\{\mathbf{g}_a(\mathbf{y})\}} p(t, \mathbf{z}), \quad (3)$$

152 or more compactly as:

$$[\mathbf{M}_a(\mathbf{y})] \{\dot{\mathbf{q}}(t, \mathbf{y}, \mathbf{z})\} + [\mathbf{K}_a(\mathbf{y})] \{\mathbf{q}(t, \mathbf{y}, \mathbf{z})\} = \{\mathbf{g}_a(\mathbf{y})\} p(t, \mathbf{z}), \quad (4)$$

153 where $\{\mathbf{q}\}$ is a vector that contains both system velocities and displacements, with dimensions
 154 $2n_D \times 1$; $[\mathbf{M}_a]$ and $[\mathbf{K}_a]$ are the augmented mass and stiffness matrices, each with dimensions
 155 $2n_D \times 2n_D$; and $\{\mathbf{g}_a\}$ is the augmented load coupling vector, with dimensions $2n_D \times 1$. Note that,
 156 within the scope of this work, the number of degrees of freedom n_D is large (e.g., in the order of
 157 tens of thousands). As a consequence, diffusion-theory approaches based on the Fokker-Planck-
 158 Kolmogorov (FPK) equation are not feasible for the state-space formulation in Equation (4),
 159 since they would require solving a prohibitively high-dimensional PDE in the $2n_D$ -dimensional
 160 state space.

161 Key dynamic responses, such as displacements, accelerations, internal stresses, or their linear
 162 combinations, are represented by $\eta_i(t, \mathbf{y}, \mathbf{z})$ for $i = 1, \dots, n_\eta$, and can be computed using the
 163 convolution integral [42]:

$$\eta_i(t, \mathbf{y}, \mathbf{z}) = \int_0^t h_i(t - \tau, \mathbf{y}) p(\tau, \mathbf{z}) d\tau, \quad i = 1, \dots, n_\eta, \quad (5)$$

164 where $h_i(t, \mathbf{y})$ is the unit impulse response function of the i -th response. Here, $p(t, \mathbf{z})$ refers to
 165 the Gaussian load, and zero initial conditions are assumed, i.e., $\mathbf{x}(0, \mathbf{y}, \mathbf{z}) = \dot{\mathbf{x}}(0, \mathbf{y}, \mathbf{z}) = \mathbf{0}_{n_D \times 1}$.
 166 Note that solving Equation (4) through modal decomposition enables expressing the response of
 167 interest as a combination of the vector $\mathbf{q}(t, \mathbf{y}, \mathbf{z})$, such that $\eta_i(t, \mathbf{y}, \mathbf{z}) = \boldsymbol{\gamma}_i^T \mathbf{q}(t, \mathbf{y}, \mathbf{z})$, where $\boldsymbol{\gamma}_i$ is a
 168 constant vector with dimensions $2n_D \times 1$. Furthermore, as depicted in e.g., [44], the unit impulse
 169 response function of the i -th response of interest is expressed as:

$$h_i(t, \mathbf{y}) = \sum_{r=1}^{2n_D} \frac{\boldsymbol{\gamma}_i^T \boldsymbol{\phi}_r(\mathbf{y}) \boldsymbol{\phi}_r(\mathbf{y})^T \mathbf{g}_a(\mathbf{y})}{\boldsymbol{\phi}_r(\mathbf{y})^T \mathbf{M}_a(\mathbf{y}) \boldsymbol{\phi}_r(\mathbf{y})} e^{\lambda_r(\mathbf{y})t}, \quad i = 1, \dots, n_\eta, \quad (6)$$

170 where $\phi_r(\mathbf{y})$ and $\lambda_r(\mathbf{y})$ represent the eigenvectors and eigenvalues, respectively, associated with
 171 Equation (4). Note that Equation (6) is only valid if \mathbf{C} is symmetric; otherwise, the right and left
 172 eigenvector problems need to be considered. In cases where n_D is large, model truncation [42] can
 173 be applied to Equation (6) to simplify the calculation or reduce computational effort by selecting
 174 a subset of eigenvectors and eigenvalues smaller than $2n_D$.

175 The Gaussian loading defined in Section 2.1 is discretized with respect to time. Consequently,
 176 the i -th response of interest at the time instant t_k can be approximated by discretizing the integral
 177 in Equation (5), as follows:

$$\eta_i(t_k, \mathbf{y}, \mathbf{z}) = \mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{z}, i = 1, \dots, n_\eta, k = 1, \dots, n_T, \quad (7)$$

178 where the vector $\mathbf{a}_{i,k}(\mathbf{y})$ of dimensions $n_{KL} \times 1$ is defined as:

$$\mathbf{a}_{i,k}(\mathbf{y}) = \sum_{m=1}^k \Delta t \epsilon_m h_i(t_k - t_m, \mathbf{y}) \boldsymbol{\psi}_m, \quad (8)$$

179 where the parameter ϵ_m is determined by the chosen integration scheme. For instance, in the case
 180 of the trapezoidal scheme, $\epsilon_m = 1/2$ when $m = 1$ or $m = k$, and $\epsilon_m = 1$ otherwise [45]. It is
 181 important to highlight that, in Equation (7), the vector \mathbf{z} is associated with the Gaussian load \mathbf{p} ,
 182 while the dependency on the design vector \mathbf{y} is exclusively linked to the structural model through
 183 the vector $\mathbf{a}_{i,k}$.

184 2.3. First Excursion Probability

185 The reliability of the system can be assessed by quantifying its performance. This is achieved
 186 by comparing the responses of interest, $\eta_i(t)$, $i = 1, \dots, n_\eta$, with the corresponding design re-
 187 quirements, b_i , $i = 1, \dots, n_\eta$, over time. This comparison is represented mathematically by the
 188 performance function $g(\mathbf{y}, \mathbf{z})$, which determines whether the response of interest exceeds a pre-
 189 scribed threshold during the action of the loading over time. The performance function is defined
 190 as follows:

$$g(\mathbf{y}, \mathbf{z}) = 1 - \max_{i=1, \dots, n_\eta} \left(\max_{k=1, \dots, n_T} \left(\frac{|\eta_i(t_k, \mathbf{y}, \mathbf{z})|}{b_i} \right) \right), \quad (9)$$

191 where $|\cdot|$ is the absolute value. The system is considered to fail when the performance function
 192 is negative, while a positive value indicates that the design requirements are met. Note that all
 193 possible realizations of the Gaussian vector \mathbf{z} that lead to failure, for a given design vector \mathbf{y} , define

194 the failure domain. This domain can be formally expressed as: $F = \{\mathbf{z} \in \mathbb{R}^{n_{KL}} : g(\mathbf{y}, \mathbf{z}) \leq 0\}$.

195 The so-called first excursion probability [1] quantifies the probability associated with the failure
196 domain and is given by:

$$p_F(\mathbf{y}) = \int_{g(\mathbf{y}, \mathbf{z}) \leq 0} f_{\mathbf{z}}(\mathbf{z}) d\mathbf{z}, \quad (10)$$

197 where $f_{\mathbf{z}}(\mathbf{z})$ represents the standard Gaussian probability density function in n_{KL} -dimensional
198 space.

199 The first excursion probability integral can be challenging to solve. Specifically, as n_{KL} in-
200 creases, the integral becomes high-dimensional and does not have a closed-form solution. In this
201 case, the integral must be estimated using advanced simulation techniques [46]. To address this,
202 within the context of this work, advanced simulation methods have been developed that take
203 advantage of the system linearity to estimate the first excursion probability [6, 7, 8, 9].

204 2.4. Geometry of the Failure Domain

205 The failure domain defined in Section 2.3 has a very special structure. Analyzing Equa-
206 tion (9), it is evident that it can be decomposed into $n_{\eta} n_T$ elementary failure domains. This
207 characteristic arises from the linearity of the system and the chosen representation method for
208 the Gaussian loading [6, 47]. The elementary failure domains, denoted as $F_{i,k}$, represent the
209 event where the response of interest η_i exceeds the design value b_i at the specific time instant
210 t_k . The subset that includes positive values of the response of interest exceeding b_i is defined
211 as $F_{i,k}^+ = \{\mathbf{z} \in \mathbb{R}^{n_{KL}} : \mathbf{a}_{i,k}^T(\mathbf{y})\mathbf{z} \geq b_i\}$. Similarly, the subset representing negative values of the
212 response of interest falling below $-b_i$ is defined as $F_{i,k}^- = \{\mathbf{z} \in \mathbb{R}^{n_{KL}} : \mathbf{a}_{i,k}^T(\mathbf{y})\mathbf{z} \leq -b_i\}$. The above
213 definitions allow the failure domain to be expressed as the union of the elementary failure domains:

$$F = \bigcup_{i=1}^{n_{\eta}} \bigcup_{k=1}^{n_T} F_{i,k}, \quad (11)$$

214 where $F_{i,k} = F_{i,k}^+ \cup F_{i,k}^-$. Furthermore, the limit state function related to the elementary failure
215 domain $F_{i,k}$ is denoted as $g_{i,k}$. In the same manner, the limit state function associated with
216 $F_{i,k}^+$ is denoted by $g_{i,k}^+$, whereas the limit state function associated with $F_{i,k}^-$ is denoted by $g_{i,k}^-$.
217 Figure 1 illustrates a representation of the elementary failure domains for the case where $n_{\eta} = 1$
218 and $n_T = n_{KL} = 2$. Note that the domains are constrained to the box boundaries, which is just
219 for drawing purposes, as in reality z_1 and z_2 cover all the real line. The domains $F_{1,1}$ and $F_{1,2}$ are
220 divided into their positive and negative parts, as schematically shown in Figure 1(b). Zones where

221 overlap occurs between the elementary failure domains can be identified, which indicates that the
 222 response of interest exceeds the threshold at both time instants simultaneously. Furthermore,
 223 although this is a bidimensional representation of the problem, the overlap between elementary
 224 failure domains can be significant, which provides insights into the high level of interaction that
 225 exists in a higher-dimensional case.

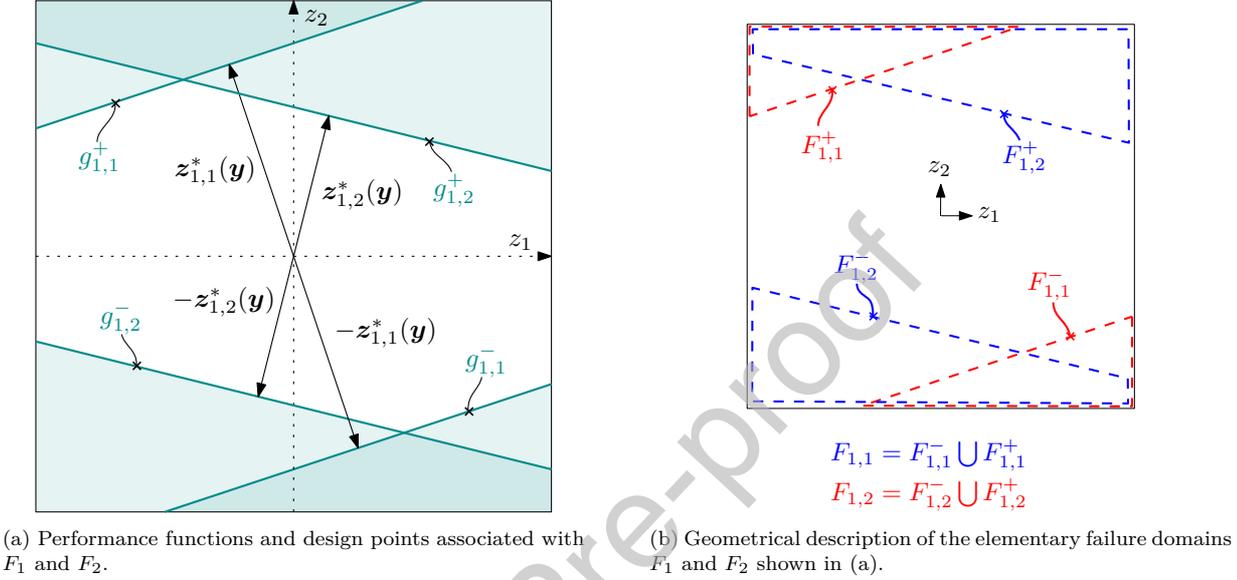


Figure 1: Elementary failure domains representation for the case where $n_\eta = 1$ and $n_T = n_{KL} = 2$.

226 The spatial definition of each elementary failure domain is given by the so-called design point
 227 [6, 47, 48]. This corresponds to the realization of \mathbf{z} within the elementary failure domain with the
 228 highest likelihood, or equivalently, to the point with the smallest Euclidean norm from the origin
 229 to a realization of \mathbf{z} within the elementary failure domain, which is given by [6, 47]:

$$\mathbf{z}_{i,k}^*(\mathbf{y}) = b_i \frac{\mathbf{a}_{i,k}(\mathbf{y})}{\|\mathbf{a}_{i,k}(\mathbf{y})\|}, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T, \quad (12)$$

230 where $\mathbf{z}_{i,k}^*(\mathbf{y})$ is the design point associated with the $F_{i,k}^+$ elementary failure domain, as illustrated
 231 in Figure 1a. The Euclidean norm of this vector is known as the reliability index, which is given
 232 by [6, 47]:

$$\beta_{i,k}(\mathbf{y}) = \frac{b_i}{\|\mathbf{a}_{i,k}(\mathbf{y})\|}, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T, \quad (13)$$

233 being the reliability index $\beta_{i,k}(\mathbf{y})$ associated with the positive $F_{i,k}^+$ and negative $F_{i,k}^-$ elemen-
 234 tary failure domains. Therefore, due to the linearity of the performance functions in the Gaus-
 235 sian space, calculating the probability of occurrence for each elementary domain is straightfor-

ward: $P[F_{i,k}^+] = P[F_{i,k}^-] = \Phi[-\beta_{i,k}]$, where $P[\cdot]$ denotes probability and $\Phi[\cdot]$ represents the one-dimensional Gaussian cumulative distribution function [6, 47]. An upper bound \hat{p}_F of the failure probability [6] is given by:

$$\hat{p}_F = \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} P[F_{i,k}] = 2 \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} \Phi(-\beta_{i,k}). \quad (14)$$

Please note that the upper bound \hat{p}_F would become equal to the failure probability only in the hypothetical case where there is no overlap between elementary failure domains. But this case is not observed in practical cases and hence, \hat{p}_F is an upper bound for p_F .

2.5. Gradient of the First Excursion Probability

The sensitivity of the first excursion probability can be assessed from Equation (10) due to its dependence on the design vector \mathbf{y} . One possible way to evaluate this sensitivity is by determining the gradient of the first excursion probability (as described in Appendix A):

$$\frac{\partial p_F(\mathbf{y})}{\partial y_q} = - \int_{g(\mathbf{y}, \mathbf{z})=0} \frac{\partial g(\mathbf{y}, \mathbf{z})}{\partial y_q} \frac{1}{\|\nabla_{\mathbf{z}} g(\mathbf{y}, \mathbf{z})\|} f_{\mathbf{z}}(\mathbf{z}) dS, \quad q = 1, \dots, n_Y, \quad (15)$$

where $\|\cdot\|$ denotes the Euclidean norm, $\nabla_{\mathbf{z}}$ is the nabla operator $\nabla_{\mathbf{z}} = [\partial/\partial z_1, \dots, \partial/\partial z_{n_{KL}}]^T$, and dS is a differential element of the limit state hypersurface $S = \{\mathbf{z} \in \mathbb{R}^{n_{KL}} : g(\mathbf{y}, \mathbf{z}) = 0\}$. However, evaluating the expression in Equation (15) is challenging, as it requires solving a $(n_{KL} - 1)$ -dimensional integral. To address this issue, an advanced simulation technique based on multidomain Line Sampling is introduced in Section 3 for estimating the integral.

3. First Excursion Probability Estimation

3.1. General Remarks

The first excursion probability is evaluated by reformulating the integral in Equation (10) in terms of the *effective contribution* of each elementary failure domain to the overall failure probability. This formulation builds upon Line Sampling [49] and its multidomain extension [9]. In the following, the fundamentals of multidomain Line Sampling are explained in detail. Please note that the description presented here differs slightly from that in [9] to facilitate implementation.

258 3.2. *Effective Contribution of the Elementary Failure Domains*

259 The first excursion probability integral shown in Equation (10) can be very challenging. This
 260 is due to the difficulty in analytically characterizing the shape of the failure domain. As shown
 261 in Section 2.4, the analytical definition of each elementary failure domain provides valuable infor-
 262 mation for assessing the first excursion probability. Therefore, the failure probability integral in
 263 Equation (10) can be expressed in terms of the contributions of each elementary failure domain
 264 [6, 9], as follows:

$$p_F(\mathbf{y}) = \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} p_{i,k}(\mathbf{y}), \quad (16)$$

265 where $p_{i,k}$, known as the *effective contribution*, is related to the elementary failure domain $F_{i,k}$
 266 and is defined as follows:

$$p_{i,k}(\mathbf{y}) = \int_{\mathbf{z} \in F_{i,k}} \frac{1}{\sum_{h=1}^{n_\eta} \sum_{j=1}^{n_T} I_{F_{h,j}}(\mathbf{y}, \mathbf{z})} f_{\mathbf{z}}(\mathbf{z}) d\mathbf{z}. \quad (17)$$

267 The term $I_{F_{h,j}}(\mathbf{y}, \mathbf{z})$ is an indicator function which is equal to 1 in case that $\mathbf{z} \in F_{h,j}$ and 0
 268 otherwise.

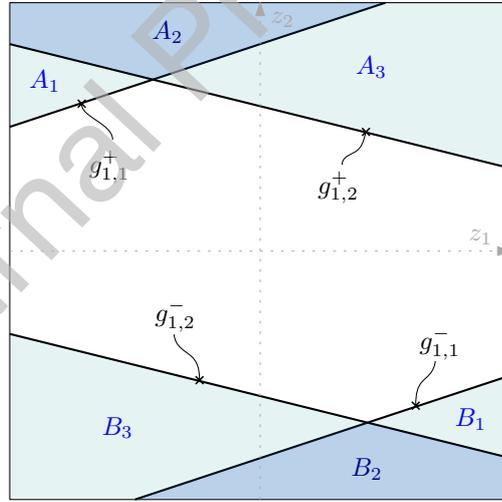


Figure 2: Elementary failure domains representation for the case where $n_\eta = 1$ and $n_T = n_{KL} = 2$.

269 The effective contribution of each elementary failure domain can be calculated by decomposing
 270 the failure domains into zones that exhibit varying levels of interaction between the elementary
 271 failure domains. This is illustrated schematically in Figure 2 for the particular case where $n_\eta = 1$
 272 and $n_T = n_{KL} = 2$. As shown in Figure 2, the subsets of the failure domain denoted by A_1 ,
 273 A_3 , B_1 , and B_3 are defined by a single elementary failure domain each, without exhibiting any

274 overlap. The remaining subsets, A_2 and B_2 , are defined by two elementary failure domains each,
 275 exhibiting overlap between them. To understand the definition of effective contribution, consider
 276 the calculation of $p_{1,2}$ in the example shown in Figure 2. The integral from Equation (17) can be
 277 separated over the domains A_2 , A_3 , B_2 and B_3 . The *discounting factor* $1/\sum_{h=1}^{n_\eta} \sum_{j=1}^{n_T} I_{F_{h,j}}(\mathbf{y}, \mathbf{z})$,
 278 from Equation (17), accounts for discounting the effective contribution resulting from the inter-
 279 action between elementary failure domains. Then, the discounting factor value is equal to 1 when
 280 integrating over domains A_3 and B_3 , and equal to $1/2$ when integrating over domains A_2 and
 281 B_2 . Therefore, it is straightforward to note that, in Equation (17), the contribution to the failure
 282 probability from the region where elementary failure domains overlap is accounted for half in the
 283 calculation of $p_{1,1}$ and half in $p_{1,2}$. Consequently, the effective contribution $p_{i,k}$ is given by the
 284 probability of occurrence of the event $F_{i,k}$, adjusted by a discounting factor that accounts for the
 285 overlap between elementary failure domains. Thus, the addition of all effective contributions $p_{i,k}$
 286 over all responses η_i , $i = 1, \dots, n_\eta$ and over all time instants t_k , $k = 1, \dots, n_T$, provides the sought
 287 failure probability $p_F(\mathbf{y})$ as shown in Equation (16).

288 3.3. Multidomain Line Sampling Formulation

289 The first excursion probability is calculated based on the effective contributions of the ele-
 290 mentary failure domains, which are estimated using Line Sampling, a simulation approach well
 291 documented in the literature, see e.g. [49]. Therefore, in the following, it is assumed that the
 292 reader is familiar with basic concepts of Line Sampling such as the important direction $\boldsymbol{\alpha}$, rota-
 293 tion of the associated coordinate system, etc. Nevertheless and for the sake of completeness, some
 294 basic concepts of Line Sampling are described in Appendix B. The extension of Line Sampling
 295 to cases where the failure domain has the special structure presented in Section 2.4 is known as
 296 multidomain Line Sampling [9]. To understand its formulation, each elementary failure domain is
 297 now associated with an important direction, defined as:

$$\boldsymbol{\alpha}_{i,k} = \frac{\mathbf{z}_{i,k}^*(\mathbf{y})}{\|\mathbf{z}_{i,k}^*(\mathbf{y})\|}, \quad (18)$$

298 where $\boldsymbol{\alpha}_{i,k}$ is a vector of unit Euclidean norm, pointing in the direction of the design point of the
 299 (i, k) -th elementary failure domain. Note that whether the vector points towards $F_{i,k}^+$ or $F_{i,k}^-$ does
 300 not affect the formulation of the method. Now, it is possible to define a rotated coordinate system

301 associated with the elementary failure domain $F_{i,k}$ as:

$$z = \mathbf{R}_{i,k} z_{i,k}^\perp + \alpha_{i,k} z_{i,k}^\parallel, \quad (19)$$

302 where $\mathbf{R}_{i,k}$ is an $n \times (n-1)$ matrix; $z_{i,k}^\perp$ is an $(n-1)$ -dimensional vector representing the coordinates
 303 in the hyperplane orthogonal to $\alpha_{i,k}$; and z^\parallel is a scalar representing the coordinate along the
 304 direction parallel to $\alpha_{i,k}$. The orthonormal basis of the new coordinate system is defined by the
 305 square matrix $[\mathbf{R}_{i,k}, \alpha_{i,k}]$. Then, $z_{i,k}^\parallel$ follows a one-dimensional standard Gaussian distribution,
 306 while $z_{i,k}^\perp$ is associated with a standard Gaussian distribution in $(n-1)$ dimensions. Note that
 307 the dependence on \mathbf{y} is omitted for $\alpha_{i,k}$ and $\mathbf{R}_{i,k}$, as the coordinate system and the associated
 308 definitions are kept fixed for the subsequent calculations in this contribution (evaluated at the
 309 current design). Figure 3 illustrates the important directions $\alpha_{1,1}$ and $\alpha_{1,2}$, as well as a rotated
 310 system generated according to $F_{1,2}$, for the case where $n_\eta = 1$ and $n_{KL} = n_T = 2$. The axes $z_{1,2}^\parallel$
 311 and $z_{1,2}^\perp$ indicate the parallel and perpendicular directions with respect to $F_{1,2}$. The boundaries
 312 of the failure and safe domains are also shown.

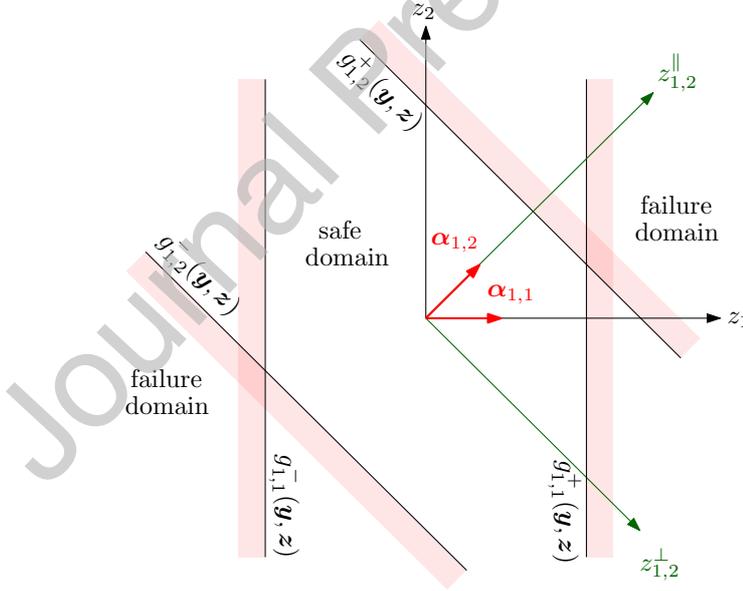


Figure 3: Schematic representation of the multidomain Line Sampling technique.

313 The rotated coordinate system from Equation (19) allows the effective contribution integral
 314 from Equation (17) to be expressed as follows:

$$p_{i,k}(\mathbf{y}) = \int_{\Omega_{i,k}^\perp} \int_{\Omega_{i,k}^\parallel} \frac{1}{\sum_{h=1}^{n_\eta} \sum_{j=1}^{n_T} I_{F_{h,j}}(\mathbf{R}_{i,k} z_{i,k}^\perp + \alpha_{i,k} z_{i,k}^\parallel)} f_{Z_{i,k}^\parallel}(z_{i,k}^\parallel) f_{Z_{i,k}^\perp}(z_{i,k}^\perp) dz_{i,k}^\parallel dz_{i,k}^\perp, \quad (20)$$

315 where $\Omega_{i,k}^\perp = \{z_{i,k}^\perp \in \mathbb{R}^{(n_{KL}-1)}\}$ and $\Omega_{i,k}^\parallel(\mathbf{y}) = \{z_{i,k}^\parallel \in \mathbb{R}^1 : |\mathbf{a}_{i,k}^T(\mathbf{y})\mathbf{z}| \geq b_i\}$ are introduced to sim-
 316 plify the notation. The above integral requires integration along a line that accounts for the
 317 interaction between different elementary failure domains. Therefore, Equation (20) provides an
 318 expression for calculating the effective contribution to the failure probability using multidomain
 319 Line Sampling.

320 3.4. First Excursion Probability by means of Multidomain Line Sampling

321 The estimation of the first excursion probability using Equation (16) presents two important
 322 challenges. First, Equation (20) requires the calculation of its inner integral, which can be com-
 323 puted analytically, a procedure that is explained later in Section 5.3. Second, Equation (16)
 324 requires the calculation of all the effective contribution terms $p_{i,k}$, where $n_\eta n_T$ can be on the
 325 order of thousands. This can be highly computationally demanding. To address this issue, the
 326 summation in Equation (16) can be estimated through simulation [7], leading to:

$$p_F(\mathbf{y}) = \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} \left(\frac{1}{\omega_{i,k}} p_{i,k}(\mathbf{y}) \right) \omega_{i,k}, \quad (21)$$

327 where $\omega_{i,k}$, $i = 1, \dots, n_\eta$, $k = 1, \dots, n_T$ correspond to weight factors such that $\omega_{i,k} > 0$ and
 328 $\sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} \omega_{i,k} = 1$. These weights can be chosen using different criteria. A good choice is to rank
 329 the importance of the effective contributions to the failure probability in proportion to the failure
 330 probability of their respective elementary failure domains. Therefore, the weights in Equation (21)
 331 play the role of a probability mass function, which is defined as [6]:

$$\omega_{i,k} = \frac{P[F_{i,k}]}{\sum_{h=1}^{n_\eta} \sum_{j=1}^{n_T} P[F_{h,j}]}. \quad (22)$$

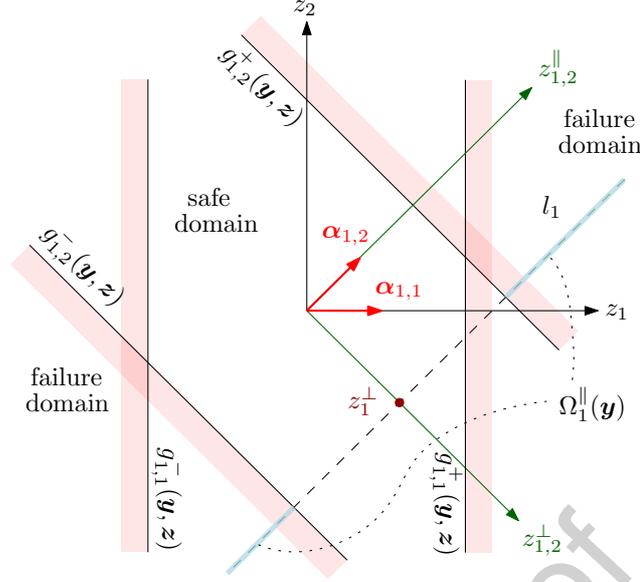


Figure 4: Schematic representation of the multidomain Line Sampling technique, including one sample realization z_1^\perp (dark red), its associated line l_1 (black dashed line), and domain $\Omega_1^\parallel(\mathbf{y})$ (light blue).

332 The first excursion probability in Equation (21) now involves the summation of a discrete
 333 random variable and integration over various continuous random variables. By simulating both
 334 types of random variables, the first excursion probability can be estimated as:

$$\tilde{p}_F = \frac{1}{N} \sum_{j=1}^N \left(\frac{1}{\omega_{s_j}} \tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp) \right). \quad (23)$$

335 Here, \tilde{p}_F is an estimate of p_F ; N represents the total number of samples; for each $j = 1, \dots, N$,
 336 an index s_j is independently sampled from the set $I = \{1, \dots, n_\eta n_T\}$ according to the probability
 337 mass function ω_{s_j} . Each s_j uniquely corresponds to a pair (i, k) , thereby identifying a particular
 338 elementary failure domain, i.e., variables that depend on a pair (i, k) , such as $\alpha_{i,k}$, are written as
 339 α_{s_j} ; the vector \mathbf{z}_j^\perp follows a $(n_{KL} - 1)$ -dimensional standard multivariate Gaussian distribution
 340 and is always defined with respect to the coordinate system associated with the selected domain
 341 s_j for sample j ; and \tilde{p}_{s_j} denotes the estimate of the effective contribution evaluated at \mathbf{y} and \mathbf{z}_j^\perp .
 342 The quantity \tilde{p}_{s_j} can be calculated as:

$$\tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp) = \int_{\Omega_j^\parallel(\mathbf{y})} \frac{1}{\sum_{h=1}^{n_\eta} \sum_{l=1}^{n_T} I_{F_{h,l}}(\mathbf{R}_{s_j} \mathbf{z}_j^\perp + \alpha_{s_j} \mathbf{z}_j^\parallel)} f_{Z_{s_j}^\parallel}(\mathbf{z}_j^\parallel) dz_j^\parallel. \quad (24)$$

343 where $\Omega_j^\parallel(\mathbf{y})$ is the portion of the failure domain lying on the line $l_j = \{\mathbf{R}_{s_j} \mathbf{z}_j^\perp + \alpha_{s_j} \mathbf{z}_j^\parallel \mid \mathbf{z}_j^\parallel \in \mathbb{R}\}$;

344 that is, $\Omega_j^{\parallel}(\mathbf{y}) = l_j \cap F_{s_j}$. In the same manner, the positive and negative portions of the domain
 345 can be defined as $\Omega_j^{+, \parallel}(\mathbf{y}) = l_j \cap F_{s_j}^+$ and $\Omega_j^{-, \parallel}(\mathbf{y}) = l_j \cap F_{s_j}^-$, respectively. It is important to
 346 note that the integral in Equation (24) must be evaluated along a line corresponding to a possible
 347 realization of the random variable \mathbf{z}_j^{\perp} . This is depicted in Figure 4, where the sample $j = 1$
 348 selects the domain index s_1 corresponding to the pair $(i, k) = (1, 2)$. Thus, the realization z_1^{\perp} is
 349 represented as a dark red point, together with its corresponding line l_1 (black dashed line), both
 350 defined with respect to the coordinate system of $F_{1,2}$. The light blue region indicates the domain
 351 $\Omega_1^{\parallel}(\mathbf{y})$ associated with this sample, whose associated numerical integration is explained in detail
 352 in Section 5.3. Therefore, Equation (23) defines the first excursion probability estimator based on
 353 *multidomain Line Sampling* [9].

354 It is straightforward to derive that the coefficient of variation, δ_{p_F} , associated with the first
 355 excursion probability estimator in Equation (23) is given by:

$$\delta_{p_F} = \frac{1}{\tilde{p}_F(\mathbf{y})} \sqrt{\frac{1}{N(N-1)} \sum_{j=1}^N \left(\left(\frac{1}{\omega_{s_j}} \tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^{\perp}) \right) - \tilde{p}_F(\mathbf{y}) \right)^2} \quad (25)$$

356 4. Sensitivity Estimation of First Excursion Probability

357 4.1. General Remarks

358 This contribution extends the multidomain Line Sampling [9] technique, originally developed
 359 for the calculation of first-excursion probabilities in linear systems subject to Gaussian loading,
 360 to also provide an estimator for the sensitivity of the first excursion probability. The calculation
 361 of the first excursion probability sensitivity, as shown in Equation (15), requires the calculation of
 362 an integral over the limit state hypersurface. The literature suggests that estimating this quantity
 363 can be done, for example, by methods such as Directional Sampling [15] and Line Sampling
 364 [50]. The unique configuration of the failure domain, as described in Equation (9), motivates the
 365 development of specialized methods to address this challenge, with the present work being one
 366 such contribution.

367 4.2. Sensitivity of the First Excursion Probability by means of Multidomain Line Sampling

368 The sensitivity of the first excursion probability can be obtained by calculating the derivative
 369 of the effective contributions from Equation (16) with respect to a design parameter y_q , leading

370 to:

$$\frac{\partial p_F(\mathbf{y})}{\partial y_q} = \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} \frac{\partial p_{i,k}(\mathbf{y})}{\partial y_q}, \quad (26)$$

371 where $\partial(\cdot)/\partial y_q$ denotes the partial derivative with respect to the design parameter y_q . The influ-
 372 ence of changes in the effective contributions due to a perturbation Δ in the design parameter y_q
 373 is illustrated schematically in Figure 5, for the case where $n_\eta = 1$, $n_T = n_{KL} = 2$, $n_Y = 1$, and the
 374 design parameter y_q is perturbed. The limit state functions without perturbation of the design
 375 parameter are denoted by $g_{1,1}(y_q, \mathbf{z})$ and $g_{1,2}(y_q, \mathbf{z})$, while the perturbed limit state functions are
 376 represented as $g_{1,1}(y_q + \Delta, \mathbf{z})$ and $g_{1,2}(y_q + \Delta, \mathbf{z})$, in both cases associated with the elementary
 377 failure domains $F_{1,1}$ and $F_{1,2}$, respectively. A sample $j = 1$ is generated according to the pair
 378 $(i, k) = (1, 2)$ in order to estimate the effective contribution $p_{1,2}$, which is approximated by \tilde{p}_1
 379 using Equation (24). In this case, it is necessary to perform an integration over a portion of the
 380 line l_1 . The perturbation Δ induces changes in the limit state functions of the elementary failure
 381 domains and, as a consequence, affects both the portion of the integration over the line l_1 , which
 382 is now represented by the domain $\Omega_1^\parallel(y_q + \Delta)$ (shown as a light blue line), and the length of the
 383 segments with different degrees of overlap. Therefore, the sensitivity of the effective contributions
 384 quantifies these changes with respect to a specific design parameter y_q and can be calculated
 385 directly from Equation (20), leading to:

$$\frac{\partial p_{i,k}(\mathbf{y})}{\partial y_q} = \int_{\Omega_{i,k}^\perp} \frac{\partial}{\partial y_q} \left(\int_{\Omega_{i,k}^\parallel} \frac{1}{\sum_{h=1}^{n_\eta} \sum_{j=1}^{n_T} I_{F_{h,j}}(\mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp + \boldsymbol{\alpha}_{i,k} \mathbf{z}_{i,k}^\parallel)} f_{Z_{i,k}^\parallel}(z_{i,k}^\parallel) dz_{i,k}^\parallel \right) f_{Z_{i,k}^\perp}(z_{i,k}^\perp) dz_{i,k}^\perp. \quad (27)$$

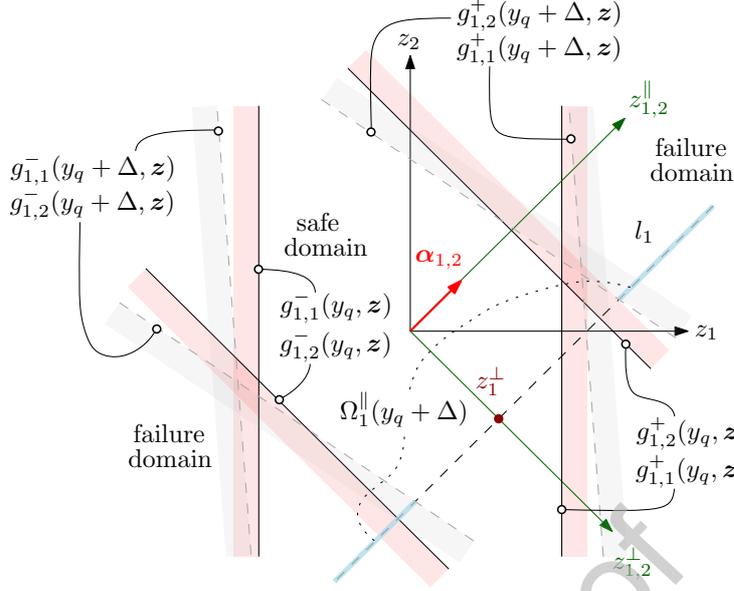


Figure 5: Schematic representation for the sensitivity of the effective contributions for the case where $n_\eta = 1$, $n_T = n_{KL} = 2$ and $n_Y = 1$.

386 The estimation of the sensitivity of the first excursion probability using Equation (26) presents
 387 two important challenges, in the same way as Equation (16). First, the inner integral in Equa-
 388 tion (27) is computed analytically, a procedure that is explained later in Section 5.3. Second,
 389 Equation (16) requires the calculation of all the partial derivatives of the effective contribution
 390 terms $\partial p_{i,k}(\mathbf{y})/\partial y_q$, where $n_\eta n_T$ can be on the order of thousands. This may incur a high com-
 391 putational cost. To address this issue, the summation in Equation (26) can be estimated through
 392 simulation [7] as in Section 3.4, resulting in:

$$\frac{\partial p_F(\mathbf{y})}{\partial y_q} = \sum_{i=1}^{n_\eta} \sum_{k=1}^{n_T} \left(\frac{1}{\omega_{i,k}} \frac{\partial p_{i,k}(\mathbf{y})}{\partial y_q} \right) \omega_{i,k}, \quad (28)$$

393 where $\omega_{i,k}$ is defined in Equation (22). The sensitivity of the first excursion probability in Equa-
 394 tion (28) now involves the summation of a discrete random variable and integration over various
 395 continuous random variables. By simulating both types of random variables, the sensitivity of the
 396 first excursion probability can be estimated as:

$$\frac{\partial p_F(\mathbf{y})}{\partial y_q} \approx \frac{\partial \tilde{p}_F(\mathbf{y})}{\partial y_q} = \frac{1}{N} \sum_{j=1}^N \left(\frac{1}{\omega_{s_j}} \frac{\partial \tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp)}{\partial y_q} \right). \quad (29)$$

397 Here, $\partial \tilde{p}_F(\mathbf{y})/\partial y_q$ is an estimate of $\partial p_F(\mathbf{y})/\partial y_q$ and $\partial \tilde{p}_{s_j}/\partial y_q$ denotes the derivative of the effective
 398 contribution with respect to the design parameter y_q , evaluated at \mathbf{y} and \mathbf{z}_j^\perp . The notation for

399 s_j , \mathbf{z}_j^\perp , and ω_{s_j} is the same described in Equation (23). Therefore, the quantity $\partial\tilde{p}_{s_j}/\partial y_q$ can be
 400 calculated as:

$$\frac{\partial}{\partial y_q} \tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp) = \frac{\partial}{\partial y_q} \left(\int_{\Omega_j^\parallel(\mathbf{y})} \frac{1}{\sum_{h=1}^{n_n} \sum_{l=1}^{n_T} I_{F_{h,l}}(\mathbf{R}_{s_j} \mathbf{z}_j^\perp + \boldsymbol{\alpha}_{s_j} z_j^\parallel)} f_{Z_{s_j}^\parallel}(z_j^\parallel) dz_j^\parallel \right). \quad (30)$$

401 Similarly to Equation (24), the integral in Equation (30) must be evaluated along the line l_j , a
 402 procedure explained later in Section 5.3. The integration domain $\Omega(\mathbf{y})$ defines the failure region
 403 corresponding to the design vector \mathbf{y} . Changes in \mathbf{y} modify the geometry of this domain, and the
 404 resulting displacement of its boundaries is the sole source of sensitivity with respect to the design
 405 parameters. Therefore, Equation (29) defines the sensitivity of the first excursion probability
 406 estimator based on *multidomain Line Sampling*.

407 The coefficient of variation, $\delta_{\partial\tilde{p}_F/\partial y_q}$, associated with the sensitivity of the first excursion prob-
 408 ability estimator in Equation (29), can be readily derived as follows:

$$\delta_{\partial\tilde{p}_F/\partial y_q} = \frac{1}{\partial\tilde{p}_F/\partial y_q} \sqrt{\frac{1}{N(N-1)} \sum_{j=1}^N \left(\left(\frac{1}{\omega_{s_j}} \frac{\partial\tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp)}{\partial y_q} \right) - \frac{\partial\tilde{p}_F(\mathbf{y})}{\partial y_q} \right)^2} \quad (31)$$

409 where Equation (31) is evaluated under the assumption that $\partial p_F(\mathbf{y})/\partial y_q$ is nonzero.

410 Note that the estimator in Equation (30) evaluates exploration lines within $\Omega_j^\parallel(\mathbf{y})$ that are
 411 guided by the elementary failure domain F_j ; as a consequence, the exploration line intersects
 412 the corresponding limit-state hyperplane of F_j orthogonally by construction, which is particularly
 413 informative for sensitivity estimation. In contrast, directional approaches such as DIS [37] and
 414 DDM [29] construct exploration lines as rays anchored at the origin and passing through a point
 415 associated with F_j , which can lead to less informative intersection angles and therefore require
 416 more simulations to achieve the same accuracy in the sensitivity estimates.

417 5. Practical Implementation

418 The implementation of Equations (24) and (30) requires special attention to four distinct
 419 aspects: the generation of samples \mathbf{z}_j^\perp , the calculation of relevant distances along the line l_j , the
 420 integration over the segment Ω_j^\parallel , and the computation of partial derivatives of the relevant terms.
 421 These issues are addressed in this section.

422 5.1. Generation of Samples \mathbf{z}_j^\perp

423 The generation of independent and identically distributed samples \mathbf{z}_j^\perp , $j = 1, \dots, N$, which
 424 follow a $(n_{KL} - 1)$ -dimensional standard multivariate Gaussian distribution, can be performed
 425 using the following procedure:

- 426 1. Define $I = \{(i, k) : i = 1, \dots, n_\eta; k = 1, \dots, n_T\}$ as the set of all index pairs. Each pair
 427 (i, k) can be mapped to a unique index in $\{1, \dots, n_\eta n_T\}$.
- 428 2. For $j = 1, \dots, N$, repeat Steps 2a–2c:
 - 429 (a) Select a pair (i, k) from I according to the weights $\omega_{i,k}$ in Equation (22), and let s_j
 430 denote the index corresponding to the selected pair.
 - 431 (b) Generate a sample \mathbf{z}_j following an n_{KL} -dimensional standard Gaussian distribution.
 - 432 (c) Calculate [49, 51]:

$$\mathbf{R}_{s_j} \mathbf{z}_j^\perp = \mathbf{z}_j - \left(\boldsymbol{\alpha}_{s_j}^T \mathbf{z}_j \right) \boldsymbol{\alpha}_{s_j} \quad (32)$$

433 To illustrate how the algorithm works, Figure 6 presents a schematic representation for the
 434 case where $n_\eta = 1$ and $n_T = n_{KL} = 2$. Following Step 2a, the selected element is $s_1 = (1, 2)$,
 435 and, as specified in Step 2b, the corresponding sample \mathbf{z}_1 is generated and represented by a black
 436 dot in Figure 6. Then, in Step 2c, the sought quantity corresponds to $\mathbf{R}_{s_1} \mathbf{z}_1^\perp$, which is a vector
 437 represented with a dark blue arrow. This vector can be calculated by subtracting the vector
 438 $\left(\boldsymbol{\alpha}_{s_1}^T \mathbf{z}_1 \right) \boldsymbol{\alpha}_{s_1}$, represented by a dark red arrow, from the original sample \mathbf{z}_1 . This step is crucial for
 439 the multidomain Line Sampling implementation, as the quantity required by the method is $\mathbf{R}_{s_1} \mathbf{z}_1^\perp$,
 440 rather than just \mathbf{z}_1^\perp . This leads to a computational advantage, as it avoids explicitly computing
 441 the rotation matrices \mathbf{R}_{s_1} for each sample.

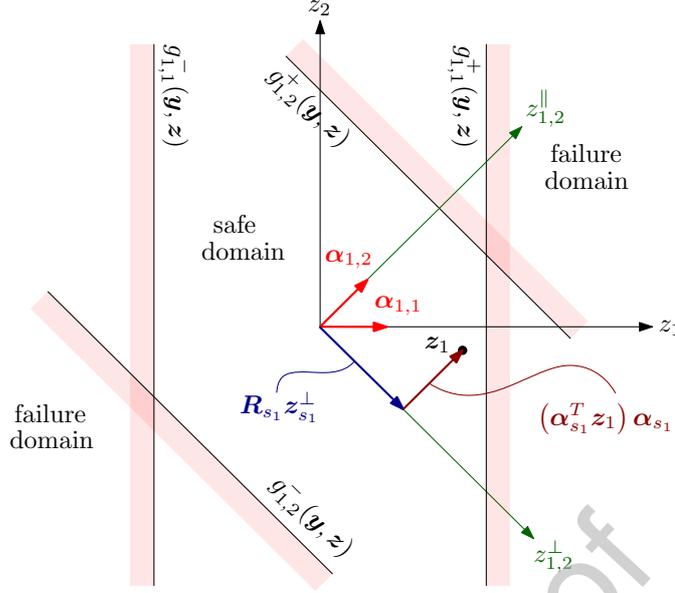


Figure 6: Representation of sample generation for the case where $n_\eta = 1$ and $n_T = n_{KL} = 2$.

442 5.2. Computation of Relevant Distances Along l_j

443 To address the line integration problem in Equations (24) and (30), it is important to recognize
 444 that, for a given realization of the sample \mathbf{z}_j , the integration domain $\Omega_j^\parallel(\mathbf{y})$ may exhibit varying
 445 degrees of overlap between elementary failure domains. Therefore, it is essential to geometrically
 446 identify the locations along the line l_j where the overlap between elementary failure domains
 447 changes. To precisely describe these locations, we use the rotated coordinate system introduced
 448 in Equation (19). Any point \mathbf{z} in standard normal space can be written, in terms of the (i, k) -th
 449 rotated coordinate system, as $\mathbf{z} = \mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp + \alpha_{i,k} c(\mathbf{y})$, where $c(\mathbf{y})$ represents how much one moves,
 450 positively or negatively, from the base point $\mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp$ along the axis defined by the important
 451 direction $\alpha_{i,k}$. Consequently, the response of interest can be expressed as:

$$\eta_i(t_k, \mathbf{y}, \mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp + \alpha_{i,k} c(\mathbf{y})) = \eta_i^\perp(t_k, \mathbf{y}, \mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp) + c(\mathbf{y}) \cdot \eta_i^\parallel(t_k, \mathbf{y}, \alpha_{i,k}), \quad (33)$$

452 where η_i^\perp is the perpendicular response of interest, and η_i^\parallel is the unit parallel response of interest,
 453 both obtained by evaluating the Gaussian vector in Equation (7) at $\mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp$ and $\alpha_{i,k}$, respectively.

454 Then, the quantities η_i^\perp and η_i^\parallel are defined as:

$$\eta_i^\perp(t_k, \mathbf{y}, \mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp) = \mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{R}_{i,k} \mathbf{z}_{i,k}^\perp, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T, \quad (34)$$

$$\eta_i^\parallel(t_k, \mathbf{y}, \alpha_{i,k}) = \mathbf{a}_{i,k}(\mathbf{y})^T \alpha_{i,k}, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T. \quad (35)$$

456 The quantity c , from Equation (33), plays a crucial role in evaluating the one-dimensional
 457 integral. To illustrate its geometric and physical meaning, Figure 7 presents the case where
 458 $n_\eta = 1$ and $n_T = n_{KL} = 2$, in the context of estimating the effective contribution $p_{1,2}$. For this
 459 case, a sample realization \mathbf{z}_1^\perp has been generated, together with the corresponding line l_1 and the
 460 integration domain $\Omega_1^\parallel(\mathbf{y})$ depicted along this line. To proceed, it is necessary to explicitly define
 461 c at the points where the response of interest meets the positive and negative thresholds, for each
 462 elementary failure domain. Along each line l_j , the distances from the point \mathbf{z}_j^\perp to the intersections
 463 with the corresponding hyperplanes, each defined by its own performance function, are called
 464 *crossing distances* and denoted by $c_{i,k}^{+,j}$ and $c_{i,k}^{-,j}$. The subscript (i, k) refers to the i -th response
 465 of interest at the k -th time instant, and the superscript j indicates the sample (and associated
 466 line) for which the distance is computed. In addition, the superscripts $(+)$ and $(-)$ indicate the
 467 distances to the positive (response η_i up-crossing its threshold b_i at time t_k) and negative (response
 468 η_i down-crossing its threshold b_i at time t_k) limit state functions, respectively. In Figure 7, the
 469 case corresponding to sample $j = 1$, with the pair of indices (i, k) given by $s_1 = (1, 2)$, is presented.
 470 Therefore, for an arbitrary sample \mathbf{z}_j^\perp with associated index s_j , evaluating Equation (33) at the
 471 points where the response equals the thresholds b_i and $-b_i$ at the time instant t_k along the line
 472 l_j , leads to:

$$b_i = \mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{R}_{s_j} \mathbf{z}_j^\perp + c_{i,k}^{+,j}(\mathbf{y}) \cdot \mathbf{a}_{i,k}(\mathbf{y})^T \boldsymbol{\alpha}_{s_j} \quad (36)$$

$$-b_i = \mathbf{a}_{i,k}(\mathbf{y})^T \mathbf{R}_{s_j} \mathbf{z}_j^\perp + c_{i,k}^{-,j}(\mathbf{y}) \cdot \mathbf{a}_{i,k}(\mathbf{y})^T \boldsymbol{\alpha}_{s_j}. \quad (37)$$

474 Then, rewriting Equations (36) and (37) in terms of the perpendicular and unit parallel responses
 475 defined in Equations (34) and (35), the distances $c_{i,k}^{-,j}$ and $c_{i,k}^{+,j}$ associated with the line l_j are
 476 defined as:

$$c_{i,k}^{-,j} = \frac{-b_i - \eta_i^\perp(t_k, \mathbf{y}, \mathbf{R}_{s_j} \mathbf{z}_j^\perp)}{\eta_i^\parallel(t_k, \mathbf{y}, \boldsymbol{\alpha}_{s_j})}, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T, \quad j = 1, \dots, N, \quad (38)$$

$$c_{i,k}^{+,j} = \frac{b_i - \eta_i^\perp(t_k, \mathbf{y}, \mathbf{R}_{s_j} \mathbf{z}_j^\perp)}{\eta_i^\parallel(t_k, \mathbf{y}, \boldsymbol{\alpha}_{s_j})}, \quad i = 1, \dots, n_\eta, \quad k = 1, \dots, n_T, \quad j = 1, \dots, N, \quad (39)$$

478 where it is assumed that the unit parallel response $\eta_i^\parallel(t_k, \mathbf{y}, \boldsymbol{\alpha}_{s_j})$ takes nonzero values. Therefore,
 479 Equations (38) and (39) provide a measure of how much it is necessary to amplify the unit parallel
 480 response $\eta_i^\parallel(t_k, \mathbf{y}, \boldsymbol{\alpha}_{s_j})$, given a perpendicular response $\eta_i^\perp(t_k, \mathbf{y}, \mathbf{R}_{s_j} \mathbf{z}_j^\perp)$, in order to reach the

481 failure condition at the time instant t_k . It is important to note that the computation of the
 482 crossing distances requires performing two dynamic analyses: one for the perpendicular response
 483 and another for the unit parallel response (see Equations (34) and (35)).

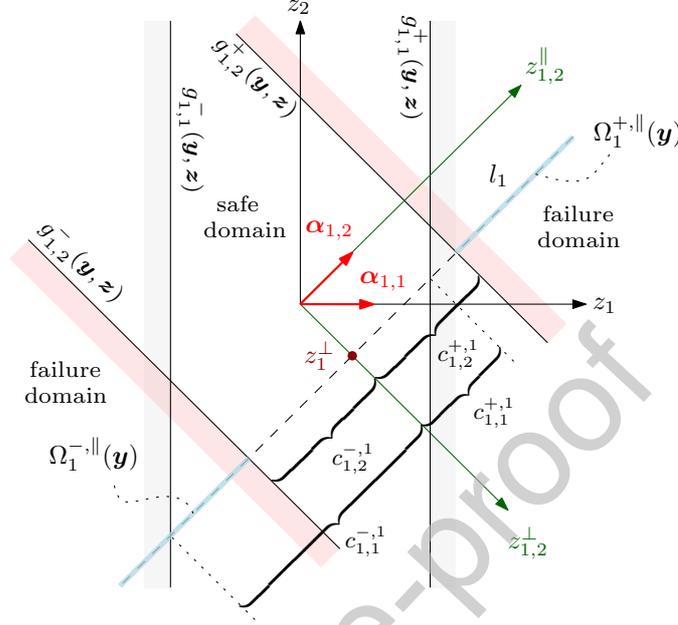


Figure 7: Crossing distances representation over line l_1 for the case where $n_\eta = 1$ and $n_T = n_{KL} = 2$.

484 5.3. One-dimensional Integration Over $\Omega_j^{\parallel}(\mathbf{y})$

485 5.3.1. Preliminary Notation

486 The estimation of the first excursion probability using multidomain Line Sampling requires the
 487 evaluation of the inner integral from Equation (20), for different realizations of \mathbf{z}_j^{\perp} , as shown in
 488 Equation (24). In order to clearly define the segments exhibiting different degrees of overlap be-
 489 tween elementary failure domains over Ω_j^{\parallel} , the distances $c_{i,k}^{-,j}$ and $c_{i,k}^{+,j}$, as defined in Equations (38)
 490 and (39), are grouped into the vector \mathcal{C}^j :

$$\mathcal{C}^j = \left\{ c_{i,k}^{-,j}, c_{i,k}^{+,j} \right\}^T, \quad \text{for } i = 1, \dots, n_\eta; \quad k = 1, \dots, n_T. \quad (40)$$

491 To facilitate integration implementation, the set is augmented with the endpoints $-\infty$ and $+\infty$,
 492 and its elements are sorted in ascending order to form the augmented vector:

$$\tilde{\mathcal{C}}_{\text{sort}}^j = \left\{ c_{[0]}^j, c_{[1]}^j, \dots, c_{[n_F]}^j, c_{[n_F+1]}^j \right\}^T, \quad (41)$$

493 where $n_F = 2n_\eta n_T$ denotes the total number of intersections between all limit state functions of

494 the elementary failure domains and the line l_j . The boundary values are set as $c_{[0]}^j = -\infty$ and
 495 $c_{[n_F+1]}^j = +\infty$, while the intermediate values $c_{[1]}^j, \dots, c_{[n_F]}^j$ correspond to the sorted entries of \mathcal{C}^j in
 496 ascending order. Each segment $[c_{[m-1]}^j, c_{[m]}^j]$, for $m = 1, \dots, n_F + 1$, defines a segment of the line
 497 l_j where the number of overlapping elementary failure domains remains constant.

498 To track how the overlap between elementary failure domains changes between two consecutive
 499 segments, we define the vector \mathcal{V}^j , whose entries, $\nu_{i,k}^{-,j}$ and $\nu_{i,k}^{+,j}$, are called *transition indicators*:

$$\mathcal{V}^j = \left\{ \nu_{i,k}^{-,j} = -1, \nu_{i,k}^{+,j} = +1 \right\}^T, \quad \text{for } i = 1, \dots, n_\eta; k = 1, \dots, n_T, \quad (42)$$

500 where each element in \mathcal{V}^j is associated with a corresponding distance in \mathcal{C}^j . The value +1 is
 501 assigned to each $\nu_{i,k}^{+,j}$ and -1 to each $\nu_{i,k}^{-,j}$. These values indicate whether an elementary failure
 502 domain becomes active (+1) or inactive (-1) when traversing the line l_j in the direction of
 503 the important direction α_{s_j} . To align the transition indicators with the sorted vector $\tilde{\mathcal{C}}_{\text{sort}}^j$, a
 504 corresponding sorted transition indicator vector $\tilde{\mathcal{V}}_{\text{sort}}^j$ is introduced. This vector is constructed by
 505 applying the same permutation used to sort \mathcal{C}^j to the elements of \mathcal{V}^j , and is extended by including
 506 two auxiliary entries at the boundaries. Specifically,

$$\tilde{\mathcal{V}}_{\text{sort}}^j = \left\{ \nu_{[0]}^j, \nu_{[1]}^j, \dots, \nu_{[n_F]}^j, \nu_{[n_F+1]}^j \right\}^T, \quad (43)$$

507 where the first and last elements are auxiliary values defined as $\nu_{[0]}^j = 0$ and $\nu_{[n_F+1]}^j = 0$, included
 508 to match the structure of $\tilde{\mathcal{C}}_{\text{sort}}^j$ in order to simplify the numerical implementation. This vector
 509 allows computing the number of active elementary failure domains in each segment $[c_{[m-1]}^j, c_{[m]}^j]$
 510 via a cumulative sum over $\nu_{[1]}^j$ to $\nu_{[m-1]}^j$, for $m = 1, \dots, n_F + 1$, as explained later in this section.

511 5.3.2. Definition of the Integration Domain

512 To define the integration domain Ω_j^\parallel , recall that integration is performed only within the failure
 513 domain F_{s_j} associated with the effective contribution being estimated, p_{s_j} ; that is, in the intervals
 514 $(-\infty, -\beta_{s_j}] \cup [\beta_{s_j}, +\infty)$ within the line l_j . Let λ_N^j and λ_P^j denote the indices in the sorted vector
 515 $\tilde{\mathcal{C}}_{\text{sort}}^j$ such that:

$$c_{[\lambda_N^j]}^j = -\beta_{s_j}, \quad c_{[\lambda_P^j]}^j = \beta_{s_j}. \quad (44)$$

516 These correspond to the positions in the sorted vector $\tilde{\mathcal{C}}_{\text{sort}}^j$ that contain the negative and positive
 517 crossing distances associated with F_{s_j} . Note that the positive and negative portions of Ω_j^\parallel can also

518 be expressed in terms of segments as:

$$\Omega_j^{-,\parallel} = \bigcup_{r=1}^{\lambda_N^j} [c_{[r-1]}^j, c_{[r]}^j], \quad \Omega_j^{+,\parallel} = \bigcup_{r=\lambda_P^j+1}^{n_F+1} [c_{[r-1]}^j, c_{[r]}^j]. \quad (45)$$

519 The vectors defined in Equations (41) and (43) are illustrated in Figure 8 and Table 1 in
 520 the context of estimating the effective contribution $p_{1,1}$, where $n_\eta = 1$ and $n_T = n_{KL} = 2$. Three
 521 different samples along the coordinate $z_{1,1}^\perp$ are generated to demonstrate how the introduced vectors
 522 can retain useful information irrespective of the sample locations and/or the degree of overlap
 523 between elementary failure domains. The lines l_1 , l_2 , and l_3 are defined from the samples z_1^\perp ,
 524 z_2^\perp , and z_3^\perp , respectively. The domains to be integrated correspond to the positive and negative
 525 portions of Ω_1^\parallel , Ω_2^\parallel , and Ω_3^\parallel , represented by the light blue lines in the figure, where the sorted
 526 crossing distances in $\tilde{\mathcal{C}}_{\text{sort}}^j$ for each l_j are also labeled. The table compiles all corresponding
 527 numerical values, including the crossing distances in $\tilde{\mathcal{C}}_{\text{sort}}^j$ and \mathcal{C}^j , the transition indicators in $\tilde{\mathcal{V}}_{\text{sort}}^j$
 528 with their associated entries in \mathcal{V}^j , and the positions λ_N^j and λ_P^j for each line. A detailed description
 529 of the figure is essential to highlight the challenges that may arise during the implementation of
 530 the method. For lines l_1 and l_3 , the samples are generated inside an elementary failure domain,
 531 resulting in three distances being negative and three being positive, respectively. For line l_1 , and
 532 similarly for l_2 , the number of active elementary failure domains, following the important direction
 533 $\alpha_{1,1}$, first decreases, and then, increases, as shown by the values in $\tilde{\mathcal{V}}_{\text{sort}}^j$. In contrast, for line l_3 ,
 534 the pattern differs, as it remains inside at least one elementary failure domain for its entire length.
 535 It is also worth noting that, in the context of estimating $p_{1,1}$, the positions of λ_N^j and λ_P^j can
 536 vary significantly when exploring different lines l_j . This behavior, although seemingly complex to
 537 assess even in a two-dimensional example, can be readily organized using the quantities shown in
 538 Table 1.

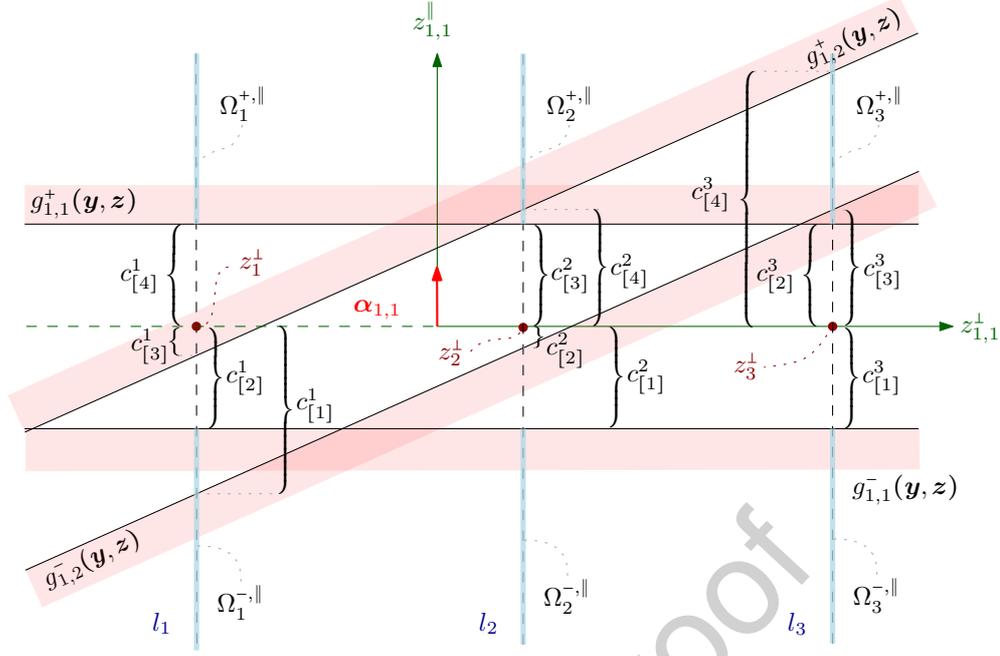


Figure 8: Schematic representation for estimating the effective contribution $p_{1,1}$, for the case where $n_\eta = 1$ and $n_T = n_{KL} = 2$, using three different samples.

Table 1: Values of the vectors \mathcal{C}^j , $\tilde{\mathcal{C}}_{\text{sort}}^j$, ν^j and $\tilde{\nu}_{\text{sort}}^j$ associated with the example presented in Figure 8.

line l_j	entry in \mathcal{C}^j	entry in $\tilde{\mathcal{C}}_{\text{sort}}^j$	entry in ν^j	entry in $\tilde{\nu}_{\text{sort}}^j$	remarks
l_1	$c_{1,2}^{-,1}$	$c_{[1]}^1$	$\nu_{1,2}^{-,1}$	$\nu_{[1]}^1 = -1$	position λ_N^1
	$c_{1,1}^{-,1}$	$c_{[2]}^1$	$\nu_{1,1}^{-,1}$	$\nu_{[2]}^1 = -1$	
	$c_{1,2}^{+,1}$	$c_{[3]}^1$	$\nu_{1,2}^{+,1}$	$\nu_{[3]}^1 = +1$	position λ_P^1
	$c_{1,1}^{+,1}$	$c_{[4]}^1$	$\nu_{1,1}^{+,1}$	$\nu_{[4]}^1 = +1$	
l_2	$c_{1,1}^{-,2}$	$c_{[1]}^2$	$\nu_{1,1}^{-,2}$	$\nu_{[1]}^2 = -1$	position λ_N^2
	$c_{1,2}^{-,2}$	$c_{[2]}^2$	$\nu_{1,2}^{-,2}$	$\nu_{[2]}^2 = -1$	position λ_P^2
	$c_{1,1}^{+,2}$	$c_{[3]}^2$	$\nu_{1,1}^{+,2}$	$\nu_{[3]}^2 = +1$	
	$c_{1,2}^{+,2}$	$c_{[4]}^2$	$\nu_{1,2}^{+,2}$	$\nu_{[4]}^2 = +1$	
l_3	$c_{1,1}^{-,3}$	$c_{[1]}^3$	$\nu_{1,1}^{-,3}$	$\nu_{[1]}^3 = -1$	position λ_N^3
	$c_{1,1}^{+,3}$	$c_{[2]}^3$	$\nu_{1,1}^{+,3}$	$\nu_{[2]}^3 = +1$	position λ_P^3
	$c_{1,2}^{-,3}$	$c_{[3]}^3$	$\nu_{1,2}^{-,3}$	$\nu_{[3]}^3 = -1$	
	$c_{1,2}^{+,3}$	$c_{[4]}^3$	$\nu_{1,2}^{+,3}$	$\nu_{[4]}^3 = +1$	

539 5.3.3. Overlap of Elementary Failure Domains

540 The definition of the *overlap count* vector is introduced to compute the number of overlapping

541 elementary failure domains per segment, defined as:

$$\mathbf{m}^j = \{m_1^j, \dots, m_{n_S}^j\}^T, \quad n_S = n_F + 1, \quad (46)$$

542 where n_S corresponds to the number of segments within the line l_j , and m_r^j is defined as:

$$m_r^j = \begin{cases} m_r^{j,n}, & r = 1, \dots, \lambda_N^j, \\ m_0, & r = \lambda_N^j + 1, \dots, \lambda_P^j, \\ m_r^{j,p}, & r = \lambda_P^j + 1, \dots, n_S, \end{cases} \quad (47)$$

543 where each $m_r^{j,n}$ and $m_r^{j,p}$ represents the number of active elementary failure domains within
 544 the r -th integration segment of $\Omega_j^{-,\parallel}(\mathbf{y})$ and $\Omega_j^{+,\parallel}(\mathbf{y})$, respectively, and m_0 is an integer number
 545 different from zero to facilitate the numerical implementation. These are computed recursively by
 546 accumulating the transition indicators $\nu_{[h]}^j$ across the segments, as follows:

$$m_r^{j,n} = n_\eta n_T + 1 + \sum_{h=1}^r \nu_{[h]}^j, \quad r = 1, \dots, \lambda_N^j, \quad (48)$$

$$m_r^{j,p} = n_\eta n_T + \sum_{h=1}^r \nu_{[h]}^j, \quad r = \lambda_P^j + 1, \dots, n_S, \quad (49)$$

547 where the +1 term in Equation (48) ensures that the first segment starts with $n_\eta n_T$ elementary
 548 failure domains active. This term is not required in Equation (49) due to the way the transition
 549 indicators are defined. The elements in \mathbf{m}^j determine the denominator used in the integrand of
 550 Equation (24). Intermediate segments between λ_N^j and λ_P^j are represented by m_0 even if they
 551 contain elementary failure domains, since they do not contribute to the integration domain $\Omega_j^{\parallel}(\mathbf{y})$.
 552 These segments can be identified by introducing a boolean *selector vector* $\boldsymbol{\kappa}^j = \{\kappa_r^j, \dots, \kappa_{n_S}^j\}^T \in$
 553 $\{0, 1\}$ with $r = 1, \dots, n_S$, yielding a vector of dimensions $n_S \times 1$, defined as:

$$\kappa_r^j = \begin{cases} 1, & r \in \{1, \dots, \lambda_N^j\} \cup \{\lambda_P^j + 1, \dots, n_S\}, \\ 0, & \text{otherwise,} \end{cases} \quad (50)$$

554 which indicates the segments that contribute to the integration over Ω_j^{\parallel} . Figure 9 illustrates the
 555 values of the overlap count vector \mathbf{m}^j for the three lines of the example in Figure 8. The values

556 shown correspond to integration segments where the selector vector satisfies $\kappa_r^j = 1$. It is worth
 557 noting that, for all lines, the first and last integration segments have an overlap count of $n_\eta n_T = 2$.
 558 For lines l_1 and l_2 , there are three valid integration segments, while for line l_3 there are four. In
 559 addition, it is straightforward to note that the number of negative and positive segments varies
 560 for each line.

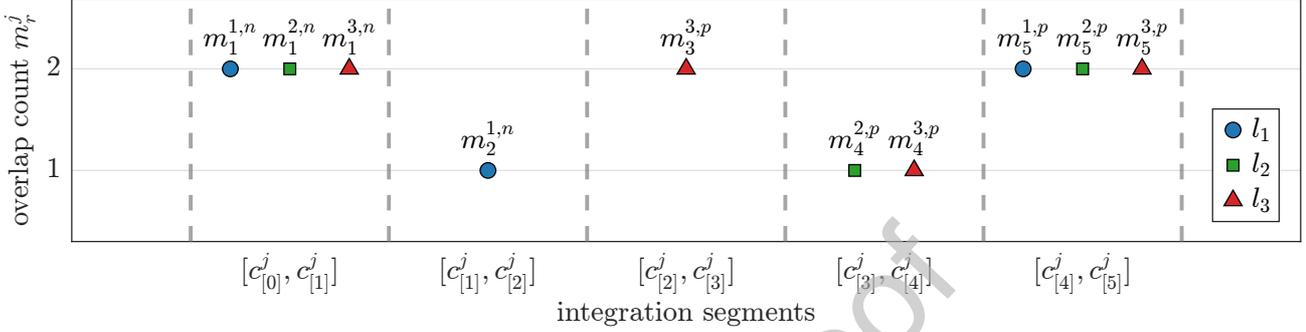


Figure 9: Overlap count vector calculation for the example presented in Figure 8.

561 5.4. Estimation of the Effective Contribution and Its Sensitivity

562 The application of the previous definitions leads to a more compact expression for the estima-
 563 tion of the effective contribution $\tilde{p}_{s_j}(\mathbf{y})$ shown in Equation (24), expressed as:

$$\tilde{p}_{s_j}(\mathbf{y}) = \sum_{r=1}^{n_S} \frac{\kappa_r^j}{m_r^j} \left(\Phi(c_{[r]}^j) - \Phi(c_{[r-1]}^j) \right). \quad (51)$$

564 Analogously, the derivative of the estimation of the effective contribution with respect to the
 565 design parameter y_q , based on Equation (30), becomes:

$$\frac{\partial}{\partial y_q} (\tilde{p}_{s_j}(\mathbf{y})) = \sum_{r=1}^{n_S} \frac{\kappa_r^j}{m_r^j} \left(\phi(c_{[r]}^j) \frac{\partial c_{[r]}^j}{\partial y_q} - \phi(c_{[r-1]}^j) \frac{\partial c_{[r-1]}^j}{\partial y_q} \right), \quad (52)$$

566 where $\partial c_{[r]}^j / \partial y_q$ denotes the partial derivative of the distance $c_{[r]}^j$ with respect to the design pa-
 567 rameter y_q , the calculation of which is detailed in Section 5.5. It is important to highlight that
 568 with the convention $m_r^j = m_0$ whenever $\kappa_r^j = 0$, all the denominators in Equations (51) and (52)
 569 remain nonzero. For completeness, Appendix C provides a detailed derivation of Equation (52)
 570 from Equation (30). It is noted that, under the adopted formulation, Equation (52) can also be
 571 obtained by direct differentiation of Equation (51).

572 The estimation of the effective contribution and its derivative, using Equations (51) and (52),
 573 allows for the computation of the first excursion probability via Equation (24), and its sensitivity

574 via Equation (30). For a given exploration line l_j , both estimators can be evaluated simultaneously.
 575 This constitutes a significant advantage, as the sensitivity estimation becomes a byproduct of the
 576 first excursion probability estimation.

577 5.5. Partial Derivatives

578 The sensitivity of the first excursion probability with respect to a design parameter, as shown
 579 in Equation (29), depends on the partial derivative of the effective contribution of the failure
 580 probability with respect to that parameter. According to Leibniz's rule [52] and Equation (52),
 581 this quantity also depends on the partial derivative of the crossing distance with respect to the same
 582 design parameter. This partial derivative can be obtained by directly differentiating Equation (38)
 583 and (39) with respect to the design parameter y_q :

$$\frac{\partial c_{i,k}^{-,j}(\mathbf{y})}{\partial y_q} = \frac{-\frac{\partial \mathbf{a}_{i,k}(\mathbf{y})^T}{\partial y_q} \mathbf{R}_{s_j} \mathbf{z}_j^\perp - c_{i,k}^{-,j} \frac{\partial \mathbf{a}_{i,k}(\mathbf{y})^T}{\partial y_q} \boldsymbol{\alpha}_{s_j}}{\eta_i^\parallel(t_k, \mathbf{y}, \boldsymbol{\alpha}_{s_j})}, \quad (53)$$

$$\frac{\partial c_{i,k}^{+,j}(\mathbf{y})}{\partial y_q} = \frac{-\frac{\partial \mathbf{a}_{i,k}(\mathbf{y})^T}{\partial y_q} \mathbf{R}_{s_j} \mathbf{z}_j^\perp - c_{i,k}^{+,j} \frac{\partial \mathbf{a}_{i,k}(\mathbf{y})^T}{\partial y_q} \boldsymbol{\alpha}_{s_j}}{\eta_i^\parallel(t_k, \mathbf{y}, \boldsymbol{\alpha}_{s_j})}, \quad (54)$$

584 where $\partial \mathbf{a}_{i,k}(\mathbf{y})^T / \partial y_q$ denotes the derivative of the vector $\mathbf{a}_{i,k}(\mathbf{y})$ with respect to the parameter y_q .
 585 Evaluating Equation (53) or (54) is equivalent to performing two sensitivity analyses. In practice,
 586 for a given line l_j , these analyses are in addition to the two dynamic analyses required, as shown
 587 in Section 5.2. In order to facilitate the implementation, the derivatives $\partial c_{i,k}^{-,j} / \partial y_q$ and $\partial c_{i,k}^{+,j} / \partial y_q$,
 588 as obtained from Equations (53) and (54), are grouped into the vector \mathcal{D}^j :

$$\mathcal{D}^j = \left\{ \frac{\partial c_{i,k}^{-,j}}{\partial y_q}, \frac{\partial c_{i,k}^{+,j}}{\partial y_q} \right\}^T, \quad \text{for } i = 1, \dots, n_\eta; \quad k = 1, \dots, n_T. \quad (55)$$

589 Following the same ordering applied to \mathcal{C}^j , the elements of \mathcal{D}^j are rearranged to form the sorted
 590 derivative vector:

$$\tilde{\mathcal{D}}_{\text{sort}}^j = \left\{ \frac{\partial c_{[0]}^j}{\partial y_q}, \frac{\partial c_{[1]}^j}{\partial y_q}, \dots, \frac{\partial c_{[n_F]}^j}{\partial y_q}, \frac{\partial c_{[n_F+1]}^j}{\partial y_q} \right\}^T. \quad (56)$$

591 The vector $\mathbf{a}_{i,k}(\mathbf{y})$, defined in Section 2.2, depends on the i -th unit impulse response function
 592 $h_i(t, \mathbf{y})$. Its derivative can be obtained by directly differentiating Equation (6), resulting in:

$$\frac{\partial \mathbf{a}_{i,k}(\mathbf{y})}{\partial y_q} = \sum_{m=1}^k \Delta t \epsilon_m \frac{\partial h_i(t_k - t_m, \mathbf{y})}{\partial y_q} \boldsymbol{\psi}_m, \quad (57)$$

593 where $\partial h_i(t, \mathbf{y})/\partial y_q$ is the partial derivative of the i -th unit impulse response function with respect
 594 to the design parameter y_q . The unit impulse response in Equation (6) is determined by the mass
 595 and damping matrices, the coupling vector, and the spectral properties of the system. Accordingly,
 596 its partial derivative with respect to a design parameter is computed via the chain rule, as described
 597 in Appendix D. It is important to note that, in this work, the partial derivatives of the eigenvectors
 598 and eigenvalues are computed using the method proposed in [39].

599 5.6. Summary

600 The first excursion probability and its sensitivity with respect to a design parameter, using
 601 multidomain Line Sampling applied to a linear system subjected to Gaussian loading, can be
 602 calculated as follows:

- 603 1. Model the Gaussian loading \mathbf{p} using the Karhunen-Loève expansion (see Equation (1)).
- 604 2. Specify the mass, damping, and stiffness matrices \mathbf{M} , \mathbf{C} , and \mathbf{K} , the design parameter
 605 vector \mathbf{y} , and the response thresholds \mathbf{b} (see Equation (2)).
- 606 3. Assemble the augmented matrices $[\mathbf{M}_a]$, $[\mathbf{K}_a]$, and the vector $\{\mathbf{g}_a\}$ (see Equations (3) and
 607 (4)).
- 608 4. Compute the vectors $\mathbf{a}_{i,k}(\mathbf{y})$ and their derivatives $\partial \mathbf{a}_{i,k}(\mathbf{y})/\partial y_q$ (see Equations (6) and (57)).
- 609 5. Multidomain Line Sampling loop for $j = 1, \dots, N$:
 - 610 (a) Generate a sample \mathbf{z}_j^\perp in the orthogonal space and define the associated line l_j (see
 611 Section 5.1).
 - 612 (b) For the line l_j , compute distances $c_{i,k}^{-,j}$ and $c_{i,k}^{+,j}$ (see Equations (38) and (39)) and build
 613 the vector \mathcal{C}^j (see Equation (40)). Analogously, compute the transition indicators $\nu_{i,k}^{-,j}$
 614 and $\nu_{i,k}^{+,j}$ and build the vector \mathcal{V}^j (see Equation (42)).
 - 615 (c) Build the sorted vectors $\tilde{\mathcal{C}}_{\text{sort}}^j$ and $\tilde{\mathcal{V}}_{\text{sort}}^j$ (see Equations (41) and (43)).
 - 616 (d) Identify the negative and positive indices λ_N^j and λ_P^j (Equation (44)).
 - 617 (e) For the line l_j , compute the derivative of the distances $\partial c_{i,k}^{-,j}/\partial y_q$ and $\partial c_{i,k}^{+,j}/\partial y_q$ (see
 618 Equations (53) and (54)) and build the vectors \mathcal{D}^j (see Equation (55)) and $\tilde{\mathcal{D}}_{\text{sort}}^j$ (see
 619 Equation (56)).

(f) Build the overlap count vector \mathbf{m}^j and the selector boolean vector $\boldsymbol{\kappa}^j$ (see Equations (48), (49) and (50)).

(g) Evaluate the effective contribution $\tilde{p}_{s_j}(\mathbf{y})$ (see Equation (51)) and its derivative $\partial(\tilde{p}_{s_j}(\mathbf{y}))/\partial y_q$ (see Equation (52)).

6. Compute the first excursion probability estimator $\tilde{p}_F(\mathbf{y})$ (Equation (23)) and its sensitivity $\partial\tilde{p}_F(\mathbf{y})/\partial y_q$ (Equation (29)) by averaging the contributions over all N lines.

To facilitate numerical implementation, the steps described above are also provided in pseudo-code form, as shown in Appendix E.

6. Examples

6.1. Overview of the examples

This section showcases the application of multidomain Line Sampling to two different problems. The first example involves a quarter-car model represented by a two-degree-of-freedom system with non-proportional damping. The second example involves a large-scale finite element model of a 16-story building assuming proportional damping. The results are analyzed and compared with existing methods in the literature to demonstrate the efficiency of the proposed method.

All numerical simulations were performed in MATLAB R2023a on a laptop equipped with an Intel® Core™ i9–12900H processor (20 cores, up to 4.9 GHz) and 32 GB RAM, running Windows 11 (64-bit). All computations were executed on the CPU.

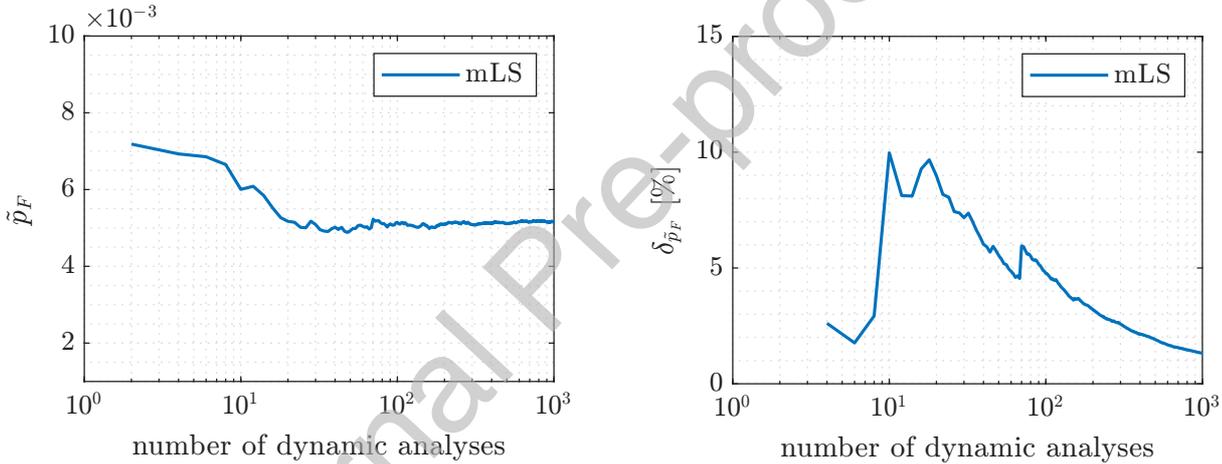
6.2. Example 1: Quarter-car model

The first example is an idealization of a car's suspension, modeled as a two-degree-of-freedom system and commonly referred to as a quarter-car model, as shown in Figure 11. This model is based on an example presented in [53]. The behavior of the system is governed by the following pair of ordinary differential equations:

$$\begin{aligned} \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix} \begin{Bmatrix} \ddot{x}_1(t, \mathbf{y}, \mathbf{z}) \\ \ddot{x}_2(t, \mathbf{y}, \mathbf{z}) \end{Bmatrix} + \begin{bmatrix} c_1 + c_2 & -c_2 \\ -c_2 & c_2 \end{bmatrix} \begin{Bmatrix} \dot{x}_1(t, \mathbf{y}, \mathbf{z}) \\ \dot{x}_2(t, \mathbf{y}, \mathbf{z}) \end{Bmatrix} + \\ \begin{bmatrix} k_1 + k_2 & -k_2 \\ -k_2 & k_2 \end{bmatrix} \begin{Bmatrix} x_1(t, \mathbf{y}, \mathbf{z}) \\ x_2(t, \mathbf{y}, \mathbf{z}) \end{Bmatrix} = \begin{Bmatrix} k_1 w(t) + c_1 \dot{w}(t) \\ 0 \end{Bmatrix}, \end{aligned} \quad (58)$$

643 where the unsprung and sprung masses of the car are $m_1 = 15$ kg and $m_2 = 290$ kg, respectively.
 644 The tire stiffness is $k_1 = 191000$ N/m, and the suspension stiffness is $k_2 = 16200$ N/m. In addition,
 645 the non-proportional damping matrix is composed of the tire damping coefficient $c_1 = 100$ Ns/m
 646 and the suspension damping coefficient $c_2 = 2500$ Ns/m.

647 The road profile $w(t, \mathbf{z})$ serves as the load input for the model. It is modeled as a zero-mean
 648 stationary Gaussian random field with a squared exponential covariance kernel, a correlation
 649 length of 3 m, and a standard deviation of 0.01 m. The car is assumed to travel at 25 m/s over
 650 a distance of 125 m. The road profile is discretized into 1001 equidistant points, corresponding
 651 to a total of $n_{KL} = 1001$ terms, which results in 1001 elementary failure domains. The expansion
 652 is not truncated, since the aim here is to illustrate the performance of the proposed sensitivity
 653 analysis. Time is discretized into intervals of $\Delta t = 0.005$ s, for a total duration of 5 s.



(a) Evolution of the first excursion probability estimator with respect to the number of dynamic analyses.

(b) Coefficient of variation of the first excursion probability estimator with respect to the number of dynamic analyses.

Figure 10: **Example 1:** Evolution of the first excursion probability estimator using multidomain Line Sampling (mLS).

654 The car's suspension design is paramount to fulfilling comfort requirements while driving over
 655 a road profile. Two key response variables can be used to assess this: the acceleration of the
 656 sprung mass and the suspension stroke, defined as the relative displacement between the sprung
 657 and unsprung masses. In this example, the latter is considered, and the response of interest is the
 658 displacement of mass m_2 with respect to mass m_1 , expressed as $\eta(t, \mathbf{y}, \mathbf{z}) = |x_2(t, \mathbf{y}, \mathbf{z}) - x_1(t, \mathbf{y}, \mathbf{z})|$.
 659 The threshold level associated with the response of interest is set to $b = 3.5 \times 10^{-2}$ m. The first
 660 excursion probability for the problem is estimated using mLS, which gives $\tilde{p}_F = 5.1 \times 10^{-3}$, as
 661 shown in Figure 10a. Furthermore and as observed from Figure 10b, the coefficient of variation

662 of the probability estimator is quite low, as it never exceeds 10%.

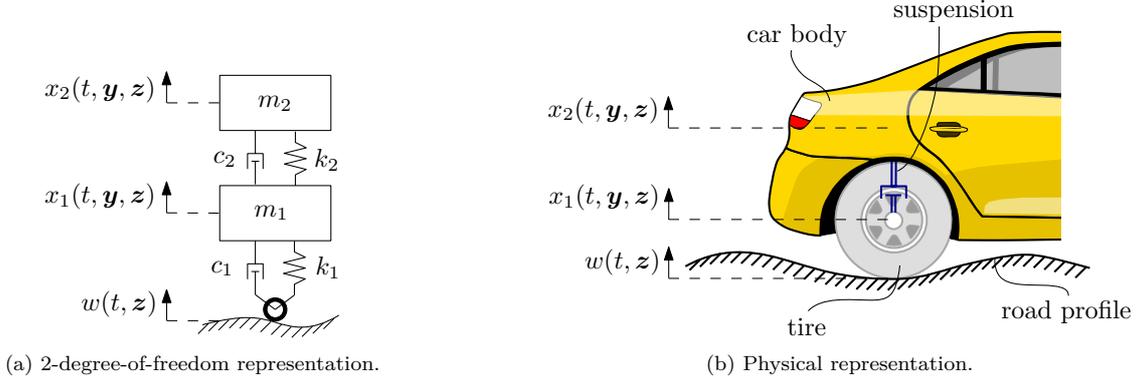


Figure 11: **Example 1:** Quarter-car model.

663 The objective is to estimate the sensitivity of the first excursion probability using mLS with
 664 respect to the design vector $\mathbf{y} = [y_1, y_2]^T$, where $y_1 = m_2$ and $y_2 = k_2$, and to establish a
 665 comparison with DIS [8] and DDM [29]. The comparison is based on the number of dynamic
 666 analyses required to obtain a single estimate of the sensitivity and its corresponding coefficient
 667 of variation. It is important to note that mLS requires two dynamic analyses and two sensitivity
 668 analyses per simulation, while DIS and DDM require one dynamic analysis and one sensitivity
 669 analysis.

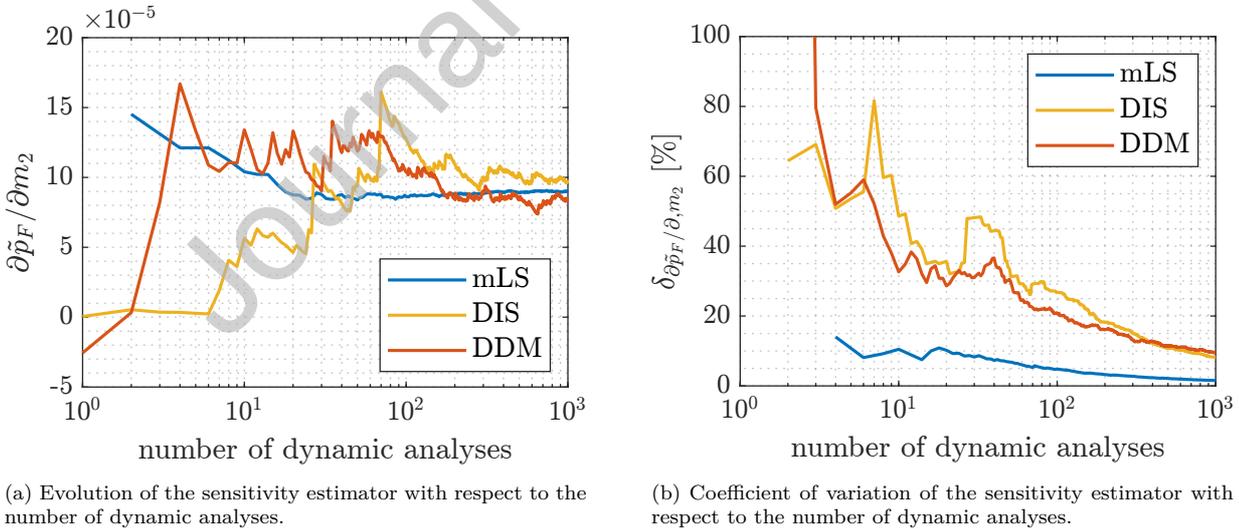


Figure 12: **Example 1:** Comparison of sensitivity results associated with m_2 using Directional Importance Sampling (DIS), the Domain Decomposition Method (DDM), and multidomain Line Sampling (mLS).

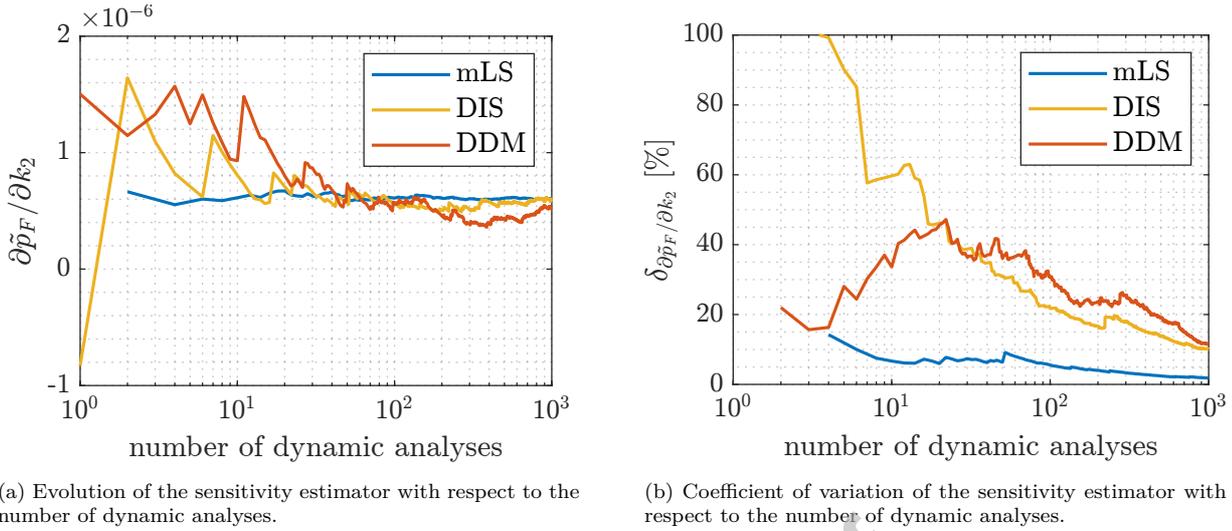


Figure 13: **Example 1:** Comparison of sensitivity results associated with k_2 using Directional Importance Sampling (DIS), the Domain Decomposition Method (DDM), and multidomain Line Sampling (mLS).

670 The evolution of the sensitivity estimators associated with m_2 and k_2 is presented in Figures
 671 12 and 13, respectively, in terms of the number of dynamic analyses performed per simulation.
 672 A total of 1×10^3 dynamic analyses is considered in this example. For the design parameter
 673 m_2 , Figure 12a shows that all methods converge to a similar value for the sensitivity estimator,
 674 with the mLS estimator exhibiting remarkably stable behavior. This is confirmed in Figure 12b,
 675 where the coefficients of variation of the DDM and DIS estimators reach approximately 10%
 676 with 1×10^3 dynamic analyses, whereas the mLS estimator achieves the same level with only
 677 20 dynamic analyses. Similarly, for the second parameter k_2 , the sensitivity estimator computed
 678 using mLS achieves a coefficient of variation of 10% after just 50 dynamic analyses, whereas the
 679 other methods again require approximately 1×10^3 dynamic analyses. Note that the discrepancies
 680 between the results at smaller numbers of dynamic analyses are purely due to random sampling
 681 variability, not to bias. All methods yield unbiased estimators, which converge as more dynamic
 682 analyses are accumulated. It is worth noting that, in Figures 12 and 13, the coefficients of variation
 683 obtained with mLS are significantly smaller than those of DIS and DDM and exhibit only a mild
 684 decrease with the number of dynamic analyses. This behavior is a direct consequence of the
 685 way mLS explores the failure domain: each exploration line typically intersects the elementary
 686 failure domain (associated with the effective contribution being estimated) perpendicular to their
 687 boundary, thereby capturing more relevant sensitivity information. In contrast, the directional
 688 approaches (DIS and DDM) explore the failure domain using lines anchored at the origin, so that,
 689 when these lines intersect the elementary failure domains, the effect of parameter perturbations

690 is less pronounced and is therefore captured less efficiently in the sensitivity estimates.

691 In Table 2, the computational effort of the different methods is compared using the speed-up
 692 factor \mathbf{S}_u , which quantifies the efficiency gains of the proposed technique. The speed-up factor
 693 is computed as the ratio between the number of dynamic analyses $N^{(q)}$ required by the reference
 694 method and that required by mLS for each parameter y_q , to reach a coefficient of variation of
 695 10%. The last column presents the mean number of required dynamic analyses for estimating one
 696 sensitivity, given by $N_{\text{mean}} = (\sum_{q=1}^{n_Y} N^{(q)})/n_Y$. For the sensitivity with respect to m_2 , mLS achieves
 697 a speed-up of 32.0 over DIS and 43.0 over DDM. Similarly, for k_2 , the respective speed-up factors
 698 are 19.9 and 30.7. These results confirm that mLS substantially reduces the computational effort
 699 required, while preserving the accuracy of the sensitivity estimates. A similar behavior is observed
 700 when considering the average computational cost, where mLS achieves a speed-up of 23.33 over
 701 DIS and 34.21 over DDM based on the mean number of dynamic analyses.

Method	$\partial\tilde{p}_F/\partial m_2$	$N^{(1)}$	$\partial\tilde{p}_F/\partial k_2$	$N^{(2)}$	\mathbf{N}_{mean}
mLS	8.75×10^{-5}	20	6.29×10^{-7}	50	35.00
DIS	1.03×10^{-4}	640	5.99×10^{-7}	993	816.50
DDM	8.82×10^{-5}	859	6.05×10^{-7}	1536	1197.50
\mathbf{S}_u (mLS vs DIS)		32.00		19.86	23.33
\mathbf{S}_u (mLS vs DDM)		42.95		30.72	34.21

Table 2: **Example 1:** Comparison of sensitivity estimates, number of dynamic analyses ($N^{(q)}$), and speed-up (\mathbf{S}_u) of mLS versus other methods, for a target coefficient of variation of 10%.

702 Furthermore, the results were validated against finite difference (FD) estimates. For the esti-
 703 mation, a central-difference scheme is used with a total of 2×10^6 independent directional sampling
 704 samples (split equally between forward and backward steps), applying a relative perturbation of
 705 $\Delta_i = 0.01 y_i$ to each parameter. The same random seeds were used in the forward and backward
 706 simulations in order to reduce estimator variance. Table 3 presents the comparison, where the
 707 proposed method achieves a coefficient of variation of 1%, revealing excellent agreement between
 708 the sensitivity estimates obtained using mLS and FD.

Method	$\partial\tilde{p}_F/\partial m_2$	$\partial\tilde{p}_F/\partial k_2$
mLS	8.80×10^{-5}	6.14×10^{-7}
FD	8.96×10^{-5}	6.25×10^{-7}

Table 3: Comparison of sensitivity estimates using mLS and FD.

709 To further examine the numerical behavior of the estimator, we investigate how the compu-
710 tational effort required for estimating the sensitivities with respect to the sprung mass m_2 and
711 the suspension stiffness k_2 depends on the magnitude of the first-excursion probability. For each
712 estimator, Figure 14 shows the mean number of dynamic analyses required by mLS to reach a
713 coefficient of variation of 5%. At each probability level, this mean is computed over 500 independ-
714 ent simulations. Different failure probabilities are generated by varying the response threshold
715 in uniform steps, covering $p_F \in [10^{-7}, 10^{-1}]$. The circles and squares correspond to the simulated
716 probability levels, and the line segments are drawn for visual guidance. For $10^{-7} \leq p_F \leq 10^{-2}$,
717 the required number of dynamic analyses grows only slowly: from about 33 to 57 for the esti-
718 mator associated with m_2 , and from about 36 to 81 for the estimator associated with k_2 . For
719 $p_F > 10^{-2}$, the computational effort increases markedly as the failure probability grows. These
720 findings suggest that, for this example, the method becomes more efficient as the failure proba-
721 bility decreases. This trend is consistent with the observations reported in [6] for a comparable
722 simulation approach.

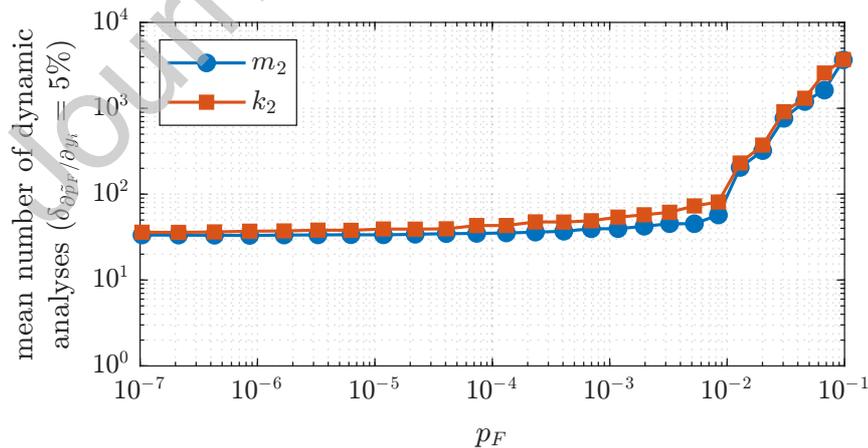


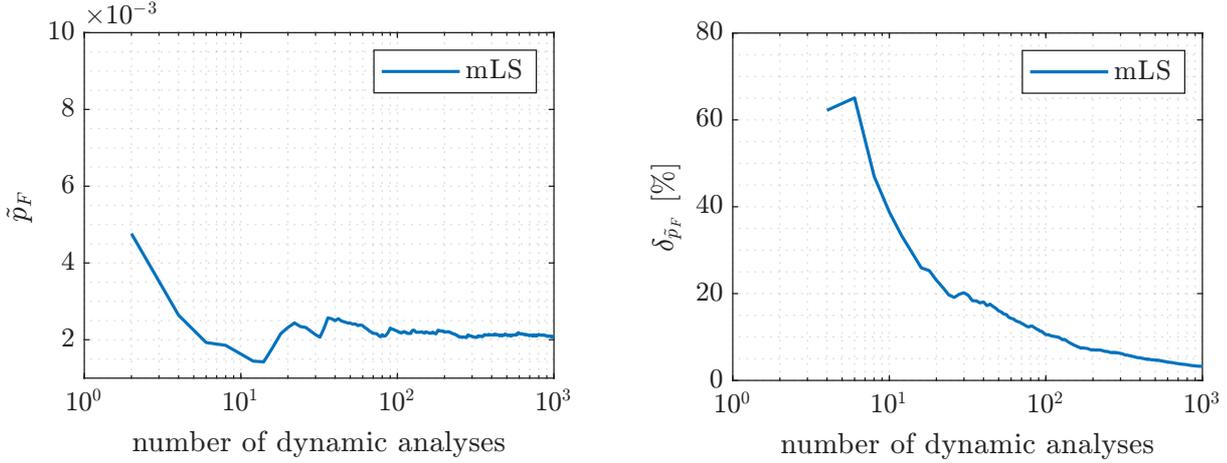
Figure 14: **Example 1:** Mean number of dynamic analyses required by mLS to reach a coefficient of variation of 5% for the sensitivity estimator associated with the parameter k_2 , as a function of the first excursion probability p_F .

723 Estimating the sensitivity with respect to the parameters m_2 and k_2 is crucial for analyzing

724 vehicle comfort when driving over a road profile. The results shown in Figure 12 indicate that an
725 increase in the car's body mass leads to a higher failure probability of the system. This increase
726 in mass reduces the system's natural frequency, which in turn results in larger displacements
727 due to low-frequency perturbations in the road profile. Moreover, as illustrated in Figure 13,
728 increasing the stiffness k_2 leads to an increase in the system's failure probability. This occurs
729 because a stiffer suspension makes the car react more to irregularities in the road, which increases
730 the displacements and, as a result, the failure probability of the system. From an engineering
731 perspective, the sensitivity estimates provided by the analysis offer guidance at the design stage
732 for tuning the sprung mass and suspension stiffness in order to satisfy comfort and serviceability
733 requirements.

734 6.3. Example 2: 16-story reinforced concrete building

735 The second example corresponds to a 3D finite element model of a 16-story reinforced concrete
736 (RC) building, involving 29466 degrees of freedom. This model is based on the example presented
737 in [37] and is illustrated in Figure 16. Each typical floor of the building includes columns and
738 beams, modeled as frame elements, as well as slabs, external walls, and a wall core, modeled
739 as shell elements, as shown in Figure 16a. The frames are represented by black lines; the slabs
740 by light gray surfaces; the external wall by pale yellow surfaces; and the RC shear wall core by
741 light blue surfaces. To highlight the most relevant aspects of the problem, the extrusion of the
742 structural elements has been intentionally omitted in Figure 16. The Young's modulus for the
743 reinforced concrete is taken as $E = 2.5 \times 10^{10}$ N/m². The floor height is 3.40 m at the first level
744 and 3.24 m from the second to the sixteenth levels. The slabs and shear walls have thicknesses of
745 18 cm and 40 cm, respectively. Moreover, the structure is evaluated for serviceability purposes,
746 assuming linear elastic behavior. The first 100 modes are retained for the dynamic analysis to
747 ensure an accurate representation of the response. Extending the scope demonstrated in the
748 previous example, the proposed method can also be applied to a case with proportional damping.
749 In this case, 5% classical damping is considered for all modes.



(a) Evolution of the first excursion probability estimator with respect to the number of dynamic analyses.

(b) Coefficient of variation of the first excursion probability estimator with respect to the number of dynamic analyses.

Figure 15: **Example 2:** Evolution of the first excursion probability estimator using multidomain Line Sampling (mLS).

750 The stochastic ground acceleration acting on the building is modeled as a modulated discrete
 751 white noise process, that is passed through a Clough-Penzien filter [54]. It has a spectral intensity
 752 of $S = 3 \times 10^{-3} \text{ m}^2/\text{s}^3$, a time duration of $T = 10 \text{ s}$, and is discretized into $n_T = 1001$ time steps
 753 with a duration of $\Delta t = 0.01 \text{ s}$. The modulation function $m(t)$ associated with the modulated
 754 (hence nonstationary) discrete white-noise input is given by:

$$m(t) = \begin{cases} (t/5)^2 & 0 \leq t \leq 5 \text{ [s]} \\ 1 & 5 < t \leq 6 \text{ [s]} \\ e^{-(t-6)^2} & t > 6 \text{ [s]} \end{cases} . \quad (59)$$

755 The Clough-Penzien filter is characterized by natural circular frequencies $\omega_{g,1} = 15.6 \text{ rad/s}$ and
 756 $\omega_{g,2} = 1.0 \text{ rad/s}$, and damping ratios $\zeta_{g,1} = 0.6$ and $\zeta_{g,2} = 0.9$. The Karhunen-Loève representation
 757 uses the same number of terms as time instants, with $n_{KL} = 1001$, since the focus of this example
 758 is to demonstrate the performance of the proposed sensitivity analysis rather than to investigate
 759 the effect of the truncation order of the expansion on the results. In addition, the structure is
 760 assumed to start from rest.

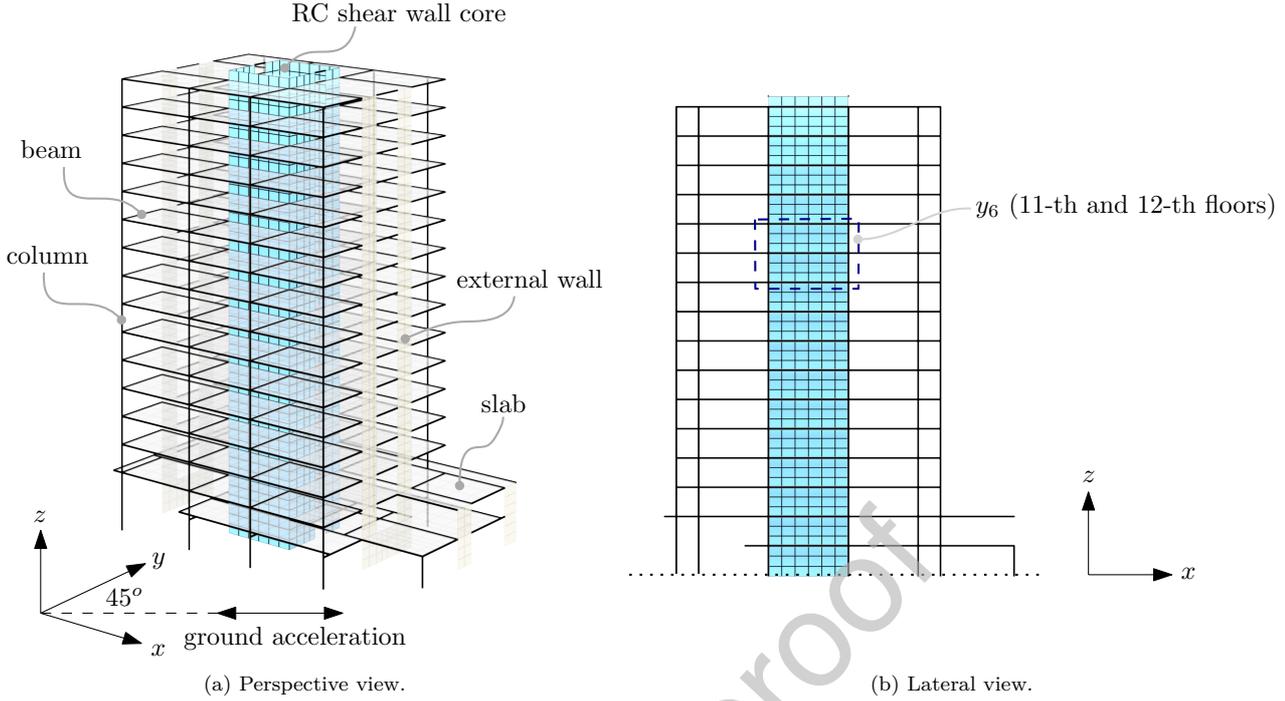


Figure 16: **Example 2:** 16-story reinforced concrete building model.

761 The responses of interest correspond to the interstory drifts of the sixteen floors of the build-
 762 ing in both the x and y directions, measured from the geometrical center of each floor during
 763 the application of the stochastic loading. Therefore, there are a total of $n_\eta = 32$ responses of
 764 interest. The threshold limit is set to 0.2% of the floor height, which corresponds to $b_{1-2} = 0.0068$
 765 m for the first floor and $b_{3-32} = 0.0065$ m for the remaining levels. This criterion aligns well
 766 with serviceability design purposes. The first excursion probability is estimated using mLS and
 767 involves a large number of elementary failure domains ($n_\eta n_{KL} = 32 \times 1001 = 32032$), and the
 768 evolution of the corresponding estimator with respect to the number of dynamic analyses is shown
 769 in Figure 15a, resulting in $\tilde{p}_F = 2.1 \times 10^{-3}$. Figure 15b presents the evolution of the coefficient of
 770 variation of the probability estimate. Although this coefficient of variation is not as small as that
 771 observed in the previous example, its decreasing trend is remarkably stable. Indeed, with only
 772 about 100 simulation runs, it is possible to estimate a relatively small failure probability with a
 773 coefficient of variation of about 10%.

774 The objective is to estimate the sensitivity of the first excursion probability with respect to the
 775 vector $\mathbf{y} = [y_1, \dots, y_8]^T$, where each component y_q represents the thickness of the RC shear wall
 776 core at the $(2q - 1)$ -th and $(2q)$ -th floors. This is illustrated in Figure 16b, where, for instance,
 777 the parameter y_6 corresponds to the thickness of the RC core walls on the 11th and 12th floors.

778 The information offered by the sought sensitivities has a very insightful impact in the context of
 779 decision making, for instance, reliability-based design [33, 27].

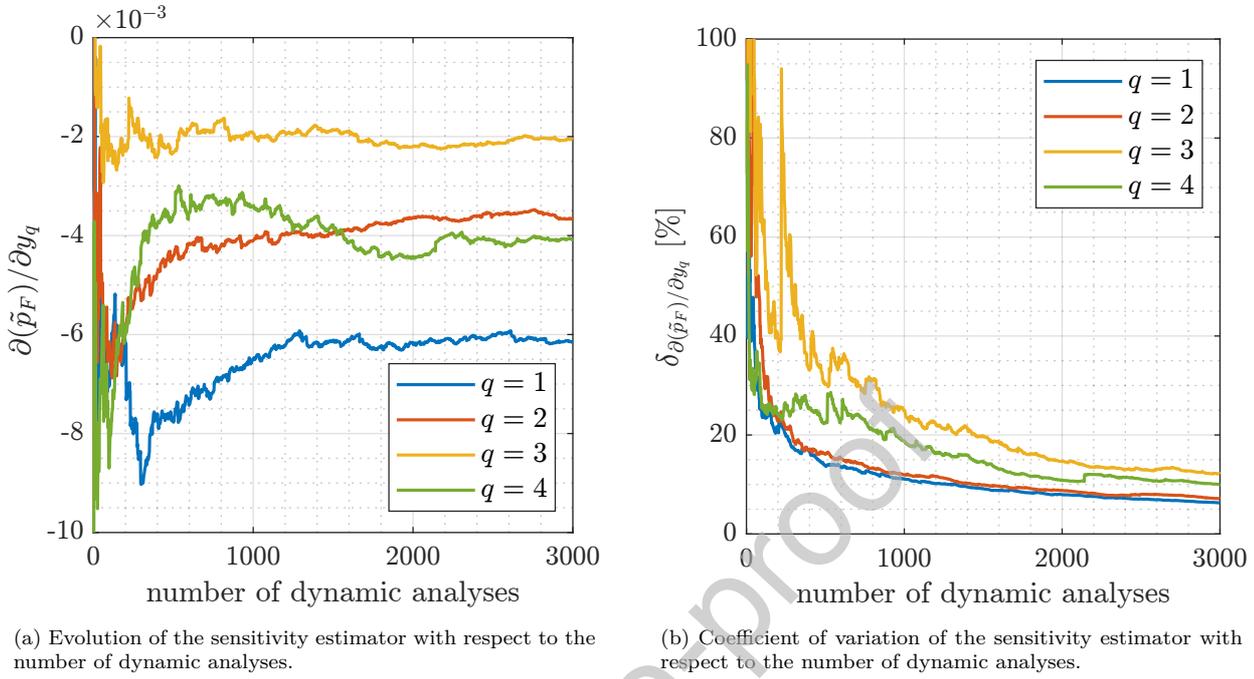


Figure 17: **Example 2:** Sensitivity results associated with $y_q, q = [1, 2, 3, 4]$ using multidomain Line Sampling (mLS).

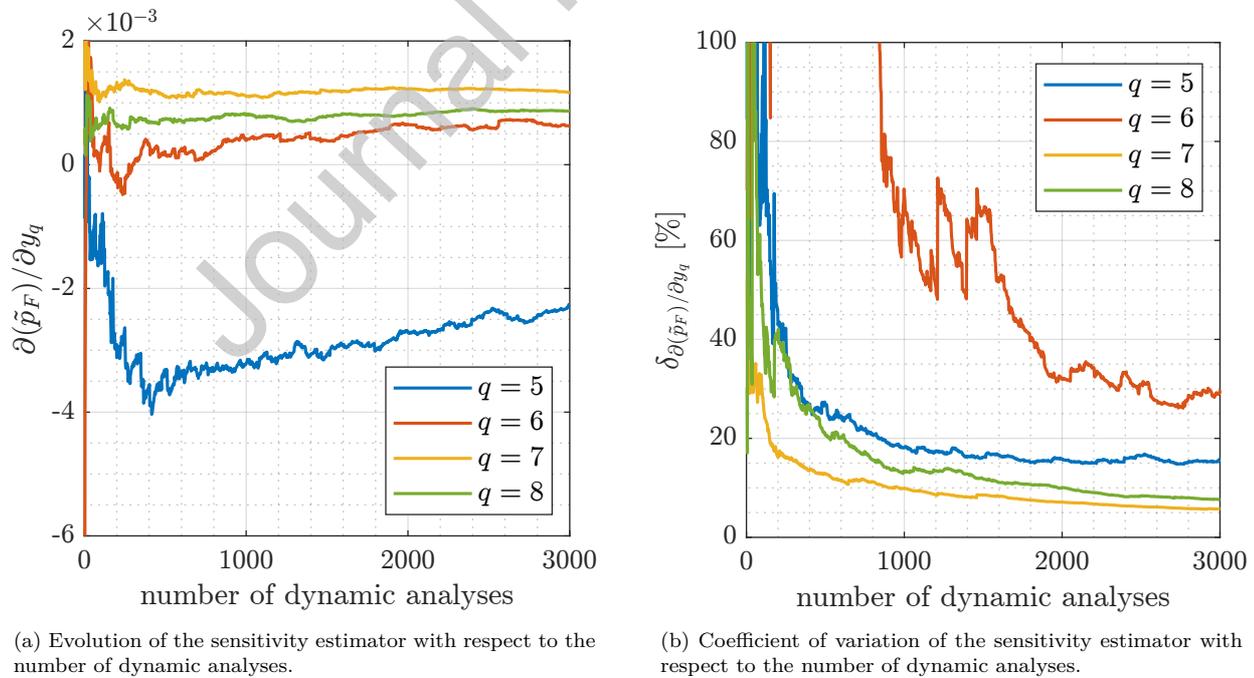


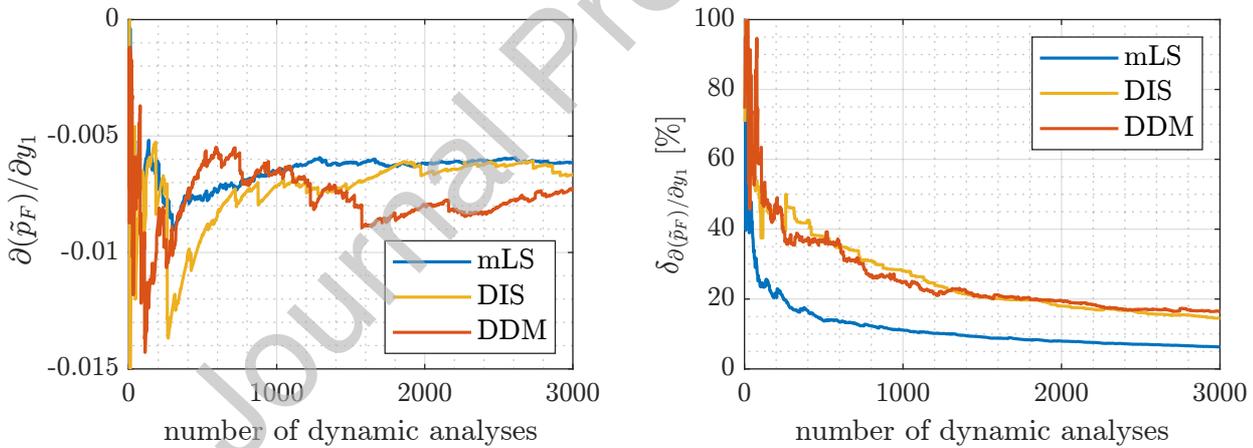
Figure 18: **Example 2:** Sensitivity results associated with $y_q, q = [5, 6, 7, 8]$ using multidomain Line Sampling (mLS).

780 The sensitivities were estimated using mLS and the results with respect to parameters y_q , $q =$
781 $1, \dots, 4$ are shown in Figures 17, while the results with respect to the parameters y_q , $q = 5, \dots, 8$
782 are illustrated in Figure 18. The evolution of the sensitivities with respect to the number of dy-
783 namic analyses, shown in Figure 17a, indicates that increasing y_q , $q = 1, \dots, 4$, leads to a decrease
784 in the failure probability of the system. In addition, Figure 17b shows that the calculation of the
785 sensitivities is computationally efficient, requiring 250 and 292 dynamic analyses for the estima-
786 tors associated with y_1 and y_2 , respectively, to reach a coefficient of variation of 20%. Moreover,
787 the estimators associated with y_3 and y_4 require 1450 and 978 dynamic analyses, respectively, to
788 achieve the same coefficient of variation. The evolution of the sensitivities with respect to the
789 number of dynamic analyses, shown in Figure 18a, indicates that increasing y_5 leads to a decrease
790 in the failure probability of the system, while increasing the values of y_q , $q = 6, 7, 8$, raises the
791 failure probability. In this case, to reach a coefficient of variation of 20%, the sensitivity related
792 to y_7 requires 148 dynamic analyses, while the one related to y_8 requires 622 dynamic analyses.
793 In addition, to reach a coefficient of variation of 20%, the sensitivity associated with y_5 requires
794 856 dynamic analyses, while the sensitivity related to y_6 requires 5402 dynamic analyses.

795 In order to demonstrate the efficiency of mLS compared with existing methods in the liter-
796 ature, such as DIS [8] and DDM [29], the results for the sensitivity related to y_1 are illustrated
797 in Figure 19. It can be observed from Figure 19a that the behavior of the estimator calculated
798 using mLS is notably more stable with fewer dynamic analyses compared to the other methods.
799 This is confirmed in Figure 19b, where approximately 250 dynamic analyses are required for mLS
800 to reach a coefficient of variation of 20%, while DIS and DDM require a total of 1632 and 1801
801 dynamic analyses, respectively, to achieve the same coefficient of variation. The observed differ-
802 ences between the estimates at lower numbers of dynamic analyses reflect statistical fluctuations
803 and are not indicative of bias. All estimators are theoretically unbiased, and the estimates con-
804 verge to the true values as the number of dynamic analyses increases. In addition, the speed-up
805 factor \mathbf{S}_u was calculated to evaluate the efficiency of mLS in comparison to DIS and DDM for
806 the sensitivity calculation with respect to all design parameters. The corresponding results are
807 presented in Tables 4 and 5. It is worth noting that mLS outperforms DDM in the calculation
808 of all sensitivities, with the lowest speed-up value being 1.05 for the sensitivity related to y_6 , and
809 the highest being 30.86 for the sensitivity related to y_7 . When compared with DIS, mLS outper-
810 forms in the calculation of sensitivities for five out of the eight parameters, with speed-up values

811 ranging from 1.97 to 7.62. The remaining sensitivities calculated using mLS slightly underperform
 812 compared to DIS, with values between 0.57 and 0.94. In general, mLS demonstrates a reduction
 813 in the computational effort for all cases, compared with the other methods that behave better
 814 for specific cases of sensitivities. This is further supported by the speed-up based on the mean
 815 number of dynamic analyses, which is 1.58 when comparing against DIS and 3.36 when comparing
 816 against DDM.

817 In this example, the results are reported in Tables 4 and 5 for a target coefficient of variation
 818 of 20% for the sensitivity estimates. This choice represents a trade-off between accuracy and com-
 819 putational cost in the context of a large-scale finite element model, where each dynamic analysis
 820 is numerically expensive. For preliminary design studies, a 20% coefficient of variation is typically
 821 sufficient, as the sign and relative magnitude of the sensitivities already provide meaningful insight
 822 into the influence of the different parameters on the failure probability. It should be emphasized
 823 that this target does not represent a limitation of the proposed method: as demonstrated in
 824 Example 1, significantly smaller target coefficients of variation (e.g., 5% or even lower) can be
 825 attained at increased computational cost.



(a) Evolution of the sensitivity estimator with respect to the number of dynamic analyses.

(b) Coefficient of variation of the sensitivity estimator with respect to the number of dynamic analyses.

Figure 19: **Example 2:** Comparison of sensitivity results associated with y_1 using Directional Importance Sampling (DIS), the Domain Decomposition Method (DDM), and multidomain Line Sampling (mLS).

Method	$\partial\tilde{p}_F/\partial y_1$	$N^{(1)}$	$\partial\tilde{p}_F/\partial y_2$	$N^{(2)}$	$\partial\tilde{p}_F/\partial y_3$	$N^{(3)}$	$\partial\tilde{p}_F/\partial y_4$	$N^{(4)}$
mLS	-8.0×10^{-3}	250	-5.1×10^{-3}	292	-1.9×10^{-3}	1450	-3.4×10^{-3}	948
DIS	-6.8×10^{-3}	1632	-3.8×10^{-3}	1336	-2.5×10^{-3}	823	-3.0×10^{-3}	708
DDM	-8.6×10^{-3}	1801	-3.9×10^{-3}	3341	-2.1×10^{-3}	6720	-3.8×10^{-3}	2607
S_u (mLS vs DIS)	6.53		4.58		0.57		0.75	
S_u (mLS vs DDM)	7.20		11.44		4.63		2.75	

Table 4: **Example 2:** Sensitivity estimates for y_q , $q = 1, \dots, 4$, number of dynamic analyses ($N^{(q)}$), and speed-up (S_u) of mLS versus other methods, for a target coefficient of variation of 20%.

Method	$\partial\tilde{p}_F/\partial y_5$	$N^{(5)}$	$\partial\tilde{p}_F/\partial y_6$	$N^{(6)}$	$\partial\tilde{p}_F/\partial y_7$	$N^{(7)}$	$\partial\tilde{p}_F/\partial y_8$	$N^{(8)}$	N_{mean}
mLS	-3.3×10^{-3}	856	6.3×10^{-4}	5402	1.3×10^{-3}	148	7.1×10^{-4}	622	1246.00
DIS	-2.8×10^{-3}	1684	6.7×10^{-4}	5059	2.0×10^{-3}	1128	7.4×10^{-4}	3392	1970.25
DDM	-2.3×10^{-3}	5660	1.0×10^{-3}	5685	1.1×10^{-3}	4568	1.1×10^{-3}	3102	4185.50
S_u (mLS vs DIS)	1.97		0.94		7.62		5.45		1.58
S_u (mLS vs DDM)	6.61		1.05		30.86		4.99		3.36

Table 5: **Example 2:** Sensitivity estimates for y_q , $q = 5, \dots, 8$, number of dynamic analyses ($N^{(q)}$), and speed-up (S_u) of mLS versus other methods, for a target coefficient of variation of 20%.

826 The variation in the number of dynamic analyses required to estimate the different sensitivities,
827 as especially seen in Table 5, is mainly due to the effect of each parameter on the geometry
828 of the elementary failure domains. This is reflected in how the distances $c_{[r]}^j$ change along an
829 exploration line l_j when varying a specific parameter. Parameters that induce stronger and more
830 global modifications in the system response generate larger changes in the crossing distances. In
831 contrast, parameters whose influence is weaker for the entire system, or more localized, lead to
832 smaller sensitivity magnitudes. In both cases, the variance of the estimator depends on how much
833 these induced changes vary from sample to sample: if the impact on the crossing distances remains
834 relatively stable across the exploration lines, the estimator variance is reduced; otherwise, larger
835 sample to sample fluctuations lead to an increase in variance. The different order of magnitude and
836 convergence patterns observed for the sensitivities associated with the parameters y_q in Figures 17
837 and 18 are a direct manifestation of this same effect.

838 From a physical point of view, the negative sensitivities with respect to y_q , $q = 1, \dots, 5$, indicate
839 that increasing the thickness of the walls associated with these design parameters helps reduce
840 the maximum displacements of the building and, consequently, the failure probability. On the

841 other hand, the positive sensitivities related to y_q , $q = 6, \dots, 8$, which indicate an increase in the
 842 thickness of the walls of the upper floors (from the 11th to the 16th floor), generate a rigid-body
 843 effect at the upper part of the building, causing an increase in the maximum displacement and,
 844 consequently, in the failure probability of the system. It is important to note that, in magnitude,
 845 the positive sensitivities are smaller than the negative ones, which implies that changing the
 846 parameters y_q , $q = 1, \dots, 5$, has a greater effect on the failure probability. The sensitivity estimates
 847 provide useful guidance for improving the building's design, especially in early decision-making
 848 stages, by identifying the structural parameters that most strongly influence the system reliability.

849 7. Conclusions and outlook

850 This work applies multidomain Line Sampling (mLS) to estimate the sensitivity of the first
 851 excursion probability in linear systems with non-proportional damping under stochastic Gaus-
 852 sian loading. Sensitivities are computed as partial derivatives of the first excursion probability,
 853 incorporating derivatives of the system's spectral properties, eigenvalues, and eigenvectors.

854 The sensitivities are computed using mLS, which requires fewer dynamic analyses, offering
 855 improved efficiency and stability compared to Domain Decomposition Method (DDM) and Direc-
 856 tional Importance Sampling (DIS). A key advantage of mLS over DIS lies in its failure domain
 857 exploration approach: each exploration line in mLS provides information on the derivative of in-
 858 dividual elementary failure domains, while DIS, along a single direction, offers information only
 859 on the derivative of the union of all failure domains. Furthermore, a key advantage of mLS over
 860 DDM is its perpendicular approach to each elementary failure domain, allowing for more precise
 861 sensitivity estimation, whereas DDM follows a directional approach, capturing less information
 862 about the sensitivity of the limit state function. Moreover, mLS has demonstrated high efficiency
 863 in calculating all sensitivities, whereas DIS and DDM are slightly outperformed in calculating
 864 some specific sensitivities.

865 Potential directions of future research could explore the following:

- 866 • The direct application of mLS to the failure probability integral, rather than to the estima-
 867 tion of the effective contribution.
- 868 • The effect of the weights as a mass probability function for estimating the effective contri-
 869 butions.

- 870 • The estimation of sensitivities with respect to additional model parameters, such as excita-
871 tion parameters and response thresholds.
- 872 • The implementation of the proposed method in reliability-based design optimization (RBDO)
873 problems.
- 874 • Extension of the proposed method to systems where the elementary failure domains are
875 weakly non-Gaussian, in line with recent developments on directional sampling methods for
876 non-Gaussian excitation [55]
- 877 • An extended formulation of mLS for settings in which, in addition to the stochastic loading,
878 selected structural parameters are also treated as random variables, particularly in cases
879 where this parametric uncertainty is comparable to or dominant over the excitation uncer-
880 tainty.

881 The above-mentioned issues are currently being investigated by the authors.

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889 Appendix A. Gradient of the failure probability

890 The expression in Equation (15) is derived by rewriting the first excursion probability from
891 Equation (10) as:

$$p_F(\mathbf{y}) = \int_{\mathbf{z} \in \mathbb{R}^{n_{KL}}} H(-g(\mathbf{y}, \mathbf{z})) f_{\mathbf{Z}}(\mathbf{z}) d\mathbf{z}, \quad (\text{A.1})$$

892 where $H(\cdot)$ denotes the Heaviside step function. Differentiating Equation (A.1) with respect to a
893 design parameter y_q yields:

$$\frac{\partial p_F(\mathbf{y})}{\partial y_q} = - \int_{\mathbf{z} \in \mathbb{R}^{n_{KL}}} \delta(-g(\mathbf{y}, \mathbf{z})) \frac{\partial g(\mathbf{y}, \mathbf{z})}{\partial y_q} f_{\mathbf{Z}}(\mathbf{z}) d\mathbf{z}, \quad q = 1, \dots, n_Y, \quad (\text{A.2})$$

894 where $\delta(\cdot)$ denotes the Dirac delta function. Using the identity [56]:

$$\int_{\mathbb{R}^n} f_1(\mathbf{x})\delta(f_2(\mathbf{x})) d\mathbf{x} = \int_{f_2(\mathbf{x})=0} \frac{f_1(\mathbf{x})}{\|\nabla f_2(\mathbf{x})\|} d\sigma, \quad (\text{A.3})$$

895 with $f_1(\mathbf{x})$ an arbitrary function, $f_2(\mathbf{x})$ a differentiable function with a non-zero gradient at the
 896 point where $f_2(\mathbf{x}) = 0$, and $d\sigma$ a differential surface element, Equation (A.1) leads to the following
 897 expression for the gradient of the failure probability [57]:

$$\frac{\partial p_F(\mathbf{y})}{\partial y_q} = - \int_{g(\mathbf{y}, \mathbf{z})=0} \frac{\partial g(\mathbf{y}, \mathbf{z})}{\partial y_q} \frac{1}{\|\nabla_{\mathbf{z}} g(\mathbf{y}, \mathbf{z})\|} f_{\mathbf{z}}(\mathbf{z}) dS, \quad q = 1, \dots, n_Y. \quad (\text{A.4})$$

898 Appendix B. Brief overview of Line Sampling

899 Line Sampling [49, 58] aims to estimate the failure probability for performance functions that
 900 are weakly or moderately nonlinear [59, 60]. For its deployment, it is assumed that a reliability
 901 problem is cast in the standard Gaussian space using an iso-probabilistic transformation. Then,
 902 it is necessary to determine the so-called *important direction* $\boldsymbol{\gamma}$, which is a unit Euclidean norm
 903 vector originating at the origin and pointing toward the nearest point on the failure domain.
 904 This important direction can be determined using different methods, as presented in [49, 61].
 905 Figure B.20 contains a schematic representation of a performance function in a two-dimensional
 906 space along the so-called important direction $\boldsymbol{\gamma}$. Due to the rotational invariance of the Gaussian
 907 distribution, the Gaussian variable \mathbf{z} in the n -dimensional space \mathbb{R}^n can be expressed in terms of
 908 a rotated coordinate system as:

$$\mathbf{z} = \mathbf{R}\mathbf{z}^\perp + \boldsymbol{\gamma}z^\parallel, \quad (\text{B.1})$$

909 where \mathbf{R} is an $n \times (n-1)$ matrix; \mathbf{z}^\perp is an $(n-1)$ -dimensional vector representing the coordinates in
 910 the hyperplane orthogonal to $\boldsymbol{\gamma}$; and z^\parallel is a scalar representing the coordinate along the direction
 911 parallel to $\boldsymbol{\gamma}$. The orthonormal basis of the new coordinate system is defined by the square
 912 matrix $[\mathbf{R}, \boldsymbol{\gamma}]$. Then, it is possible to demonstrate that $z^\parallel = \boldsymbol{\gamma}^T \mathbf{z}$ and $\mathbf{z}^\perp = \mathbf{R}^T \mathbf{z}$, both of which
 913 are associated with standard Gaussian distributions in one dimension and $(n-1)$ dimensions,
 914 respectively. It is important to note that, for practical purposes, an explicit form of the matrix \mathbf{R}
 915 is not required (as explained later in Section 5.1).

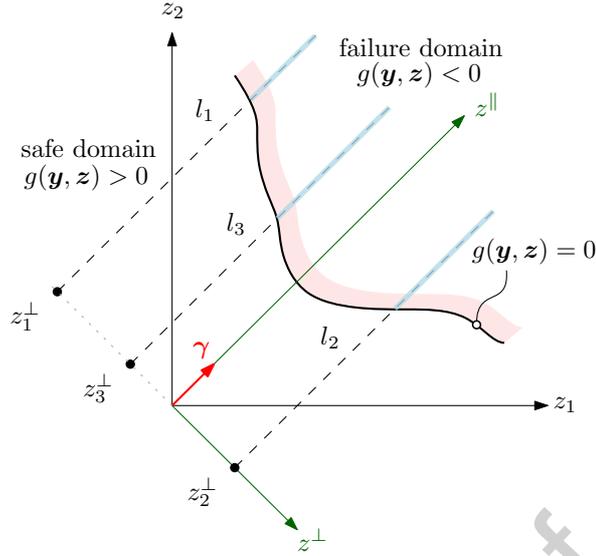


Figure B.20: Schematic representation of Line Sampling technique.

916 Once the problem is formulated in Gaussian space and the important direction has been de-
 917 termined, the Line Sampling scheme operates by generating z_j^\perp , $j = 1, \dots, N$ samples distributed
 918 according to a standard Gaussian distribution in $(n-1)$ dimensions, where N denotes the number
 919 of samples lying in the hyperplane orthogonal to the important direction γ or, equivalently, along
 920 the coordinates z_j^\perp . Then, lines are generated from these samples, denoted as l_j , $j = 1, \dots, N$,
 921 which are parallel to the important direction γ and pass through the aforementioned samples z_j^\perp .
 922 Finally, the numerical integration is performed along the one-dimensional lines within the failure
 923 domain, and the failure probability is estimated by averaging their contributions. The advantage
 924 of employing the rotated coordinate system is that it reduces the multidimensional integration
 925 problem to one-dimensional line integrals in standard Gaussian space, which can be evaluated
 926 efficiently. The entire process is illustrated in Figure B.20, where the performance function is
 927 denoted as $g(\mathbf{y}, \mathbf{z})$. The generated samples are z_1^\perp , z_2^\perp , and z_3^\perp , and the corresponding lines
 928 generated from these samples are denoted as l_1 , l_2 , and l_3 , respectively. The integration over
 929 the failure domain is represented by the light blue line. Note that, since this is a bidimensional
 930 example, z^\perp corresponds to a scalar, whereas in high-dimensional problems, it corresponds to a
 931 vector.

932 Appendix C. Derivative of the effective contribution $\tilde{p}_{s_j}(\mathbf{y}, z_j^\perp)$.

933 The dependence of the effective contribution on the design vector \mathbf{y} is explicitly represented
 934 through the integration domain $\Omega(\mathbf{y})$, as discussed in Section 4.2 and shown in Equation (30).

935 This derivative can equivalently be expressed in terms of the indicator function $I_{F_{i,k}}$. Considering
 936 the exploration line $l_j(u) = R_{s_j} \mathbf{z}_j^\perp + \alpha_{s_j} u$, and the integration domain $\Omega_j^\parallel(\mathbf{y})$ decomposed into
 937 $(-\infty, -\beta_{s_j}(\mathbf{y})) \cup [\beta_{s_j}(\mathbf{y}), +\infty)$, Equation (30) can be written as:

$$\begin{aligned} \frac{\partial}{\partial y_q} \tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp) = \\ \frac{\partial}{\partial y_q} \left(\int_{-\infty}^{-\beta_{s_j}(\mathbf{y})} \frac{I_{F_{s_j}}(l_j(u), \mathbf{y})}{\sum_{h=1}^{n_\eta} \sum_{\ell=1}^{n_T} I_{F_{h,\ell}}(l_j(u), \mathbf{y})} f_{Z_{s_j}^\parallel}(u) du + \int_{\beta_{s_j}(\mathbf{y})}^{+\infty} \frac{I_{F_{s_j}}(l_j(u), \mathbf{y})}{\sum_{h=1}^{n_\eta} \sum_{\ell=1}^{n_T} I_{F_{h,\ell}}(l_j(u), \mathbf{y})} f_{Z_{s_j}^\parallel}(u) du \right), \end{aligned} \quad (\text{C.1})$$

938 where along the exploration line the distribution is standard Gaussian, hence $f_{Z_{s_j}^\parallel}(u) = \phi(u)$.

939 Using the definitions introduced in Section 5 for the crossing distances $c_{[r]}^j(\mathbf{y})$, the overlap
 940 count $m_{[r]}^j$, and the selector $\kappa_{[r]}^j$, it is possible to write:

$$I_{F_{s_j}}(l_j(u), \mathbf{y}) = \sum_{r=1}^{n_S} \kappa_{[r]}^j \left(H(u - c_{[r-1]}^j(\mathbf{y})) - H(u - c_{[r]}^j(\mathbf{y})) \right), \quad (\text{C.2})$$

$$\sum_{h=1}^{n_\eta} \sum_{\ell=1}^{n_T} I_{F_{h,\ell}}(l_j(u), \mathbf{y}) = \sum_{r=1}^{n_S} m_{[r]}^j \left(H(u - c_{[r-1]}^j(\mathbf{y})) - H(u - c_{[r]}^j(\mathbf{y})) \right). \quad (\text{C.3})$$

941 Thus, the fraction from Equation (C.1) can be expressed as:

$$\frac{I_{F_{s_j}}(l_j(u), \mathbf{y})}{\sum_{h=1}^{n_\eta} \sum_{\ell=1}^{n_T} I_{F_{h,\ell}}(l_j(u), \mathbf{y})} = \sum_{r=1}^{n_S} \frac{\kappa_{[r]}^j}{m_{[r]}^j} \left(H(u - c_{[r-1]}^j(\mathbf{y})) - H(u - c_{[r]}^j(\mathbf{y})) \right). \quad (\text{C.4})$$

942 The derivative of the Heaviside function with a moving threshold is given by

$$\frac{\partial}{\partial y_q} H(u - c_{[r]}^j(\mathbf{y})) = -\delta(u - c_{[r]}^j(\mathbf{y})) \frac{\partial c_{[r]}^j(\mathbf{y})}{\partial y_q}.$$

943 Therefore, differentiating Equation (C.4) with respect to the design parameter y_q yields:

$$\frac{\partial}{\partial y_q} \left[\frac{I_{F_{s_j}}(l_j(u), \mathbf{y})}{\sum_{h=1}^{n_\eta} \sum_{\ell=1}^{n_T} I_{F_{h,\ell}}(l_j(u), \mathbf{y})} \right] = \sum_{r=1}^{n_S} \frac{\kappa_{[r]}^j}{m_{[r]}^j} \left(\delta(u - c_{[r]}^j(\mathbf{y})) \partial_{y_q} c_{[r]}^j(\mathbf{y}) - \delta(u - c_{[r-1]}^j(\mathbf{y})) \partial_{y_q} c_{[r-1]}^j(\mathbf{y}) \right). \quad (\text{C.5})$$

944 Finally, using the sifting property [62] $\int_{-\infty}^{\infty} \phi(u) \delta(u - c) du = \phi(c)$ in Equation (C.1) with

945 Equation (C.5) gives:

$$\frac{\partial}{\partial y_q} \tilde{p}_{s_j}(\mathbf{y}, \mathbf{z}_j^\perp) = \sum_{r=1}^{n_S} \frac{\kappa_{[r]}^j}{m_{[r]}^j} \left(\phi(c_{[r]}^j(\mathbf{y})) \partial_{y_q} c_{[r]}^j(\mathbf{y}) - \phi(c_{[r-1]}^j(\mathbf{y})) \partial_{y_q} c_{[r-1]}^j(\mathbf{y}) \right), \quad (\text{C.6})$$

946 which is Equation (52).

947 Appendix D. Derivative of unit impulse response function

948 The unit impulse response function in Equation (6) can be rewritten as

$$h_i(t, \mathbf{y}) = \sum_{r=1}^{2n_D} A_{r,i}(\mathbf{y}) B_r(t, \mathbf{y}), \quad (\text{D.1})$$

949 where $i = 1, \dots, n_\eta$, with

$$A_{r,i}(\mathbf{y}) = \frac{\boldsymbol{\gamma}_i^T \boldsymbol{\phi}_r(\mathbf{y}) \boldsymbol{\phi}_r(\mathbf{y})^T \mathbf{g}_a(\mathbf{y})}{\boldsymbol{\phi}_r(\mathbf{y})^T \mathbf{M}_a(\mathbf{y}) \boldsymbol{\phi}_r(\mathbf{y})}, \quad (\text{D.2})$$

$$B_r(t, \mathbf{y}) = e^{\lambda_r(\mathbf{y})t}, \quad (\text{D.3})$$

950 for $r = 1, \dots, 2n_D$. The partial derivative of $h_i(t, \mathbf{y})$ with respect to the design parameter y_q ,
951 where $q = 1, \dots, n_Y$, is given by

$$\frac{\partial h_i(t, \mathbf{y})}{\partial y_q} = \sum_{r=1}^{2n_D} \left(\frac{\partial A_{r,i}(\mathbf{y})}{\partial y_q} B_r(t, \mathbf{y}) + A_{r,i}(\mathbf{y}) \frac{\partial B_r(t, \mathbf{y})}{\partial y_q} \right). \quad (\text{D.4})$$

952 Defining

$$A_{r,i}^{\{1\}}(\mathbf{y}) = \boldsymbol{\gamma}_i^T \boldsymbol{\phi}_r(\mathbf{y}) \boldsymbol{\phi}_r(\mathbf{y})^T \mathbf{g}_a(\mathbf{y}), \quad A_r^{\{2\}}(\mathbf{y}) = \boldsymbol{\phi}_r(\mathbf{y})^T \mathbf{M}_a(\mathbf{y}) \boldsymbol{\phi}_r(\mathbf{y}),$$

953 the derivatives in Equation (D.4) are expressed as

$$\frac{\partial A_{r,i}(\mathbf{y})}{\partial y_q} = \left(\frac{\partial A_{r,i}^{\{1\}}(\mathbf{y})}{\partial y_q} A_r^{\{2\}}(\mathbf{y}) - A_{r,i}^{\{1\}}(\mathbf{y}) \frac{\partial A_r^{\{2\}}(\mathbf{y})}{\partial y_q} \right) \frac{1}{(A_r^{\{2\}})^2}, \quad (\text{D.5})$$

$$\frac{\partial B_r(t, \mathbf{y})}{\partial y_q} = t e^{\lambda_r t} \frac{\partial \lambda_r}{\partial y_q}. \quad (\text{D.6})$$

954 Then, the terms in Equation (D.5) are computed as

$$\frac{\partial A_{r,i}^{\{1\}}(\mathbf{y})}{\partial y_q} = \gamma_i^T \left(\frac{\partial \phi_r}{\partial y_q} \phi_r^T \mathbf{g}_a + \phi_r \left(\frac{\partial \phi_r^T}{\partial y_q} \mathbf{g}_a + \phi_r^T \frac{\partial \mathbf{g}_a}{\partial y_q} \right) \right), \quad (\text{D.7})$$

$$\frac{\partial A_r^{\{2\}}}{\partial y_q} = \frac{\partial \phi_r^T}{\partial y_q} \mathbf{M}_a \phi_r + \phi_r^T \left(\frac{\partial \mathbf{M}_a}{\partial y_q} \phi_r + \mathbf{M}_a \frac{\partial \phi_r}{\partial y_q} \right). \quad (\text{D.8})$$

955 To evaluate these expressions, the derivatives of the eigenvalue $\lambda_r(\mathbf{y})$ and eigenvector $\phi_r(\mathbf{y})$
 956 with respect to y_q are required. These can be obtained by solving the system proposed in [39],
 957 which decouples the sensitivities of each eigenpair:

$$\begin{pmatrix} \mathbf{K}_a - \lambda_r \mathbf{M}_a & -\mathbf{M}_a \phi_r \\ -\phi_r^T \mathbf{M}_a & 0 \end{pmatrix} \begin{pmatrix} \frac{\partial \phi_r}{\partial y_q} \\ \frac{\partial \lambda_r}{\partial y_q} \end{pmatrix} = \begin{pmatrix} -\left(\frac{\partial \mathbf{K}_a}{\partial y_q} - \lambda_r \frac{\partial \mathbf{M}_a}{\partial y_q} \right) \phi_r \\ \frac{1}{2} \phi_r^T \frac{\partial \mathbf{M}_a}{\partial y_q} \phi_r \end{pmatrix}. \quad (\text{D.9})$$

958 This formulation assumes mass-normalized mode shapes, i.e., $\phi_r^T (-\mathbf{M}_a) \phi_r = 1$, and applies
 959 when eigenvalues are distinct. The main advantage of this approach is its independence from the
 960 remaining eigenpairs, which simplifies derivative evaluations in modal truncation scenarios [63].

961 Appendix E. Pseudo-code for mLS implementation

962 For ease of numerical implementation, the following pseudo-code summarizes the procedure
 963 described in Section 5.6, explicitly listing the variables involved at each step together with their
 964 governing equations.

Pseudo-code 1: Numerical workflow of the mLS-based sensitivity estimator

Step 1: Representation of Gaussian loading:

\mathbf{p} ; Eq. (1)

Step 2: Definition of structural system:

$\mathbf{M}, \mathbf{C}, \mathbf{K}, \mathbf{y}, \mathbf{b}$; Eq. (2)

Step 3: Formulation of augmented matrices:

$[\mathbf{M}_a], [\mathbf{K}_a], \{\mathbf{g}_a\}$; Eqs. (3)–(4)

Step 4: Calculation of spectral vectors:

$\mathbf{a}_{i,k}, \partial_{y_q} \mathbf{a}_{i,k}$; Eqs. (6),(57)

Step 5: mLS loop:

for $j = 1$ to N do

(a) \mathbf{z}_j^\perp, l_j ; Sec. 5.1

(b) $c_{i,k}^{\pm,j}, v_{i,k}^{\pm,j}$; Eqs. (38)–(42)

(c) $\tilde{\mathcal{C}}_{\text{sort}}^j, \tilde{\mathcal{V}}_{\text{sort}}^j$; Eqs. (41),(43)

(d) λ_N^j, λ_P^j ; Eq. (44)

(e) $\partial c_{i,k}^{\pm,j} / \partial y_q, \mathcal{D}^j, \tilde{\mathcal{D}}_{\text{sort}}^j$; Eqs. (53)–(56)

(f) $\mathbf{m}^j, \boldsymbol{\kappa}^j$; Eqs. (48)–(50)

(g) $\tilde{p}_{s_j}, \partial_{y_q} \tilde{p}_{s_j}$; Eqs. (51),(52)

Step 6: Aggregation:

$\tilde{p}_F, \partial_{y_q} \tilde{p}_F$; Eqs. (23),(29)

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1161 **Declaration of interests**

1162 The authors declare that they have no known competing financial interests or personal rela-
1163 tionships that could have appeared to influence the work reported in this paper.

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